

*ALIGNING DEMAND FOR SPARE PARTS
WITH THEIR
UNDERLYING FAILURE MODES*

THESIS

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AFIT/GLM/LAL/95S-9

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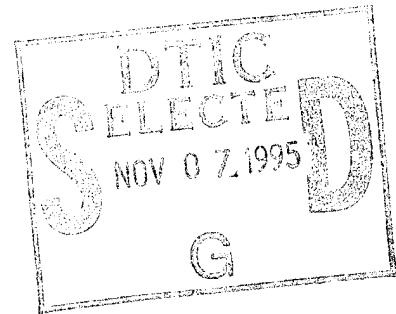
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Presented to the Faculty of the Graduate School of Logistics
and Acquisition Management of the Air Force Institute of Technology

Air University

In Partial Fulfillment of the
Requirements for the Degree of
Master of Science in Logistics Management

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Acknowledgments

The purpose of this study was to determine if demands or maintenance actions are correlated to operational characteristics of the weapon system at the specific work unit code level. The results of this research may help Air Force logisticians predict reparable spares demand with a greater degree of certainty, which could improve overall operational effectiveness while also saving constrained Air Force budget dollars.

Historical literature focusing on demand forecasting problems was reviewed to achieve an understanding of the inherent problems in aligning weapon system failures with demands. F-15C databases from 1993 and 1994 were used as input data and the methodology focused on developing multiple or Poisson regression forecasting models or a suitable Poisson process estimation technique.

This research could not have been completed without the assistance of several people. We are deeply indebted to our advisors, Major Terrance Pohlen and Major Lee Lehmkuhl, for keeping us motivated and on the right path. Our thesis reader, Dr. Dan Reynolds, was also a key player by ensuring our statistical analysis was sound.

We would like to thank three other individuals who played a vital role in our research process. These individuals are: Major Mark Kraus, Dr. Guy Shane, and Major Kevin Lawson. The time they took with us to write/debug SAS or FORTRAN routines and explain data analysis techniques was greatly appreciated.

We would also like to thank our sponsor, Lt Col Dave Peterson, and Mr. Michael Slay from LMI, for their assistance, feedback, and timely inputs of data.

Most of all, we are thankful for our wives and families who stood by us every step of the way in fulfilling this thesis research and living the AFIT experience.

Steven D. Kephart

Richard C. Roberts

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List of Acronyms

AFMC -- Air Force Materiel Command
AOR -- area of responsibility
CAMS -- Consolidated Aircraft Maintenance System
CUMFH -- cumulative flying hours
CUMFH2 -- cumulative flying hours squared
CUMFHNSORT -- cumulative flying hours times number of sorties
D029 -- WRSK Requirements Computation System
D041 -- Recoverable Consumption Item Requirements System
DLR -- depot level reparable
ECM -- electronic counter measure
ERRC -- expendability, recoverability, repairability category
FHNSORT -- cumulative flying hours times number of sorties
FMC -- fully mission capable
ICT -- integrated combat turn
LMI -- Logistics Management Institute
LRU -- line-replaceable unit
MDS -- mission design series
METRIC -- Multi-Echelon Technique for Recoverable Item Control
MRSP -- mobility readiness spares package
NOP -- non-optimized items
NSORT -- number of sorties
NSORT2 -- number of sorties squared
O & M -- operations and maintenance
PBR -- percentage of base repair
RAPS -- Rotable Allocation and Planning System
REALM -- Requirements/Execution Availability Logistics Module
REMIS -- Reliability and Maintainability Information System
RRR -- remove/repair/replace
SAS -- Statistical Analysis System
SRAN -- stock record account number
SRU -- shop replaceable unit
TAIL -- aircraft tail number
VTMR -- variance-to-mean ratio
WMP -- war and mobilization plan
WRSK -- war readiness spares kit
WSMIS -- Weapon System Management Information System
WUC -- work unit code

Abstract

Current Air Force demand forecasting systems, D041 and REALM, which are used to compute reparable authorizations and Mobility Readiness Spares Package configuration quantities, assume demand is driven solely on a flying hour basis. The purpose of this study was to evaluate the relationship between reparable demands, flying hours, and number of sorties. This study is unique because it analyzes the demand, flying hour, sortie relationship at the work unit code level, in an attempt to improve reparable demand forecasting. A three phase methodology is used as the basis for the work unit code level analysis.

The first phase used multiple linear regression to determine a relationship at various levels of the work unit code. Multiple linear regression provided limited correlation between demands, flying hours, and sorties at the work unit code level. Any resulting multiple regression models provided poor estimates of expected demands when a residual analysis was performed against a validation data set.

The second phase used Poisson regression to evaluate the integer, count nature of the demands variable used in the analysis. The Poisson regression results also exhibited poor correlation between demands, flying hours, and number of sorties at the work unit code level.

The third phase fitted a Poisson process to the data in the study. The Poisson process did produce better results than multiple or Poisson regression. However, the Poisson process performed poorly in estimating future demands at the work unit code level, based on historical flying hour and sortie driven demand rate occurrences.

The results of this study support previous demand forecasting research which has

been unable to demonstrate an accurate demand forecasting relationship between demands, flying hours, and number of sorties. Nevertheless, follow-on work unit code level research is suggested with a larger data set. Also, variables other than flying hours and sorties should be considered to evaluate the erratic, uncertain nature of reparable demand forecasting.

ALIGNING DEMAND FOR SPARE PARTS

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I. Introduction

Military aircraft, like other Department of Defense assets, experience failures of their component parts. Major problems faced in forecasting demand for aircraft reparable spare parts are the uncertainties in determining exactly “When?” spares will fail and “What quantities?” need to be ordered to support a weapon system over a specified timeframe. If spares failures and subsequent demand forecasts could be predicted with certainty, the United States Air Force would procure and stock only the required number of assets to support assigned weapon systems. However, failures of aircraft spares are uncertain in nature and the Air Force stocks large quantities of spares to protect against this uncertainty in the system.

Accurate forecasting of reparable spare quantities is definitely important in a peacetime environment to attain mission readiness and minimize operational costs. However, accurate spares forecasting plays an even more critical role in a wartime environment. War readiness spares are authorized in configurations known as Mobility Readiness Spares Packages (MRSPs). MRSP spare quantities are limited to established wartime authorizations and the overall capacity limitations of the deployable bins, which comprise the MRSP. Due to these limitations, accurate and reliable MRSP spares forecasts become a crucial logistical objective and are necessary to ensure vital tactical and strategic objectives are achieved.

In order to accurately forecast peacetime and MRSP spares requirements levels, a relation must be determined between failures of spare parts, the factors that drive the failures, and the demands which are generated. This chapter introduces the issue of reparable spares forecasting and is divided into the following sections: Background and Important Research Aspects. The background section provides a discussion of issues important to spares demand forecasting. Initially, the indentured component structure of aircraft spares and the difference between consumable and reparable spares will be presented. Following the initial discussion of component structure and spares categorization, the current Air Force systems used to estimate reparable and MRSP spares demand, the Recoverable Consumption Item Requirements System (D041) and the Requirements/Execution Availability Logistics Module (REALM), will be reviewed. To conclude the background section, recent research conducted by Headquarters United States Air Force (HQ USAF/LGSI) and the Logistics Management Institute (LMI) will be presented to exhibit a potential problem with D041 and REALM, which both currently use only flying hours to compute/forecast reparable and MRSP spares quantities.

The section on important research aspects will cover significant areas of this demand forecasting research study. In this section, the problem statement, research objectives, research questions, methodology, assumptions, scope, limitations, and implications will be briefly discussed. To conclude the chapter, an overall research summary will be presented.

Background

Reparable spares demand forecasting is a difficult and involved process. However, the indentured structure of aircraft components and the categorization of spares, into consumables and reparables, allow for line item tracking of specific aircraft failures and demands, which are later used to compute spares requirements levels.

Indentured Component Structure. As Isaacson explains, "Aircraft are assumed to have an indentured component structure: they are composed of line replaceable units (LRUs) that are composed of shop replaceable units (SRUs) that are composed of what are called sub SRUs" (Isaacson, 1988:4). When an aircraft experiences a failure of an LRU or SRU component spare, the failure normally results in a demand on base supply. Therefore, base supply could more accurately stock repairable LRU and SRU spares if these failures could be determined with a greater degree of certainty. However, the failures of repairable aircraft spares are uncertain and erratic, resulting in a difficult demand forecasting process. The indentured LRU and SRU components of an aircraft are also further categorized as either consumables or repairables.

Consumables and Repairables. Consumables and repairables are defined as:

Consumables are those items which are expended, consumed or used up beyond recovery in the process of the use for which they were designed or intended
Repairables are defined as those items that may be repaired or reconditioned and returned to a serviceable condition for reuse. (Christensen, 1985:1)

Repairable LRUs, and certain SRUs, are reconditioned or repaired in the field through a process known as the base level repair cycle system (Christensen, 1985:2). The base level repair cycle is the first echelon of a two echelon system known as the Aircraft Logistics Support Network. The echelon above base level repair is known as the depot level (Isaacson, 1988:6). The Air Force Logistics Support Network is depicted in Figure 1-1 on the next page. Depending on a particular base's percentage of base repair, and an LRUs or SRUs expendability, recoverability, repairability category (ERRC) code and technical order specifications, the LRU or SRU can be repaired at the base or at the depot level (Christensen, 1985:1).

LRUs and SRUs that can be repaired are referred to as repairables or repair cycle assets. Repairables are typically complex, expensive, and have low demand rates (Sherbrooke, 1992:45). "Ninety-five percent of all money spent on supplies stocked in a

typical base supply organization is spent on repair cycle assets” (Christensen, 1985:2).

However, in spite of this large investment, “Reparable assets consist of only five percent

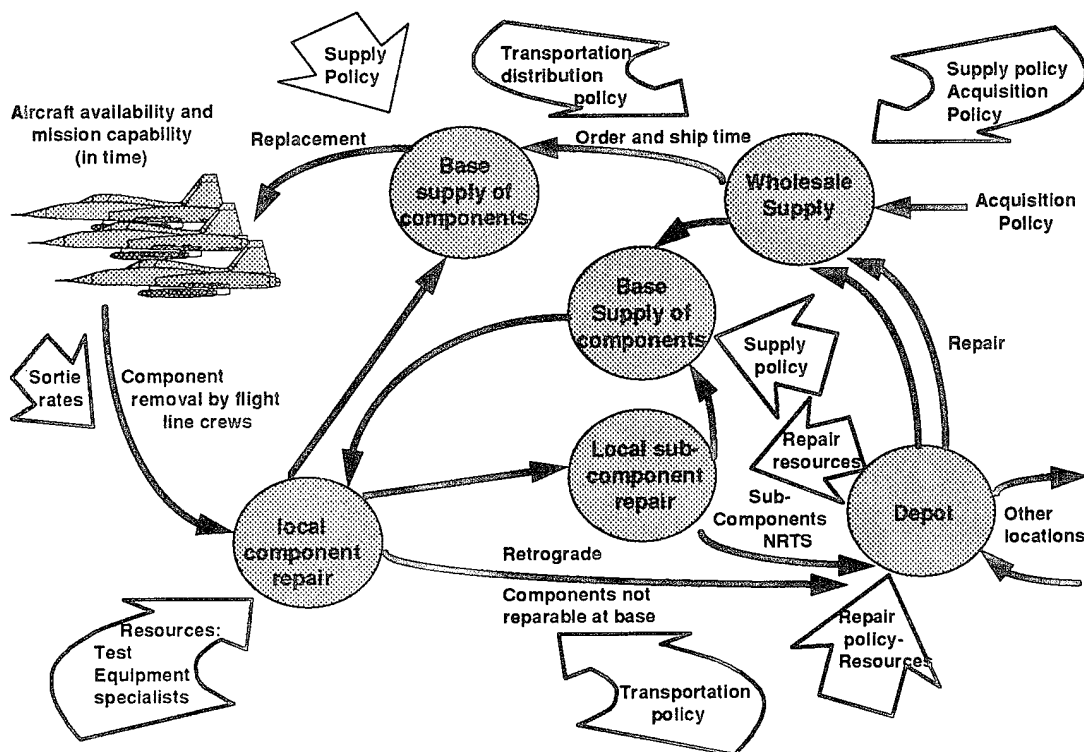


Figure 1-1. Aircraft Logistics Support Network (Isaacson, 1988:6)

of the total line items in the Air Force inventory because of their high cost and reparability” (Christensen, 1985:2).

Reparables become extremely important in supporting Air Force weapon systems, especially in a wartime environment. The Air Force also gains an economic advantage by procuring only the required number of reparables, due to the large investment required to build-up home station and MRSP inventories. However, forecasting demand for aircraft reparable spare parts is a difficult task. This difficulty arises from: “(a) substantial variability in spares demands, even in peacetime (statistical uncertainty), and (b) instability

in force structure, force beddown, flying hour programs, funding profiles, item reliabilities, and other characteristics (state-of-the-world uncertainty)” (Adams, 1993:1).

This statistical and state-of-the-world uncertainty appear to produce offsetting effects on demand forecasting. The DoD, particularly the Air Force, has sponsored significant research to reduce the statistical uncertainty and variability in demand forecasting. However, the current downsizing of the DoD, representative of state-of-the-world uncertainty, contributes to the demand forecasting uncertainty. Another aspect of the inherent demand forecasting uncertainty is possibly found in the current D041 system of the Air Force Materiel Command (AFMC).

D041. The D041 system computes requirements for aircraft reparable spares and uses an eight-quarter moving average to estimate demand for assets, which, as Adams explains, is “A technique that gives no more weight to recent observations than to older, less relevant observations” (Adams, 1993:1). D041 also assumes flying hours and the number of demands are proportional and follow a linear relationship (Adams, 1993:2). In other words, the more a weapon system is flown, the demand for spares should increase at a proportional rate. By assuming demands are only related to flying hours, D041 provides adequate estimates for those reparable spares which actually fail on a flying hour basis. However, D041 could over or under stock those items which fail according to weapon system operational characteristics other than flying hours. In comparison to D041, REALM is another flying hour based forecasting system, which focuses on computing MRSP spares estimates.

REALM. According to Abell, “REALM is the software module of AFMC’s Weapon System Management Information System (WSMIS) that computes requirements for war readiness spares” (Abell, 1993:xxx). However, REALM also assumes demands are related to only flying hours and computes requirements for only flying hour driven parts (Clarkson, 1994:4).

Initial problems with REALM requirements computations were highlighted in the Coronet Warrior exercise conducted in 1988. During this exercise, the 94 Tactical Fighter Squadron from Langley AFB, was supported by a remove/repair/replace (RRR) War Readiness Spares Kit (WRSK) assessed at C-2 for sorties. However, despite this tailored down kit, the demands for spare parts were less than expected; only approximately 35 percent of the assets in the kit were issued during the exercise (Pipp, 1988:1). Coronet Warrior lead to the conclusion that "Demand/break rate data bases need major review, especially in regard to non-optimized and wartime adjustment factor items" (Pipp, 1988:3). As depicted in Coronet Warrior, the WSMIS/REALM calculation of the number of required reparable spares, based purely on flying hours, proved to be inaccurate. Nevertheless, Coronet Warrior was considered a single data point to be used as a benchmark in further investigation of reparable demand forecasting.

To summarize Air Force spares demand forecasting, Air Force aircraft have an indentured component structure and are comprised of both consumable and reparable spares. The demand for reparable spares is normally low, erratic, and uncertain in nature, which presents difficulty in accurately forecasting demand. The primary reparable and MRSP spares demand forecasting programs, D041 and REALM, assume demands are driven solely on a flying hour basis. However, recent research by HQ USAF and the LMI indicates the linear relationship between demands and flying hours is questionable.

Recent Research. In a 1994 research study conducted by HQ USAF/LGSI and the LMI, an analysis of Operation DESERT SHIELD/STORM data reflected the strictly flying hour based approach for estimating spares demand is not totally accurate (HQ USAF and LMI, 1994:6). As shown in Figure 1-2 on the next page, HQ USAF/LGSI and the LMI determined that assuming demands are proportional to flying hours tends to overstate demands. If an assumption is made demands are purely sortie based, demands would be understated. The "truth," or the actual number of demands/sortie, lies between

flying hours and sorties. Thus, individual parts may be sortie driven, flying hour driven, or a combination of the two (HQ USAF and LMI, 1994:6).

The uncertainty in the relationship between demands, flying hours, and sorties is the impetus behind this research. This research covers new territory in demand

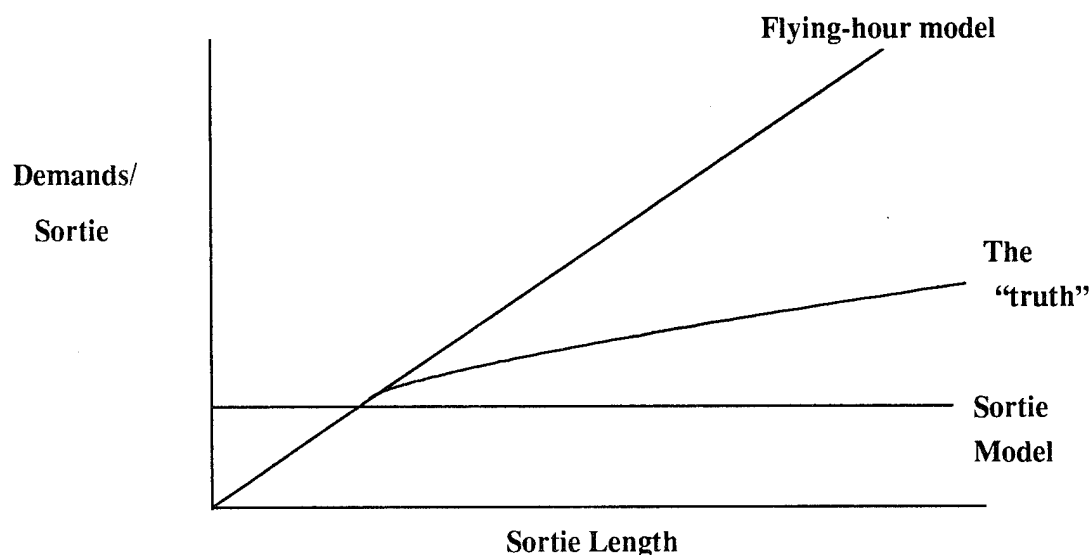


Figure 1-2. Graph of Demands/Sortie versus Sortie Length (HQ USAF and LMI, 1994:6)

forecasting because it analyzes the relationship between reparable demands, flying hours, and sorties at the work unit code level. The research goal is to develop a flying hour, sortie, demand relationship at the work unit code level, which can be used to develop more accurate, reliable MRSP configurations. The important aspects of the research study will now be covered in the next section.

Important Research Aspects

To provide an overview to this research study, a brief synopsis of the following research aspects will be provided: problem statement, research questions, methodology, scope, limitations, assumptions, and implications.

Problem Statement. The specific problem is to determine whether demands/maintenance actions of aircraft reparable spare parts are correlated, at the work unit code level, to flying hours and/or number of sorties. Current requirements models assume a direct, linear relationship to the number of flying hours. However, demands/maintenance actions could be driven by other factors, or a combination of factors, such as flying hours and/or number of sorties.

Research Objectives. The overall objective of this research is to expand on previous demand forecasting research and focus on a different aspect of demand forecasting. This aspect is a “pioneering” attempt to align demands, flying hours, and sorties at the work unit code level. The study has two research objectives. The first objective is to determine if demands/maintenance actions, flying hours, and number of sorties are correlated at the work unit code level. Based on extent of correlation from the first objective, the second objective is to identify specific work unit code decision rules which estimate the demands, or maintenance actions, given a specified quantity of flying hours and/or number of sorties. If correlation exists between demands/maintenance actions, flying hours, and number of sorties at the work unit code level, it could possibly be used as a more accurate means of forecasting reparable spare parts demand used in the computation of Air Force peacetime and wartime/MRSP requirements levels.

For the purposes of this study, the work unit code level is defined in the following manner. A typical work unit code is five alpha-numeric digits, for example, 11A99. 11A99 is considered the “five-digit” work unit code level. The two, three, and four work unit code levels correspond to the number of the same digits in any similar group of work unit codes. For example, 11A9_ is the “four digit” work unit code level, while 11A __ is the “three digit” level. The “two digit” work unit code level is considered 11 ___. The blank spaces represent any other alpha-numeric digits. For analysis purposes, the work unit codes are aggregated into two, three, four, or five digit levels by matching the same

first two, three, four, or five digits of the work unit codes. A breakdown of the work unit code structure and a comparison of the two digit work unit codes to respective weapon system components is included in Appendix A. The study targets specific work unit code levels because Air Force maintenance organizations track all maintenance actions performed on a weapon system through the use of the five digit work unit codes.

Research at the work unit code level is important for the following reason: If the number of specific work unit code level maintenance actions can be correlated to flying hours and/or sorties, a match of the work unit code to its corresponding national stock number could provide forecasts of the number of spares required for a specified flying hour and sortie profile. Furthermore, if a specified wartime mission profile is known, the work unit code level/national stock number match could be used to determine the required spares configuration needed in any deployable MRSPs. Another significant factor for focusing the research at the work unit code level is that the data used in this research tracks maintenance actions on the aircraft, not demands on base supply. In reality, each maintenance action on the aircraft may or may not result in a demand on supply. However, to simplify the study, spares demand and maintenance actions are assumed to be equivalent.

Research Questions. The following research questions are developed for this research study:

1. Is there a relationship between demands/maintenance actions, flying hours, and number of sorties at the work unit code level?
2. Can decision rules be established to forecast demands/maintenance actions based on a spares work unit code alone?

These research questions will be tested and answered by evaluating the extent of correlation between demands/maintenance actions, flying hours, and number of sorties at the work unit code level. The analytical techniques of multiple regression, Poisson

regression, and fitting of a Poisson process will be used to evaluate the data and are outlined in the following section covering methodology.

Methodology. D041 and REALM currently assume a direct linear relationship between spares demand and flying hours (Clarkson, 1994:4). This research methodology will use a three phase approach to determine the extent of correlation between the criterion variable, demands/maintenance actions, and the predictor variables, cumulative flying hours and number of sorties. Phase One uses multiple regression. Phase Two focuses on Poisson regression, while Phase Three evaluates the demands/maintenance actions, flying hours, and sortie relationship by fitting a Poisson process. Each phase of the methodology will now be presented.

Multiple Regression. The objective of multiple regression is to construct a probabilistic model that relates a dependent, or criterion variable, Y, to more than one independent or predictor variable (Devore, 1991:526). The criterion variable used for the multiple regression is demands or maintenance actions. The predictor variables are cumulative flying hours and number of sorties. Upon obtaining the criterion and predictor variables from the data, multiple linear regression will be performed to determine whether or not maintenance actions, cumulative flying hours, and number of sorties are correlated.

Multiple regression will be performed against “reduced,” first-order models and “full,” second-order, interaction models. Specific hypotheses will be developed to evaluate the multiple regression models. The hypotheses to be tested for the reduced model are:

$$H_o : \beta_1 = \beta_2 = \dots \beta_k = 0$$

$$H_a : \text{at least one } \beta_i \neq 0 \text{ (} i = 1, \dots, k \text{)}$$

Variables with p-values greater than $\alpha = 0.05$ will be considered as non-contributing factors in any reduced multiple regression models.

Additional multiple regression analysis will be attempted to introduce second order and interaction terms to the multiple regression model. The “reduced” first order multiple regression model and a “full” multiple regression model, which contains all higher order or interaction variables, will be tested to determine if additional terms contribute to the models. The hypotheses to be tested are:

$$H_o : \text{model is } Y = \beta_0 + \beta_1x_1 + \beta_2x_2 + \varepsilon \text{ (reduced model)}$$

$$H_a : \text{model is } Y = \beta_0 + \beta_1x_1 + \beta_2x_2 + \beta_3x_1^2 + \beta_4x_2^2 + \beta_5x_1x_2 + \varepsilon \text{ (full model)}$$

If H_a is true, p-value comparisons to an $\alpha = 0.05$ will be performed to determine which higher order or interaction terms contribute to the model. Upon completing the multiple regression analysis, Phase Two of the methodology, Poisson regression, will be used to analyze the data.

Poisson Regression. Poisson regression is based on the discrete Poisson distribution and is normally used to evaluate count data. The data used in this study represents discrete counts of maintenance actions occurring at specific work unit code levels for a specified number of flying hours and sorties. The Poisson distribution is suitable for this analysis because the number of demands/maintenance actions changes and is dependent on the level of flying hours and sorties experienced by each individual aircraft.

Three Poisson regressions will be performed against specific work unit code levels. Initially, two Poisson regressions will be run, one for cumulative flying hours and one for number of sorties. The third Poisson regression will include both cumulative flying hours and number of sorties. The primary measure of fit in Poisson regression is the deviance, which is similar to the residual error in linear regression. Generally, the smaller the value of the deviance, the better the fit of the model (Statistix, 1985:183). The model with the smallest deviance will be used to estimate demands or maintenance actions from a validation data set. Comparison testing of p-values to an $\alpha = 0.05$ will determine which

variable coefficients actually contribute to the resulting Poisson regression model. Upon completing Poisson regression, the data will be analyzed as a Poisson process in Phase Three of the methodology.

Poisson Process. Fitting a Poisson process is similar to Poisson regression in that both techniques are based upon the discrete Poisson distribution. The random variable of interest represents the total number of X occurrences of some phenomenon during a specified period of time or within a specified region. If the physical process generating the occurrences satisfies three conditions, (stationary, independent time increments, and the probability of two or more occurrences in time t is some function of t), then the distribution of X must be a Poisson distribution (Degroot, 1986:254,255). For this study, the Poisson process is fit to the data to model the number of demands or maintenance actions that occur over a specified number of flying hours or sorties.

In analyzing the data by fitting a Poisson process, parameters will be estimated from a 1993 data set. These parameters will be used to calculate the expected number of maintenance actions from a 1994 validation data set. Confidence intervals and probabilities of falling outside the bounds of the confidence intervals will also be calculated. Residuals on the Poisson process will be determined by subtracting the expected number of 1994 maintenance actions from the actual number of 1994 maintenance actions. Also, a null hypothesis that the 1993 and 1994 demand rates are the same will be tested against the alternate hypothesis that the demand rates are different for each two digit work unit code level.

The three phase research methodology uses multiple regression, Poisson regression, and fitting of a Poisson process. Initial analysis with multiple regression provided limited results. After meetings with the AFIT Statistics Department and a review of the data structure, Poisson regression and fitting of a Poisson process were attempted to improve upon the limited results obtained through multiple regression. The

methodology was eventually segmented into the three phases in an attempt to find the optimal method of aligning demands, flying hours, and sorties at the work unit code level. The following section covers the assumptions of the research.

Assumptions. The assumptions made in performing this study are outlined as follows: First, the demands/maintenance actions which are generated are assumed to be steady state and all aircraft are assumed to be configured in the same manner. This assumption is necessary because different types of sorties suggest different requirements be placed on the aircraft. For example, an air-to-air sortie requires different "demands" of the aircraft, as compared to an electronic counter measure sortie. Second, a determination cannot be made from the data as to which aircraft flew what type of sortie. It is assumed any one sortie flown is similar to any other sortie. Finally, every sortie that contains a work unit code is assumed to result in a demand on supply or a maintenance action on the aircraft. In other words, demands and maintenance actions are assumed to be equivalent. However, in reality, a certain percentage of malfunctions on an aircraft would result in "cannot duplicate" or "bench check serviceable" actions by maintenance personnel, which do not generate actual spares demands against base supply.

Scope. The scope of the study outlines the extent and outlook of the research. The extent of the research is focused on one weapon system and its reparable spares failures, in order to determine any relationship between demands/maintenance actions, cumulative flying hours, and number of sorties. The weapon system of interest in this study is the F-15C. If a relationship can be identified for one specific MDS, similar techniques may be applied to other types of aircraft. However, because the F-15C is a fighter aircraft, the results may be skewed positively or negatively from other types of non-fighter aircraft. For example, the same electronics equipment may be found on an F-15 and a C-141 transport plane. The fighter aircraft may experience significantly more stress because of increases in G-forces or rapid increases/decreases in acceleration. Due

to these factors, a reparable item, such as electronics equipment, on a fighter may fail at an earlier rate than that of a C-141. Also, transport aircraft typically have longer sortie profiles than fighters, thus there could be more or less reparable failures on a transport as compared to a fighter, depending on the specific component. Simply said, the type of mission performed by different weapons systems may be a large factor in generating demands/maintenance actions and this will not be analyzed. However, since all aircraft inevitably break, the methodology could possibly be used to derive failure rate demand patterns for other types of aircraft.

The outlook of the study is focused on a specific time period and uses Reliability and Maintainability Information System (REMIS) data. The data bases used in this study are from a discrete timeframe covering less than a year, thus, the study in a time sense is narrow. This narrow timeframe does increase the possibility of abnormal demand figures. The data is worldwide F-15C data and covers only peacetime flying profiles. The results would undoubtedly be different, to some extent, if the data were from a wartime or even an exercise environment. Also, only REMIS data will be evaluated in this study. Due to the downloads of Consolidated Aircraft Maintenance System (CAMS) data into REMIS, the REMIS database was considered the best source to obtain demand/maintenance action, flying hour, and sortie information on an operational weapon system.

Limitations. Limitations are the extreme points or boundaries which restrict or confine the research. Several of the limitations include: lack of diversity in the type of aircraft studied, timeline of the study, constraining factors within the data, environmental inputs, and the quality of maintenance performed on each aircraft. The study is limited to only one type of aircraft in order to simplify assumptions used in the methodology. The timeline of the study is limited to discrete periods based on the data downloaded from REMIS. The 1993 database covered eight months of data, while the 1994 database

contained only five months of data. Larger data sets covering longer timeframes could improve overall analysis employed in the methodology.

A major limitation to this study is the data. Initial sorting of the data files indicated numerous work unit codes with only a small number of occurrences. However, this is typical for low demand, reparable items. Due to this small number of occurrences, the number of regressions performed against specific, individual work unit codes will be limited, especially at the five digit work unit code level. A solution to the limited number of occurrences is to aggregate the work unit codes at the two, three, or four digit level and perform the regression/estimation analysis at the various work unit code levels.

Also, 160 of the aircraft were deleted from the 1993 data set due to an absence of maintenance action data, leaving a population of 247 aircraft. However, the 1994 data set contained data on 340 aircraft, but for only a five month period. Despite deleting 160 aircraft from the 1993 data and the limited timeframe covered by the 1994 data, both data sets are still quite large. Each data set contains nearly 50,000 combined sortie and maintenance action images, which provides a large sample suitable for analysis.

A final limitation with the data is the number of available predictor variables. The predictor variables used in the multiple and Poisson regression analysis were cumulative flying hours and number of sorties. However, in analyzing "reduced" versus "full" regression models, additional carriers can be added by taking these predictor variables to a power, or linearly combining the variables, and inserting them in the model. Based on the REMIS data used in this study, only flying hour and sortie data could be obtained. The availability of additional predictor variables, such as engine cycles or "logged" times when a specific component is in use, for example, electronic countermeasure components, could improve the correlation present within the regression models.

Environmental factors are also a limitation to the study. Factors such as heat, humidity, and sand, will not be considered, however, these variables are possibly a

contributing factor to demands or maintenance actions. Finally, the quality of maintenance performed is also vital to a spares performance. Past histories of each airframe play key roles in future failure rates of reparable. Despite this fact, the age and accumulated flying hours on each airframe will not be analyzed.

All of the previously discussed factors are limitations of this study. After identifying these limitations, demands or maintenance actions for specific reparable can be identified by the tangible variables contained in this report. The limited variables play a more minor role.

Implications. From this research, a better model for estimating demands or maintenance actions for reparable could be developed, thus potentially saving the Air Force tight budget dollars. Initial estimates by HQ USAF/LGSI and LMI predict substantial savings by estimating demand from both flying hours and sorties. Based on the 1986 USAF War and Mobilization Plan (WMP-5), an F-15 C/D 30-day MRSP costs approximately \$14.7 million. Under the HQ USAF/LGSI and LMI combined flying hour and sortie approach, cost of a similar MRSP, based on the 1993 WMP-5, drops to \$7.2 million (HQ USAF/LGSI and LMI, 1994:9). Therefore, this research has the potential to significantly reduce the spares requirement for Air Force budget dollars.

Air Force logisticians can also make better decisions by having better failure predictors. By obtaining accurate knowledge of failure characteristics of spares, a logistician can use the most appropriate demand forecasting techniques to stock required quantities of critical items, which ultimately enhances operational readiness. Another implication of this research is to design more efficient and effective MRSP configurations, allowing for more combat capability per dollar. Having the right number of parts going to the right place at the right time, not only saves money, but also conserves vital mobility airlift capability. Other implications include: more accurate capability assessments, better

predictions of failures to demands, and more accurate allocations of scarce resources in a downsizing military environment.

Summary

The problem of forecasting failures of reparable aircraft spares is filled with uncertainty. This uncertainty complicates the issue of determining accurate reparable spares levels. Given the time-tested problem of forecasting demand for reparable spare parts, an improved method is needed to determine what drives specific parts to fail, when these parts will fail, and how many need to be ordered. The current D041 and REALM computation processes, which are based on the linear relationship of demands to flying hours, are being closely scrutinized by HQ USAF and the LMI.

HQ USAF and the LMI have studied the effects of incorporating both flying hours and sorties as predictors of failures and demand. However, this research was aggregated across entire weapon systems. The distinction of this research is to target demands/maintenance actions at the specific work unit code level and determine if a relationship exists between demands/maintenance actions, cumulative flying hours, and number of sorties. Capability to properly identify demand rates for specific spares, or groups of spares, becomes a key objective in forecasting reparable spares. The system can be improved upon and the bottom line of this research is to seek out a better method.

The remainder of this research focuses on aligning demand/maintenance actions for spares with flying hours, and sorties, at specific work unit code levels. Chapter II provides a literature review covering historical research on forecasting demands for spares, current models used to calculate spares requirements, and a discussion on how MRSPs are authorized, built, and maintained. Chapter III outlines the methodology and covers the origin and make-up of the data. Chapter IV presents and analyzes the results

obtained from applying the research methodology. Finally, Chapter V presents the conclusions of the research and suggests recommended areas of follow-on research.

II. Literature Review

Introduction

The purpose of this literature review is to examine historical research completed on determining accurate demands for aircraft spares and how these demand forecasts affect MRSP spares levels or requirements. The research covered in this literature review includes only past research examining the relationship between reparable failures and/or predictor variables, which is the focus of this study. Demand research focusing on other analysis methods, such as time series analysis, will not be reviewed.

The chapter will review the historical demand forecasting research and examine the models used to calculate reparable spares demands. A review of reparable spare parts management in the civilian sector will also be presented. Following the civilian sector review, lessons learned from past exercises, such as Coronet Warrior and Bull Rider, will be reviewed to highlight the necessity for further research in ascertaining whether all spares demands are purely flying hour driven or are somehow affected by other operational characteristics of the weapon system. The chapter will conclude with a discussion of how demand forecasting affects MRSPs and how MRSPs are currently authorized, built, and maintained.

Historical Review

RAND Research. Forecasting demands for aircraft recoverable spare parts has challenged researchers for decades. Beginning in the 1950s, RAND Corporation began pursuing the problem of forecasting demands for aircraft recoverable spare parts. One of the earliest works on aligning spares demands with their underlying failure modes was conducted by Geisler, Brown and Hixon of the RAND Corporation in July 1954 (Adams, 1993:4). This research effort targeted three B-47 bases and compiled over 1300 aircraft

months of data, which included both consumables and recoverables (Geisler, 1954:ii).

After several months of research and data analysis, the researchers concluded:

It is shown that there is a surprisingly low amount of demand both as to kind and quantity of aircraft spare items,... Furthermore, we could find no significant correlation between the kinds or quantity of items demanded and the aircraft flying activity, measured in flying hours, landings or aircraft months. (Geisler, 1954: ii)

Following this initial research, Geisler and Brown published a report concerning the lack of significant correlation between the number of demands and flying activity. In this report, Geisler and Brown concluded:

... the daily combined demand over all items show more variation than expected from the Poisson distribution ... if the Poisson distribution is used to represent the demand pattern for spare items because of its mathematical convenience, the actual distribution for either individual items or combined may be more extreme, in that the variance of the distribution will be greater than the mean value of demand. (Brown, 1954:ii,iii)

From this initial research, it was recognized very early that accurate prediction of demand for aircraft spares is an extremely complicated process and not easy to determine.

The Poisson distribution, which is discussed by Geisler and Brown, "is the most widely known and often used form of stochastic model with important mathematical properties that make it especially tractable and useful" (Abell, 1993:xxx). If the time separating demands for aircraft spares follows an exponential distribution, the quantity of spares demanded during a specified period of time is said to be Poisson distributed (Sherbrooke, 1992:21). The probability that a specific event x occurs under the Poisson distribution is represented by:

$$p(x|\lambda \tau) = \frac{(\lambda \tau)^x e^{-\lambda \tau}}{x!}$$

where e = base of the natural logarithm system and has numerical value of approximately 2.71828
 τ = mean resupply time

λ = mean rate of demand

x = number of units in resupply (Feeney, 1966:393).

The event “ x ” can be thought of as a failure of a spare that ultimately results in a demand. For the Poisson distribution, the variance of X , $\text{Var}[X]$, where X is a random variable, is equal to the expected value of X , $E[X]$. Therefore, the variance to mean ratio, [VTMR], is equal to one (Sherbrooke, 1992:21). However, the majority of processes modeled by a Poisson distribution show some variability and are referred to as compound Poisson distributions. As Christensen explains, “The main feature of the compound Poisson distribution is that the variance can exceed the mean” (Christensen, 1985:7). Although the compound Poisson is theoretically more accurate, the simple Poisson is assumed to provide reasonable answers due to the difficulty in calculating variances of individual items (Mitchell, 1983:445).

Following Geisler and Brown’s research, which concluded demand for aircraft spares did not follow a Poisson distribution, Youngs, Geisler and Mirkovich of RAND performed a study focusing on the method of confidence intervals as applied to the Poisson distribution. In this study, the researchers determined, “Since the true demand rate for a supply item is seldom known, it is necessary to estimate it from statistical demand data. This means that the Poisson parameter can at best be estimated subject to sampling error, i.e., it can only be trapped within certain intervals with a specified probability” (Youngs, 1954:1).

Realizing historical demands may not be indicative of true demand predictions, the research by Youngs, Geisler and Mirkovich was the initial study which lead the movement away from the Poisson distribution to the negative binomial distribution, as a means of modeling demand for aircraft spares (Adams, 1993:7). The negative binomial distribution generalizes the Poisson distribution and, as Adams explains, “The negative binomial distribution applies to situations in which events occur at random, but the variance of the number of events in nonoverlapping time intervals of equal length is higher than allowed

by the Poisson distribution” (Adams, 1993:xxi). Over short time intervals, the variance to mean ratio of demand may not change to a great extent. However, “Over longer time periods, the variance to mean ratio of demand changes, sometimes substantially, and is greater than one” (Sherbrooke, 1992:58). To compensate for this demand uncertainty, the negative binomial distribution uses both the mean and variance of demand and recognizes variance to mean ratios greater than one (Sherbrooke, 1992:58). The negative binomial distribution is represented as follows:

$$neg(x) = \binom{a+x-1}{x} b^x (1-b)^a$$

where

$$a = \mu / (V-1), a > 0$$

$$b = (V-1) / V, 0 < b < 1$$

μ = mean

V = variance to mean ratio, $V > 1$ (Sherbrooke, 1992:60-61).

In 1956, Bernice Brown, also from RAND, used the prior RAND demand research from 1954 and 1955 to publish a formal research memorandum exploring the demand characteristics of aircraft spares (Adams, 1993:8). In this memorandum, Brown realized the majority of prior research was only performed on a single aircraft, the B-47, and much work remained to be accomplished in the area of demand forecasting (Brown, 1956:iii). Nevertheless, Brown concluded:

Low average demand rates are characteristic of a large proportion of all aircraft parts. The slow moving, low cost parts account for a small fraction of the total dollar value of issues, but because of their large number and, often their essentiality to the functioning of the aircraft, they constitute a significant logistics problem. Demand for most spare parts tends to be erratic. (Brown, 1956:vii)

To account for this low demand of a large proportion of aircraft spares, T. A. Goldman of RAND performed research in 1957, which made the first suggestion of possibly looking at spares in aggregate groups instead of single line items (Adams,

1993:13). Goldman's research concluded: "The family of parts rather than the individual part number should be the basic unit in demand analysis and forecasting. The levels of demand for spares appears to be associated with certain fundamental characteristics of the part" (Goldman, 1957:vi).

In 1963, following Goldman's research, H. S. Campbell of RAND focused on using multiple correlation and regression analysis as a method for predicting demand for aircraft spares (Adams, 1993:16). He used data from the B-52 aircraft in researching spares demands against seven operational variables: "Sorties flown, flying hours, flying hours at low altitude, bombing navigation training units, fire control system usage, ECM system usage, and periodic inspections" (Campbell, 1963:14). Using multiple correlation and regression analysis with net demands as the dependent variable, Campbell found coefficients of multiple determination, R^2 , values ranging from a high of 0.74 for electronic systems to a low of 0.20 for fire control and gunnery systems (Campbell, 1963:22,23). After analyzing his results, Campbell concluded: "Demands appeared to be related to flying hours and sorties, with the former providing the stronger relationship. Other operational variables showed little relationship, and multiple correlations of demand on several operational variables typically showed little improvement over flying hours alone" (Campbell, 1963:v,vi). A major difference in Campbell's work as compared to the early research of Geisler, Brown and Hixon was Campbell's research included only recoverable items (Adams, 1993:16).

One of the major themes present throughout the 1950s and mid 1960s research was that low demand drives uncertainty in determining future spares levels (Adams, 1993:9). However, in 1964, Feeney and Sherbrooke of RAND introduced a system oriented, instead of an item oriented, approach to forecasting demand for spare parts. Feeney and Sherbrooke's research centered around a mathematical technique known as Bayesian inference, which attempts to reduce the uncertainty in demand forecasting by

studying the performance of comparable items in a system (Feeney, 1964:vi). Bayesian inference is used to “combine the prior distribution on all items with a Poisson demand process to estimate a posterior distribution for demand of each individual item” (Sherbrooke, 1992:73).

Feeney and Sherbrooke also applied Palm’s Theorem into the forecasting of demand for recoverable spares (Adams, 1993:18). As Sherbrooke explains, “Palm’s Theorem estimates the steady state probability distribution of the number of units in repair from the probability distribution of the demand process and the mean of the repair time distribution” (Sherbrooke, 1992: 21). Palm’s Theorem assumes the following:

1. The demand process for an item is Poisson distributed with an annual mean of λ .
2. The demand and repair processes are independent of each other.
3. The repair time for each failed unit is independent, identically distributed with a mean of τ years, represented by τ . Also, the distribution is unspecified.
4. Slack service capacity exists. There are always infinite channels.

Given these assumptions, Palm’s Theorem generalizes that the number of units in repair at any time is Poisson distributed with mean, $\lambda\tau$ (Crawford, 1981:8). Palm’s Theorem is also useful in generalizing the basic repairable pipeline quantity model. The basic repairable pipeline quantity model is represented by:

$$s = RCQ + OSTQ + NCQ + SLQ + K$$

where

s = pipeline stock

RCQ = repair cycle quantity; $DDR \times PBR \times RCT$

$OSTQ$ = order and ship time quantity; $DDR \times (1 - PBR) \times OST$

NCQ = not repairable this station(NRTS)/condemned quantity;
 $DDR \times (1 - PBR) \times NCT$

SLQ = safety level quantity; $C(\sqrt{3 \times (RCQ + OSTQ + NCQ)})$

K = constant, .5 if unit cost is greater than \$750, or .9 if unit cost is \$750 or less

C = C factor or number of standard deviations to protect against stockouts
 DDR = daily demand rate
 PBR = percent of base repair
 RCT = repair cycle time
 OST = order and ship time
 NCQ = NRTS cycle time (Christensen, 1985:4).

Palm's Theorem allows for calculations of the specific number of assets in a reparable pipeline. For example, on a base with 100 percent base repair, $PBR = 1$, an item that averages 10 demands per year, $\lambda = 10$, and takes 0.3 years to repair, $\tau = 0.3$, the average number of assets in the pipeline is 3; $\lambda\tau = (10)(0.3) = 3$.

Feeney and Sherbrooke's research into Bayesian inference and Palm's Theorem was the final RAND research focusing exclusively on the demand forecasting problem (Adams, 1993:19). What soon followed was the start of a series of models focusing not only on the demand forecasting problem, but also the effects of demand on reparable stockage policy.

The first model to appear was known as the Multi-Echelon Technique for Recoverable Item Control (METRIC) (Adams, 1993:19). Over the next several years, variations of METRIC, known as Mod-METRIC, Vari-METRIC, and Dyna-METRIC, were also developed. A brief summary on each of the METRIC models will now be presented:

METRIC. METRIC was developed by Craig Sherbrooke of RAND Corporation in 1966 and is a method for estimating aircraft spares requirements in a multi-echelon, base-depot inventory system (Adams, 1993:xxi,20). METRIC considers only single indenture items (LRUs) and has a system-wide objective of minimizing expected backorders (Sherbrooke, 1992:47). METRIC attempts to improve overall system-wide performance by optimizing procurement of new assets while also evaluating the effects of redistributing on-hand assets between depot and base levels (Sherbrooke, 1968:3). METRIC estimates spares requirements in consideration of the following assumptions:

1. The decision as to whether a base repairs an item does not depend on the stock levels or the workload.
2. The base is resupplied from the depot, not by lateral resupply from another base.
3. The (S-1,S) inventory policy is appropriate for every item at every echelon.
4. Optimal steady-state stock levels are determined.
5. System-wide objective is minimizing expected backorders.
6. System is conservative. There are no condemnations.
7. Demand data from different bases can be pooled (Sherbrooke, 1992:46-47).

Despite METRIC's system-wide objective of minimizing backorders across reparable LRU spares requirements, there are three critiques of the METRIC model:

1. METRIC doesn't consider the LRU/SRU indenture relationship.
2. METRIC doesn't consider end item availability, which is important if cannibalization is allowed.
3. METRIC's minimization of backorders tends to drive the purchase of too many low cost items.

Despite these critiques, METRIC has the distinction of being the first practical application of multi-echelon inventory theory in the Air Force (Sherbrooke, 1986:311).

Mod-METRIC. Mod-METRIC extended the METRIC model by considering the indentured or hierarchical LRU/SRU parts structure of weapons systems (Muckstadt, 1973:474). All of the assumptions previously outlined for the METRIC model also apply to Mod-METRIC. However, there are additional assumptions specific to Mod-METRIC:

1. LRUs are expensive and degrade the mission when they fail. Thus, the percentage of base repair (PBR) should be close to one.

2. SRUs are relatively inexpensive and are remove/replace. There is some PBR for SRUs, but it is more economical to have extra stock of SRUs and fill the depot reparable pipeline with these assets.

3. Every LRU failure is the result of just one SRU failure.

4. SRUs belong to only one LRU (Sherbrooke, 1992:63).

Mod-METRIC's main objective is to minimize LRU backorders, while subject to a cost constraint on both LRU's and SRU's (Muckstadt, 1973:481). Mod-METRIC's original Air Force use was to compute reparable spares requirements on the F-15 weapon system (Muckstadt, 1973:481).

Vari-METRIC. Vari-METRIC was developed by Mike Slay of the Logistics Management Institute (LMI) in 1980 (Sherbrooke, 1986:311). Vari-METRIC was designed to correct inaccuracies in the original METRIC model. When METRIC was first developed, it was clear that the model clearly understated backorders. However, in most instances the errors were not large and the simplicity of the METRIC model overshadowed any inaccuracies (Sherbrooke, 1986:311). Vari-METRIC attempts to improve upon the METRIC model by incorporating the negative binomial approach in estimating expected backorders. By incorporating the negative binomial approach, Vari-METRIC uses both the mean and the variance to compute the number of units in resupply (Sherbrooke, 1992:97,98). Vari-METRIC assumes the following:

1. It is appropriate to use the (S-1,S) inventory policy at each echelon.

2. Repair capacity and parts are ample. Repair time is independent of the number of units already in repair.

3. Poisson demand with a mean that is constant, independent of the number of units in repair or resupply. Pipeline quantities have a negative binomial distribution.

4. No units are condemned.

5. No lateral support. All resupply comes from the depot (Sherbrooke, 1986:311).

Vari-METRIC improves upon METRIC and Mod-METRIC in computing spares requirements, but it is not used extensively in the Air Force. Vari-METRIC is a complex model requiring a significant amount of data and most spares requirements computations produce fairly close results by using the METRIC model.

Dyna-METRIC. Dyna-METRIC was originally developed by R. J. Hillestad of RAND in 1980 (Adams, 1993:21). Dyna-METRIC (Version 4) is an analytic model that uses mathematical equations to forecast how logistics support processes would effect a flying unit's capability in a dynamic wartime environment (Isaacson, 1993:1). All METRIC models prior to Dyna-METRIC considered only steady state conditions, for example, the relatively stable flying activity which is normally experienced in peacetime flying profiles. Dyna-METRIC primarily focuses on the dynamic, flying hour environment, representative of wartime scenarios, and attempts to model spares requirements based on the uncertainty of demands generated during wartime flying activity (Sherbrooke, 1992:184).

Dyna-METRIC has evolved through several enhancements and numerous versions of the model have been developed (Isaacson, 1993:1). Enhanced versions of Dyna-METRIC can generate two different assessment reports. The first report determines performance measures such as spares and aircraft availability. The second report, the problem LRUs report, assesses requirements and identifies a list of problem parts whose support resources and processes constrain aircraft availability (Isaacson, 1993:12-15).

To effectively model the dynamic wartime environment, Dyna-METRIC (Version 4.6) incorporates the following assumptions:

1. LRU demands are proportional to either flying hours or sortie rate.

2. Demands arrive randomly, with a known mean and variance according to either a Poisson or negative binomial distribution.

3. Demands and service process times are independent.

4. Repair and transportation times have known probability distributions (exponential or deterministic mean).

5. There is unconstrained repair capability and no lateral resupply.

6. All aircraft deployed to a single base are identical.

7. Pipeline segments are additive.

8. Aircraft performance measures are computed after attrition.

9. Under the full cannibalization policy, holes are instantly consolidated on as few aircraft as possible.

10. Ability to cannibalize is all or nothing.

11. Repair times vary by component, while transportation times vary by base.

Overall, Dyna-METRIC provides capability assessments by assessing the effects of wartime dynamics while projecting operational performance measures and identifying potential problems (Isaacson, 1993:1). Table 2-1 on the next page summarizes the four METRIC models.

RAND has definitely lead the way in forecasting demand for aircraft reparable spares within the Air Force. Despite the uncertainty and inherent problems in forecasting demand of reparable spares, current Air Force systems perform reasonably well in providing the best available support to the myriad of weapon systems. To provide a contrasting view to Air Force management of reparable spares, management of reparable parts in the civilian sector will now be presented.

Reparable Parts Management in the Civilian Sector. The commercial airline industry encounters problems very similar to those found in the Department of Defense

for estimating demand for reparable (rotatable) aircraft parts. American Airlines Decision Technologies Division developed a PC-based decision support system called the Rotable Allocation and Planning Systems (RAPS) to provide forecasts of rotatable parts demand.

Table 2-1. Summary of METRIC Models

| MODEL | METRIC | MOD- METRIC | VARI- METRIC | DYNA- METRIC |
|-------------------------------|---|--|---|---|
| Indenture | Single | Multiple | Multiple | Multiple |
| Echelons | Multiple | Multiple | Multiple | Multiple |
| Number of Items | Multiple | Multiple | Multiple | Multiple |
| Location | Multiple | Multiple | Multiple | Multiple |
| Demand Assumptions | Steady state, independent, and stochastic demand (Poisson) | Steady state, independent, and stochastic demand (Poisson) | VTMR > 1, independent, stochastic, Poisson demand. Pipeline quantities have negative binomial distribution | Dynamic instead of steady state. Stochastic, multi-period. Considers time dependent scenarios. |
| Objective | Minimize expected backorders | Minimize LRU backorders | Maximize aircraft availability | Readiness, sustainability, and sortie generation |

RAPS provided a multi-million dollar benefit for American Airlines, upon initial implementation, through the identification of over and under allocated parts (Tedone, 1989:62).

RAPS uses linear regression to establish relationships between monthly part removals and various functions of monthly flying hours. Demand data is kept current through monthly updates from an 18-month rolling horizon of spares removals and flying hour data. Coefficients of determination corresponding to the best regressions are calculated and RAPS analyzes possible forecasts based on flying hours or functions of flying hours. Regressions are evaluated on best fit and statistical significance. The entire process of generating demand forecasts by linear regression is completely automated and demand forecasts are sampled periodically to validate the continued use of the system. As a demand forecasting tool, RAPS incurred no costs due to shortages when demand was spread over a month. However, in a worst case scenario modeling total monthly demand on a single day, shortages did occur for some critical spares.

A major benefit of RAPS is an increase in the number of rotatable parts which can be analyzed in a single day. An audit trail is also created to record dates and times of parts analysis. A one time savings of \$7 million and a recurring annual savings of \$1 million were realized by American Airlines on a fleet of over 400 aircraft. RAPS is now the standard tool for allocating rotatables at American Airlines for over 50,000 different types of reparable parts. The impact has been a multi-million dollar improvement in the quantity of on-hand inventory resulting in streamlined generation of spares demand into a more effective, reliable process (Tedone, 1989:68).

RAPS is similar to Air Force demand forecasting systems in that RAPS also uses flying hours to forecast demand. Although RAPS has worked extremely well for American Airlines, the program could prove to be too large to manage in the United States Air Force. American Airlines focuses on managing reparable spares for a fleet of

approximately 400 aircraft. In contrast, the Department of Defense and the United States Air Force support a significantly larger number of aircraft, which are stationed across the globe. Nevertheless, the Air Force continues to strive in improving available demand forecasting systems and exercises/real world scenarios provide unique opportunities to produce valuable lessons learned.

Lessons Learned

Numerous exercises, and even Operation DESERT SHIELD/STORM, have provided an exceptional test bed to determine whether or not the various demand forecasting models, particularly Dyna-METRIC, have accurately forecasted demands. A review of lessons learned from past exercises will now be covered.

Exercises. When Dyna-METRIC was first developed, skepticism existed as to whether or not the model produced accurate results. Several MAJCOM exercises tested and validated the accuracy of the model. Tactical Air Command initiated the testing and validation with the Coronet Warrior exercises (Rhodes, 1988:74). The initial exercise was Coronet Warrior I, which was held at Langley AFB, Virginia in 1987 (Pipp, 1988:1). During Coronet Warrior I, 24 F-15 C/D aircraft were flown for 30 days, at wartime sortie rates, with only a remove/repair/replace WRSK, assessed at C-2 for sorties, for logistical spares support. As a compounding factor, maintenance was limited to an avionics intermediate shop and cannibalization actions (Pipp, 1988:1). Given these input parameters, Dyna-METRIC estimated only 4 aircraft to be FMC at the end of 30 days. The actual number of aircraft FMC after 30 days was 17 (Pipp, 1988:1).

Indeterminate factors impacted the difference between the Dyna-METRIC predictions and the actual results. In general, demands were less than predicted (Page, undated:7). Only approximately 35% of the items in the WRSK were issued during the exercise (Pipp, 1988:1). However, innovative maintenance actions and a "teamwork"

focus, on repairing versus replacing those spares requiring maintenance, definitely influenced the reduced number of supply demands. Another factor influencing the number of demands was that most of the Coronet Warrior sorties were integrated combat turns (ICT). During ICTs, the aircraft is refueled, rearmed and relaunched while never shutting down power to systems and one engine. Because most electrical components, and engines, tend to break from the stress of heating and cooling, there were less spares failures because components were not powered down between sorties (Page, undated:7).

The actual Coronet Warrior results were run through Dyna-METRIC model after the exercise was complete. Dyna-METRIC estimated 16 aircraft FMC at the end of 30 days, which is very close to the actual number of 17 FMC aircraft (Page, undated:8).

Given these results, Tactical Air Command concluded:

The Dyna-METRIC model works well, but further refinement to repair logic may improve the models. Demand/break rate data bases need major review ... more accurate estimates of cannibalization and maintenance times must be included in stockage methodology. This would contribute to the development of better and less expensive WRSK. (Pipp, 1988:3)

The Bull Rider exercise conducted by Strategic Air Command and Volant Cape exercise conducted by Military Airlift Command produced similar results to Coronet Warrior; there were more aircraft FMC and fewer spares demands than predicted by Dyna-METRIC. Overall, using Dyna-METRIC to determine requirements in a dynamic environment works well (Page, undated:12). However, one of Dyna-METRIC's underlying assumptions is that LRU demands are proportional to either flying hours or sortie rate. Failures of some spares in a readiness spares kit may or may not be driven by flying hours, sorties, or a combination of known or unknown operational factors. The uncertainty in determining exact causes of failures is again the thrust of this research. The uncertainty also presents formidable problems in determining exact quantities of assets required in Mobility Readiness Spares Packages (MRSP).

Authorizing, Building, and Maintaining MRSPs

MRSPs are additive stockage levels, above a base's peacetime operating stock, for operational squadrons to support their wartime taskings as outlined in the USAF War and Mobilization Plan (WMP) (Clarkson, 1994:2). Two of Dyna-METRIC's main uses are to determine spares requirements, as well as, a list of problem spares, which become crucial when configuring and building MRSPs.

The process of authorizing, building, maintaining, and ultimately deploying MRSPs is extremely complex. Actions must be accomplished at the field level, the Major Commands, the Air Logistics Centers and Headquarters United States Air Force (HQ USAF)(Clarkson, 1994:2). All MRSPs must first be authorized in the MRSP Authorization Document published by the HQ USAF War and Mobilization and Planning Office (HQ USAF/XOX). An MRSP cannot be built and fielded unless an authorization exists in the MRSP authorization document. The operational requirements, (for example, sorties, utilization rates), plus direct support objectives and operational employment concepts are the major factors which drive the scope and depth of the MRSP (Clarkson, 1994:2). Figure 2-1 on the next page provides a top to bottom diagram of the how an MRSP is authorized.

Once an MRSP is authorized, it enters a lengthy review cycle of approximately one year. The review cycle is designed to ensure the spares package includes all necessary spares to support the scenario for which the package was authorized. The annual review process is comprised of three separate processes: the pre-review, the review, and the post-review (Clarkson, 1994:3). During the pre-review, system program directors suggest additions and deletions to the MRSPs based on the historical usage data and feedback from the major commands (Clarkson, 1994:4). The review process involves analyzing demand data to determine actual spares requirements to include in the MRSP. Sources of

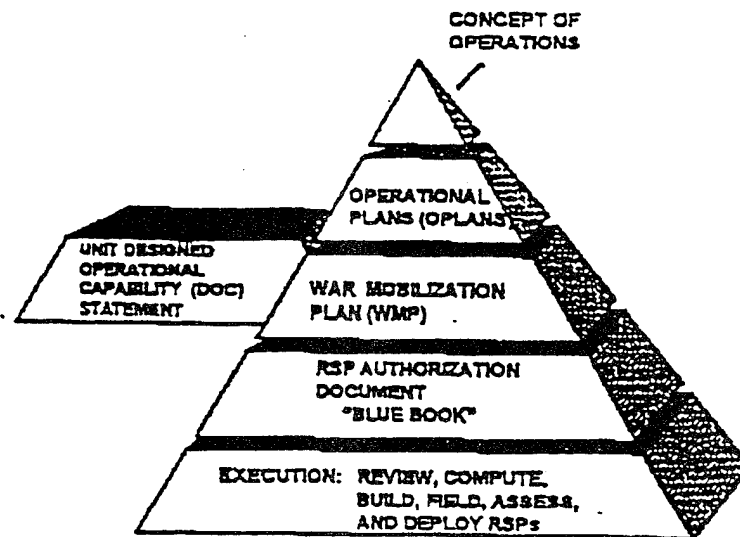


Figure 2-1. How an MRSP is Authorized (Clarkson, 1994:3).

demand data include: D041 rates and factors (which provide worldwide averages on spares usage) and usage data from major command MRSP managers.

After the requirements determination is complete, the Requirements/Execution Availability Logistics Module (REALM) is used to actually compute the MRSP requirements (Clarkson, 1994:4). REALM is the software module of Air Force Materiel Command's Weapon System Management Information System (WSMIS) that computes requirements for war readiness spares (Abell, 1993:xxx). WSMIS uses Dyna-METRIC to assess the wartime capability of existing MRSPs (Blazer, 1988:26). The system used to assess WRSK requirements, prior to Dyna-METRIC, was the WRSK Requirements Computation System (D029). However, in studies performed comparing the requirements cost, backorder performance, and aircraft supportability between the WRSK Requirements Computation System (D029) and Dyna-METRIC, the results revealed Dyna-METRIC computed better, more efficient wartime requirements (Blazer, 1988:26).

Nevertheless, REALM has one main disadvantage. It will only compute requirements for flying hour driven parts. The parts assumed to be non-flying hour driven are known as non-optimized (NOP) items. The NOP requirements are computed external to the guidelines of REALM (Clarkson, 1994:4). The impact of inaccurate forecasts of NOP items can be shown in the make-up of a B-1B, 6 PAA, MRSP. For example, a B-1B, 6 PAA, 14 day MRSP is comprised of 248 line items (669 units) equating to a cost of approximately \$69.2 million. Approximately 45 percent (91 line items, 157 units) of the MRSP is made up of NOP items resulting in a cost of \$31 million. Because NOP items are expensive and comprise such a large part of this B-1B MRSP, the \$31 million price tag suggests a better method is needed to ensure we are putting the right number of NOP items in MRSPs to minimize cost while maximizing aircraft availability (Clarkson, 1994:6).

Prior to the actual REALM computation, the post-review, which is the last of the three MRSP review processes, is conducted. The post review is a final audit check by the system program directors and MAJCOM MRSP managers to further validate established requirements before the REALM computation (Clarkson, 1994:5). Once the post-review is complete, REALM computes the war readiness spares requirements. REALM also uses an embedded model known as the Aircraft Sustainability Model (ASM) to perform a marginal analysis computation to find the most efficient and least cost mix of spares to provide optimum aircraft availability for any given operational support objective (Clarkson, 1994:5).

Normally, during the three step review process, two MRSP kits are built and reviewed. One kit is the contingency kit which supports the force structure and configuration of a squadron for the next fiscal year. The second kit is the buy kit which supports the projected force structure and configuration two years into the future. The

standard time is six months from the annual review to fielding of a new or reviewed MRSP (Clarkson, 1994:3).

The complete process of building MRSPs is extremely expensive to the DoD, not only for the time and manpower invested, but also for the price of the reparable spares. MRSPs are basically comprised of reparable items which are high cost items with low demands. Because the USAF, as a segment of the DoD, spends such a large amount of money on MRSP spares, an economic advantage is gained in procuring only the required number and correct mix of reparable spares. The required number of spares can be more accurately determined if a better method of predicting future spares failures, which ultimately drive demands, can be established.

Currently, REALM, as well as Dyna-METRIC, only focus on flying hour or sortie driven spares failures (Clarkson, 1994:4). Recent research by HQ USAF/LGSI and the LMI in the area of forecasting readiness spares requirements, and lessons learned from past exercises, as well as, Operations DESERT SHIELD/STORM, indicate demand for spares may not be purely flying hour driven (HQ USAF and LMI, 1994:6). Based on this research, REALM and Dyna-METRIC may overstate or understate spares requirements and over or under-allocate available budget dollars on the wrong quantities of readiness spare parts. An accurate method of forecasting spares requirements, based on the most likely failure modes of particular spares, is required. An accurate forecasting method would assist the Air Force in balancing aircraft availability against available budget dollars, when building and configuring MRSPs.

Summary

Forecasting demand for reparable spares is a difficult and involved task. In this literature review, past research performed by RAND Corporation in the area of reparable spares demand forecasting and civilian reparable parts management have been reviewed.

An underlying pattern in all research up through and including Dyna-METRIC is that spares demands are driven on a purely flying hour or sortie basis. However, exercises, DESERT SHIELD/STORM, and recent research by HQ USAF and the LMI reveal all spares may not fail on a strictly flying hour or sortie basis. The problems introduced by forecasting demands based on inaccurate failure modes produce inadequate MRSP configurations, which significantly impact wartime support. The primary objective of this research is to align demands and failure modes at the work unit code level to hopefully improve spares demand forecasting. The methodology to accomplish this research is outlined in the next chapter, along with an explanation of our data and its origin.

III. Methodology

Introduction

The purpose of this study is to determine if demands or maintenance actions of reparable spares are related to cumulative flying hours and number of sorties at the work unit code level. If reparable demands, flying hours, and sorties are related at the work unit code level, this relationship could be used to develop more accurate reparable spare part demand forecasts, leading to economical and efficient Air Force MRSP configurations.

A detailed method is required to conduct a research study and achieve the study's purpose. The detailed method of conducting a study is termed the methodology. This chapter covers the methodology and outlines the research design, research questions, research hypotheses and instruments, and the process used for variable validation. The data collection, gathering, and sorting, population and sample size, and data limitations are also presented. The chapter concludes with a section covering implementation of the research design in a step-by-step sequence.

Research Design

A researcher must develop a distinct path to follow through the various phases of a study to successfully achieve the research objectives. This path is known as the research design. The research design provides three essential benefits to the research process:

First, the design is a plan for selecting the sources and types of information used to answer the research question. Second, it is a framework for specifying the relationships between the study's variables. Third, it is a blueprint that outlines each procedure from the hypotheses to the analysis of the data. (Cooper and Emory, 1995:114)

Several different perspectives must also be considered in developing a research design. These perspectives include: degree of problem definition, method of data

collection, researcher control of variables, purpose of the study, time dimension, and topical scope (Cooper and Emory, 1995:115-117). Each of these perspectives will now be discussed in the context of how they correspond to the development of this research design.

The degree to which the problem is defined can be determined by the type of study to be conducted. The two types of studies are: an exploratory study and a formal study. Exploratory studies deal with discovering prospective research areas and developing testable hypotheses. A formal study is the follow-on process to the exploratory study and performs in-depth testing of the proposed exploratory hypotheses (Cooper and Emory, 1995:115). In the area of aircraft spares demand forecasting, significant research has already been conducted, particularly by the RAND Corporation, HQ USAF, and the Logistics Management Institute (LMI), as outlined in Chapter Two. A common theme of this prior research is that spares demand is influenced by operational aircraft measures, such as flying hours or numbers of sorties. A major distinction of this research study is the evaluation of the relationship between spares demand, or maintenance actions, flying hours, and number of sorties at the work unit code level. Because this research is a follow-on process to prior research, the study is considered a formal study.

Once the problem is defined and the type of study is determined, the method of data collection must be established. Data collection can be performed by two methods: through observation or through use of a survey. Observational data collection deals with using data recorded through observation of a process or activity. Survey data collection requires collecting responses from individuals through the use of questionnaires or survey instruments (Cooper and Emory, 1995:115).

This research study uses observational data collection. The source of data is the Reliability and Maintainability Information System (REMIS) maintenance and sortie database. REMIS data from May to December 1993 and February to June 1994 is used

and covers worldwide failures of F-15C reparable spares. The 1993 and 1994 data sets are extensive and each contain approximately 50,000 combined maintenance action and sortie images. REMIS data is used because it includes the sorties flown by an aircraft and the corresponding maintenance actions performed by support personnel. Thus, the maintenance actions, or demands, can be matched to the exact sortie which caused the failure. The REMIS data is also used to establish three primary variables for the research study. These variables are: demands, or maintenance actions, the cumulative number of flying hours and the number of sorties, which were observed for occurrences of the same work unit code on each particular aircraft.

Once the variables are established, researcher control of variables is categorized into either experimental design or ex post facto design. In experimental design, the researcher attempts to control variables in order to determine if any one variable affects any other variable. In ex post facto design, the researcher has no control over variables and reports findings based on given data and the interactions of the variables (Cooper and Emory, 1995:115-116). Relationships between maintenance actions, cumulative flying hours, and number of sorties will be determined based on quantities obtained from the REMIS data. Because variable values come directly from the REMIS data, and no researcher control over the variables is obtained, this study incorporates ex post facto design.

Although the variables in the study are defined from REMIS data, the purpose of the research study must also be defined. The purpose of a study can be either descriptive or causal. Descriptive studies attempt to answer the "Who, what, where, when, or how much?" within the objectives of the research. Causal studies focus on the question of "Why?" and attempt to demonstrate relationships between variables (Cooper and Emory, 1995:116). This research is a causal study that attempts to show a relationship, at the

work unit code level, between demands/maintenance actions, cumulative flying hours, and number of sorties .

Upon defining the purpose of the study, the time dimension of the study must be determined. The time dimension can be either cross-sectional or longitudinal. Cross-sectional studies are performed once and examine a specific instance in time. Longitudinal studies are carried out several times over a lengthy time period (Cooper and Emory, 1995:116). This research study is a cross-sectional study because it examines data from May to December 1993 and February to June 1994 and will only be performed a single time.

One of the primary areas in a research process involves study and analysis of the results, which can be thought of as the topical scope of the research. The topical scope of the research can take the form of a statistical study or a case study. Statistical studies, as explained by Cooper and Emory, "Are designed for breadth rather than depth. Hypotheses are tested quantitatively. Generalizations about findings are presented based on representativeness of the sample and the validity of the design" (Cooper and Emory, 1995:116). Case studies emphasize qualitative data and the "contextual analysis" of several events, which examine their underlying relationships (Cooper and Emory, 1995:116-117). The topical scope of this thesis research is a statistical study. The REMIS data is quantitative in nature. Also, hypotheses are developed and tested through the use of regression and Poisson process analysis techniques to determine whether or not a relationship exists between demands/maintenance actions, cumulative flying hours, and numbers of sorties at the work unit code level.

In summary, the research design used in this study was developed from the following perspectives. The research is a formal study using an observational data collection method. Ex post facto research design is used and the study is causal in nature. Finally, the study is cross-sectional in relation to time and the topical scope is statistical.

The step-by-step implementation of the research design is discussed in detail later in the chapter. The following section outlines the research questions.

Research Questions

To determine if a relationship exists between demands/maintenance actions, flying hours, and sorties at the work unit code level, the following research questions are developed for subsequent answer:

Research Question One. Is there a relationship between demands/maintenance actions, flying hours, and number of sorties at the work unit code level? To thoroughly develop an answer to the first research question, the following investigative questions will be addressed:

1. Are demands or maintenance actions, flying hours, and number of sorties correlated at the two digit level of specific work unit codes?
2. Do demands or maintenance actions, flying hours, and number of sorties show more, or less, correlation at the three, four, or five digit level of work unit codes, as compared to the two digit level?

The two, three, four, and five digit work unit code levels used in this study are thoroughly explained in Chapter One and Appendix A.

Research Question Two. The second research question is: Can decision rules be established to forecast demands/maintenance actions based on a spares work unit code alone? This question addresses matching the number of demands or maintenance actions to specific work unit codes, which can be used to estimate demand for reparable spares and establish MRSP configuration quantities. Research question two will be answered based on the extent of correlation obtained in analyzing and answering the first research question.

Research Hypotheses and Instruments

The initial relationship between demands or maintenance actions, cumulative flying hours, and number of sorties will be assessed through a three phase process, which uses multiple linear regression, Poisson regression, and fitting of a Poisson process. A three phase process is used to determine the best relationship between demands/ maintenance actions, cumulative flying hours, and number of sorties at the work unit code level. Multiple linear regression is appropriate in Phase One because it establishes probabilistic models that describe the linear relation between the criterion variable, demands/ maintenance actions, and the predictor variables, cumulative flying hours and number of sorties.

However, the criterion variable, demands/maintenance, represents discrete “counts” of the number of demands/maintenance actions. For the purposes of this study, the discrete “counts” of demands/maintenance actions are assumed to be a function of the predictor variables, cumulative flying hours and number of sorties. The counts of discrete events are normally modeled as a Poisson distribution. Thus, in Phase Two, Poisson regression is appropriate and considers the “count” nature of the criterion variable to determine whether or not a relationship exists between demands/maintenance actions, flying hours, and number of sorties. Of course, Poisson regression is based on the use of the Poisson distribution (Myers, 1990:333).

A Poisson process, which is also based on the Poisson distribution, is appropriate for Phase Three in order to model the discrete demands/maintenance actions occurring at the work unit code level for a specified quantity of flying hours or number of sorties. Poisson parameters are calculated from the 1993 data set and are used to estimate occurrences for the 1994 data set. Residual analysis is used to justify the fit of the Poisson process.

By using the three phase process, the first research question will be answered before attempting to answer research question two. The first research question establishes whether or not there is a relationship at the work unit code level between the criterion variable (demands/maintenance actions) and the predictor variables (cumulative flying hours and number of sorties). For the regression techniques, the determination of strong correlation between the criterion and predictor variables will be based on the coefficient of multiple determination (R^2) and hypotheses testing of the linear β coefficients in the regression models. The fit of the Poisson process will be evaluated from the residual analysis comparing actual 1994 demands/maintenance actions to expected demands/maintenance actions and hypotheses testing. The overall goal is to obtain the model with the best fit. The three phases of the research design, multiple regression, Poisson regression, and a Poisson process, will now be presented.

Multiple Regression. The first order, "reduced," model of the multiple regression analysis will take the form: $Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \varepsilon$. The criterion variable used for the multiple regression is demands/maintenance actions and is represented by Y. The predictor variables are cumulative flying hours, x_1 , and number of sorties, x_2 . The x_1 and x_2 are referred to as carriers because they carry information about Y within the regression model (Devore, 1991:526). The parameter β_0 is the Y intercept of the regression plane (Neter and Wasserman, 1985:227). The parameters β_1 and β_2 are referred to as the partial regression coefficients. β_1 indicates the mean response per unit increase in x_1 , the cumulative flying hours, while β_2 indicates the mean response per unit increase in x_2 , the number of sorties (Neter and Wasserman, 1985:228-229).

Upon obtaining the criterion and predictor variables from the data, multiple linear regression will be performed to determine whether or not demands/maintenance actions, cumulative flying hours, and number of sorties are correlated. The determination of strong correlation between the variables in the model will be based on the coefficient of

multiple determination, R^2 , and hypotheses testing of the linear β coefficients in the regression models. The values of R^2 can range from 0 to 1. The higher the value of R^2 , the greater the correlation between the variables in the regression model. If R^2 is equal to 1, the model fits the data perfectly and all observations fall directly on the fitted response surface (Neter and Wasserman, 1985:240).

Specific hypotheses will be tested for the “reduced” multiple regression models. The specific hypotheses test will be set up as follows:

$$H_o : \beta_1 = \beta_2 = \dots = \beta_k = 0$$

$$H_a : \text{at least one } \beta_i \neq 0 (i = 1, \dots, k)$$

In testing $H_o : \beta_1 = \beta_2 = \dots = \beta_k = 0$ against $H_a : \text{at least one } \beta_i \neq 0 (i = 1, \dots, k)$, the following test, referred to as the test of model utility, will be used:

Null hypothesis: $H_o : \beta_1 = \beta_2 = \dots = \beta_k = 0$

Alternative Hypothesis: $H_a : \text{at least one } \beta_i \neq 0 (i = 1, \dots, k)$

$$\text{Test statistic value: } f = \frac{R^2 / k}{(1 - R^2) / [n - (k + 1)]}$$

Rejection region for a level α test: $f \geq F_{\alpha, k, n - (k + 1)}$

R^2 = coefficient of multiple determination,

n = number of data points;

k = number of carriers.

(Devore, 1991:535)

In evaluating the best regression model, cumulative flying hours, which is represented by coefficient β_1 , and the number of sorties, which is represented by coefficient β_2 , will either contribute or not contribute to the regression model depending on the outcome of the hypotheses tests in the test for model utility. For example, if the null hypothesis: $H_o : \beta_1 = \beta_2 = \dots = \beta_k = 0$ is true, then neither cumulative flying hours or a number of sorties contribute to the model or correlate highly with demands/ maintenance actions (the criterion variable). However, if the alternative hypothesis, $H_a : \text{at least one } \beta_i \neq 0 (i = 1, \dots, k)$ is true, then either cumulative flying hours, number of

sorties or both contribute to the model. If the alternate hypotheses is true, comparison of p-values to an $\alpha = 0.05$ will be used to determine which variables contribute to the “reduced” model. Variables with p-values greater than 0.05 will be considered as non-contributing factors in any “reduced” multiple regression models.

In order to determine whether additional carriers contribute to the “reduced” model, the “reduced” model will be tested against a “full” model. The “full,” second order, interaction multiple regression model, containing all higher order or interaction variables, takes the following form: $Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \beta_3 x_1^2 + \beta_4 x_2^2 + \beta_5 x_1 x_2 + \epsilon$. x_1^2 represents cumulative flying hours squared. x_2^2 represents the number of sorties squared. $x_1 x_2$ represents the linear combination of cumulative flying hours and number of sorties. β_3, β_4 , and β_5 are the corresponding regression coefficients. The hypotheses to be tested are:

H_o : model is $Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \epsilon$ (reduced model)

H_a : model is $Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \beta_3 x_1^2 + \beta_4 x_2^2 + \beta_5 x_1 x_2 + \epsilon$ (full model)

In testing H_o versus H_a , the following test will be used:

$$\text{Test Statistic value: } f = \frac{(SSE1 - SSE2) / p}{MSE2}$$

Rejection region: $f \geq F_{\alpha, n-k-1}$

SSE1 = the unexplained variation for the reduced model.

SSE2 = the unexplained variation for the full model.

MSE2 = Residual Mean Square Error from full model.

n = the number of points.

k = the number of regressor variables in the full model.

p = the number of additional regressor variables added to the reduced model to obtain the full model.

(Devore, 1991:536)

If H_a is true, indicating the full model is more appropriate, comparison of p-values to an $\alpha = 0.05$ will be used to determine which higher order or interaction variables contribute to the full model.

A residual will be calculated for the resulting multiple regression model. The residual is the difference between the actual value of demands/maintenance actions, obtained from the 1994 validation data, and the expected number of maintenance actions obtained from applying 1994 flying hour and sortie values to the regression model built from the 1993 data. As Devore explains, "If the residuals are small in magnitude, then much of the variability in observed Y values appears to be due to the linear relationship of the variables in the model and Y, while large residuals suggest quite a bit of inherent variability in Y relative to the amount due to the linear relation" (Devore, 1991:464).

Tests for normality and autocorrelation will also be performed on the regression models. The test for normality will be performed using the Wilk-Shapiro/Rankit Plot. To test for autocorrelation, the Durbin-Watson test will be used. Autocorrelation is the positive correlation of the regression model error terms over time. The Durbin-Watson test determines if variables in the model need to be transformed. The Durbin-Watson test is also used to determine whether or not additional higher order terms, such as x^2 or x^3 , need to be added to the model to correct the autocorrelation problem. If autocorrelation is a problem in the regression model, multicollinearity effects could result through the addition of extra variables. When the independent variables in a regression model are correlated among themselves, the variables exhibit intercorrelation, or what is known as multicollinearity (Neter and Wasserman, 1985:250). An indication of multicollinearity is represented by large changes in the Beta coefficients, when independent variables are added to the model. After analyzing the data with multiple regression, Poisson regression will be used in Phase Two.

Poisson Regression. Poisson regression will be used to evaluate the mean number of demands/maintenance actions, as a parameter of the Poisson distribution. The demands/maintenance actions are assumed to represent a Poisson mean and are a function of the predictor or regressor variables, cumulative flying hours and number of sorties.

The model for Poisson regression is represented as follows:

$$p(y_i; \beta) = \frac{e^{-t_i[\mu(x_i, \beta)]} [t_i \mu(x_i, \beta)]^{y_i}}{y_i!} \quad (i = 1, 2, \dots, n)$$

The function $\mu(x_i; \beta)$ represents the Poisson mean and is referred to as the link function.

The t is assumed to be 1 because the “basis” is taken as a single type of aircraft, the F-15C. The link function relates the predictor or regressor variables to the distribution mean and must always be nonnegative, as well as, user specified (Myers, 1990:334). For this study, the Statistix analytical software package will be used to perform the Poisson regressions and the link function is chosen to be represented by $x_i' \beta$ where $x_i' \beta > 0$. x_i' and β are vectors containing the variables and estimated regression coefficients.

Poisson regression uses an iteratively reweighted least squares technique to produce the maximum likelihood estimators for the regression coefficients in β (Myers, 1990:335). After estimation of the regression coefficients, the Poisson regression model is represented as follows:

$$\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2 + \dots + \hat{\beta}_k x_k$$

$\hat{\mu}$ is the estimate of the mean number of demands or maintenance actions. $\hat{\beta}_0, \hat{\beta}_1$, and $\hat{\beta}_2$ are the maximum likelihood estimates of the regression coefficients. x_1 represents the cumulative flying hours, while x_2 represents the number of sorties.

The primary measure of fit in Poisson regression is the deviance. “The deviance plays a role similar to the residual error in linear regression. The deviance can be thought

of as the “distance measure” between the fitted model and the actual data - the smaller, the better” (Statistix, 1985:183).

In analyzing the data in this study, three Poisson regressions will be performed against specific work unit code levels. Initially, two Poisson regressions will be run, one for cumulative flying hours and one for number of sorties. The third Poisson regression will include both cumulative flying hours and number of sorties. The model with the smallest deviance will be used to estimate demands or maintenance actions from the 1994 validation data set. Comparison testing of p-values to an $\alpha = 0.05$ will determine which variable coefficients actually contribute to the resulting Poisson regression model. If p is less than or equal to $\alpha = 0.05$, the variable contributes to the model. If p is greater than $\alpha = 0.05$, the variable does not contribute to the model. A residual will also be calculated for the resulting Poisson regression model. After performing the Poisson regressions, the last segment of the research design, a Poisson process, will be attempted.

Poisson Process. A Poisson process represents a random variable, such as the total number of demands/maintenance actions, that occur during a specified period of time or within a specified region, for example, a period of flying hours or number of sorties. In order to justify and fit a Poisson process, three conditions must be achieved:

1. The number of occurrences in any two disjoint intervals of time must be independent of one another (Degroot, 1986:254). For instance, although an aircraft may break several times over a specified time period, the probability that at least one break will occur in an upcoming time interval is unchanged.

2. The probability of an occurrence during a smaller time interval must be approximately proportional to the length of the interval. In other words, the process is assumed to be stationary over time (Degroot, 1986:255). This study deals with aircraft data that experience both surges and lulls in operations, which possibly violates this

condition. However, it is assumed over the long-term that steady state conditions prevail and the second Poisson process condition is satisfied.

3. The probability that there will be two or more occurrences in any particular very short interval of time must have a smaller order of magnitude than the probability that there will be just one occurrence (Degroot, 1986:255). Therefore, it is assumed that the probability of two demands or maintenance actions in a small interval is negligible in comparison to the probability of one demand or maintenance action.

In achieving the three conditions listed above, the number of demands or maintenance actions in a fixed interval of time, t , will have a Poisson distribution and the mean is represented by λt . For this study, λ is a positive constant and is the expected number of demands or maintenance actions per cumulative flying hours or number of sorties. Thus, if X is a random variable representing the number of demands or maintenance actions, and has a Poisson distribution with parameter λ , the expected value of X , $E(X)$, is equal to the variance of X , $V(X)$, which both equal λ (Devore, 1991:120).

In evaluating the data sets in this study by fitting a Poisson process, λ 's based on both flying hours and sorties will be calculated from a 1993 data set for specific two digit work unit code levels. The 1993 λ factors will be multiplied by cumulative flying hours and number of sorties, which are obtained from a 1994 validation data set. The resulting, expected number of two digit work unit code maintenance actions will be compared to the actual number of corresponding two digit work unit code maintenance actions in the 1994 data set. A residual will be calculated for the Poisson process by taking the actual number of maintenance actions and subtracting the expected number of maintenance actions. The closer the residual value is to zero, the better the estimation and fit of the Poisson process.

To use a model based on the Poisson process, similar conditions must be maintained between timeframes as outlined in condition two above. Thus, to ensure the demand rates are similar for the 1993 and 1994 data sets, hypotheses testing will be

performed. The null hypothesis, $H_o: \lambda_{93} = \lambda_{94}$, the 1993 and 1994 demand rates are the same, will be tested against $H_a: \lambda_{93} \neq \lambda_{94}$, the 1993 and 1994 demand rates are different, for each two digit work unit code level. The two sample, two sided F-test for equal population variances will be used to test the hypotheses because under the assumption of a Poisson distribution, the mean equals the variance, which both equal λ .

Upon analyzing the data with multiple regression, Poisson regression, and fitting a Poisson process, a comparison of residuals will be performed to determine which technique produces the best estimates of expected demands/maintenance actions. After an appropriate regression model or Poisson process estimation has been developed for the specific work unit code levels, the results will be analyzed to address research questions one and two.

Variable Validation

Variable validation ensures only variables which contribute to the linear regression models are included in the models. The available REMIS data allows for computation of the criterion variable, demands/maintenance actions, and the predictor variables, cumulative flying hours and number of sorties occurring at the work unit code level. These variables were selected because demands/maintenance actions, cumulative flying hours, and number of sorties are the only variables readily obtainable from the REMIS database. Although additional variables may contribute to the relationship between demands/maintenance actions, flying hours, and sorties, these additional variables were not available. Also, the major B-52 regression study conducted by H.S. Campbell in 1963 did consider additional variables other than flying hours and sorties. However, the results indicated the additional variables did not significantly contribute to the regression models (Campbell, 1963:vi). Nevertheless, based on the HQ USAF/LMI research covered in Chapter One, spares demand is now believed to fall somewhere between the pure sortie

and pure flying hour curves. Therefore, flying hours and sorties were chosen as the predictor or estimation variables.

The predictor variables are validated through the use of multiple regression and the comparison of the f -statistic to the critical rejection regions. Poisson regression or Poisson process procedures are also used as sources of variable validation. In building the linear regression models, the predictor variables are added in the multiple regression and additions of higher order variables are tested through the use of the full model. Depending on the outcome of hypothesis testing and deviance values, p -value comparisons to an $\alpha = 0.05$ also validate which variables do and do not contribute to the regression models.

Data Collection, Gathering, and Sorting

The method of data collection is observational. Two databases are used in this study and were received from Mr. Michael Slay at the Logistics Management Institute (LMI). The analysis database contains worldwide REMIS data on the F-15C for the period May to December 1993. The validation database contains F-15C REMIS data for the period February to June 1994. The following data discussion focuses on the 1993 data set. However, similar procedures were used to set up and manipulate the 1994 data set.

The 1993 F-15C REMIS analysis data was separated into two distinct files, one covering maintenance data and the other covering sortie data. REMIS compiled the data from the base level Consolidated Aircraft Maintenance System (CAMS) for worldwide USAF bases who operate the F-15C. The maintenance database covers maintenance actions performed on the F-15C during the specified timeframe. Inclusive in this database is the following information: tail number, Julian date, work unit code, action taken code, when discovered code, how malfunctioned code, the beginning time for maintenance, sortie number, sortie length, sortie mission, days from sortie to maintenance, the number

of sorties, and the Stock Record Account Number (SRAN) of the base assigned the specific aircraft. The sortie database lists all sorties of each F-15C for the applicable timeframe. Inclusive in this database is the following information: tail number, sortie, Julian date, sortie number, mission code, sortie length, number of sorties, and SRAN.

Both the 1993 maintenance and sortie databases were transferred to a mainframe VAX computer in order to use the sorting capabilities of the Statistical Analysis System (SAS). This transfer was necessary due to the combined size of both data files (over 58,000 images), which was too large for a personal computer to handle. The maintenance file was initially sorted by work unit code to identify which specific aircraft tail numbers contained which work unit codes. Also, each aircraft tail number represented in the maintenance file could possibly have several occurrences of the same work unit code. To track which sortie actually resulted in the maintenance action of each specific work unit code, fields from the maintenance database were matched against similar fields in the sortie database. Once the match was complete, both the maintenance and sortie files were merged together into one file. The merged file allowed for the tracking of which sortie resulted in a maintenance action and also allowed for easy calculation of the number of maintenance actions, cumulative flying hours, and number of sorties experienced by each aircraft in the database. However, a lack of maintenance action data on 160 of the aircraft in the 1993 data set made it impossible to track which aircraft sortie resulted in a maintenance action on the aircraft. Therefore, these 160 aircraft were deleted from the 1993 database. The remaining 1993 data set population covered 247 aircraft. The 1994 data set had no missing maintenance action data and contained 340 aircraft.

A FORTRAN program was written to query the merged file and calculate the number of maintenance actions at the 2, 3, 4, and 5 digit work unit code level for each aircraft tail number in the data set. The FORTRAN program also calculates the cumulative flying hours and number of sorties for each aircraft in the data set. The

FORTTRAN program produces files each containing 247 records for the 1993 data set and 340 records for the 1994 data set, which represent the number of aircraft in each data set. Each data record contains the following fields: aircraft tail number (TAIL), work unit code (WUC), demands (DMDS)/maintenance actions (or the number of occurrences of the specific work unit code), cumulative flying hours (CUMFH), number of sorties (NSORT), cumulative flying hours squared (CUMFH2), number of sorties squared (NSORT2), and cumulative flying hours times number of sorties (FHNSORT or CUMFHNSORT). The files generated by the FORTTRAN program, representing each work unit code level, were transferred into Statistix format and used to perform the regression and Poisson process analysis in the study. The SAS programs used to merge and sort the 1993 and 1994 data are included in Appendix B. The FORTTRAN programs are included in Appendix C, along with a sample of 1993 and 1994 data output files.

Population and Sample Size

The 1993 and 1994 maintenance and sortie databases contain information on a single type of aircraft, the F-15C. The 1993 maintenance database totals 12,989 individual entries. The 1993 sortie database contains 45,770 entries. After deleting the 160 aircraft that had no corresponding maintenance history in the 1993 sortie database, the resulting sortie/maintenance action, merged file still contains 32,240 entries for the eight month time period.. The 1994 maintenance database contains 25,177 individual entries. The 1994 sortie database contains 23,079 entries for the five month period. Only failures of reparable spares are considered within the data.

Data Limitations

The data is limited in scope because it only includes one type of fighter aircraft, the F-15C for the periods May to December 1993 and February to June 1994. In analyzing the data, regression and Poisson process analyses will be run against the criterion variable,

demands/maintenance actions, and the predictor variables, cumulative flying hours and number of sorties, created for each work unit code level. However, the number of regressions will be limited due to the frequency of individual work unit code occurrences. Initial sorting of the data files indicated several work unit codes show only a small number of occurrences within the data sets, which implies low demand rates. Meaningful regression results cannot be achieved by regressing on only two or three points. The lack of significant numbers of specific work unit code occurrences will limit the work unit codes available for development of work unit code decision rules.

Despite the small number of occurrences for specific work unit codes, some of the individual work unit codes have larger number of occurrences. In obtaining a representative sample across all work unit codes, only those work units code levels with more than 30 occurrences will be evaluated. The 30 occurrences limit was established to ensure there were enough points in the regression and Poisson process analysis to provide meaningful results. Performing a regression or Poisson process analysis on only a few points (less than 30) would probably not produce meaningful results. A solution to the problem of limited occurrences at the four and five digit work unit code levels is to aggregate the work unit codes at the two or three digit level.

Also, the number of predictor variables is limited within the data. The available predictor variables are cumulative flying hours and number of sorties. However, additional carriers can be added to full regression models by squaring or linearly combining these predictor variables. Transformations of the data may also be necessary due to a lack of correlation (small R^2 values).

Because the timeframe covered by the data is discrete, the first maintenance action obtained on each aircraft for specific work unit codes will only accumulate flying hours and sorties from the starting point of the data set. Based on this limitation, the total number of maintenance actions experienced by each aircraft is calculated for the

cumulative flying hours and number of sorties flown during the discrete timeframe covered by the data set. A more accurate analysis could possibly consider time between maintenance actions, however, the limited number of occurrences of specific work unit codes precludes this type of analysis.

Research Design Implementation

The research design is implemented in three phases. Phase One covers multiple regression. Phase Two deals with Poisson regression, while Phase Three relates use of a Poisson process. In each phase, the data files obtained from the FORTRAN programs are used to obtain points for the multiple or Poisson regressions or parameters for the Poisson process. The three research design phases are intended to answer the first research question: "Is there a relationship between demands/maintenance actions, flying hours, and number of sorties at the work unit code level?" The steps of each research design phase will now be presented.

Phase One: Multiple Regression. In building a multiple regression model, a four stage process will be followed:

Stage One: Data collection and preparation

Stage Two: Reduction of the number of independent variables

Stage Three: Model refinement and selection

Stage Four: Model Validation (Neter and Wasserman, 1985:433)

To implement the four stage model building process, the following steps will be used:

Step 1: Obtain FORTRAN files from the 1993 data set for each specific two, three, four, and five digit work unit code level. Transfer the files into Statistix format in order to perform multiple regression analysis.

Step 2: Establish the criterion and predictor variables from the data. The criterion variable is the number of demands or maintenance actions experienced by each aircraft. The predictor variables are the cumulative number of flying hours and number of sorties flown by each aircraft. Each aircraft tail number in the data set represents a point on the regression plane.

Step 3a: Using Statistix Analytical software, perform multiple linear regression on the first order, “reduced” regression model. Use the test for model utility and *f*-statistic/ F-Distribution procedures to test the following hypotheses for each specific two, three, four, and five digit work unit code level:

$$H_o : \beta_1 = \beta_2 = \dots \beta_k = 0$$

$$H_a : \text{at least one } \beta_i \neq 0 \ (i = 1, \dots, k)$$

If H_a is true, use p-value comparisons to an $\alpha = 0.05$ to determine which variables contribute to the reduced model.

Step 3b: Perform multiple regression in Statistix on the “full,” second order, interaction model. Use *f*-statistic/F-Distribution procedures to test the following hypotheses:

$$H_o : \text{model is } Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \epsilon \text{ (reduced model)}$$

$$H_a : \text{model is } Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \beta_3 x_1^2 + \beta_4 x_2^2 + \beta_5 x_1 x_2 + \epsilon \text{ (full model)}$$

If H_a is true, use p-value comparisons to an $\alpha = 0.05$ to determine which variables contribute to the full model.

Step 4: To validate the model, a residual will be calculated from the 1994 data set. Using the reduced or full model obtained from the 1993 data, substitute 60,254.5 flying hours for x_1 , the cumulative flying hours, and 38,666 sorties for x_2 , the number of sorties. The 1994 data set has a total of 60,254.5 flying hours and 38,666 sorties. The value of *Y* obtained from the 1993 regression model, using the 1994 flying hour and sortie values, is the expected number of maintenance actions for 60,254.5 flying hours and

38,666 sorties. Compare the expected value of 1994 maintenance actions to the actual number of 1994 maintenance actions. The actual value minus the expected value is the residual.

Phase Two: Poisson Regression. Phase two of the research design implementation involves Poisson regression. Poisson regression will be performed according to the following steps:

Step 1: Obtain FORTRAN files from the 1993 data set for each specific two, three, four, and five digit work unit code level. Transfer the files into Statistix format in order to perform Poisson regression analysis.

Step 2: Perform three separate Poisson regressions on the two digit level work unit code files. The first regression will only include cumulative flying hours as a predictor variable. The second regression will only include number of sorties as the predictor variable. The final regression will include both cumulative flying hours and number of sorties as predictor variables.

Step 3: Compare deviance values and select the model with the smallest deviance. The smaller the value of the deviance, the better the fit of the Poisson regression model. Use p-value comparisons to an $\alpha = 0.05$ to determine which variables contribute to the full model.

Step 4: Similar to the multiple regression analysis, calculate a residual using the model with the smallest deviance and the 1994 cumulative flying hour and sortie values of 60,254.5 and 38,666, respectively.

Phase Three: Poisson Process. Phase three of the research design implementation fits a Poisson process to the data. The Poisson process will be fit according to the following steps:

Step 1: Calculate the λ values for the Poisson process from the 1993 data set on the basis of flying hours and sorties. λ will take the following form: the cumulative 1993

maintenance actions at a specific two digit work unit code level divided by either cumulative flying hours or total number of sorties.

Step 2: Use the 1993 λ values from Step 1 to calculate the expected number of maintenance actions for the 1994 data set, based on flying hours and sorties. (Multiply the 1993 flying hour or sortie based λ values by 60,254.5 flying hours or 38,666 sorties).

Step 3: Calculate a $\pm 2\sigma$ confidence interval for the Poisson process, based on flying hours or sorties. Note: If X is a random variable and has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$. Thus, $\sigma = \sqrt{\lambda}$.

Step 4: Calculate the probability that the number of maintenance actions is below the lower bound of the confidence interval or above the upper bound of the confidence interval for both the flying hour and sortie based models.

Step 5: Compare the actual number of 1994 maintenance actions, for each specific two digit work unit code level, with the expected values calculated from Step 1 above to obtain a residual based on flying hours and sorties.

Step 6: Use the two-sample, two sided F test for equal population variances to perform the hypotheses test $H_0: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$.

The residual results from the three phases of the research design implementation will also be analyzed/compared to determine which method produces the best models and exhibits a relationship between demands/maintenance actions, flying hours, and sorties at the work unit code level. Upon analyzing the data by multiple regression, Poisson regression, and fitting a Poisson process, research question two: "Can decision rules be established to forecast demands/maintenance actions based on a spares work unit code alone?" will be answered based on the best regression model or Poisson process estimation obtained from the resulting three phase residual analysis.

Summary

This chapter outlines the research methodology of this study. The chapter also describes the research design, research questions, research hypotheses and instruments, and the process used for variable validation, as well as, the data collection, data gathering, population and sample size, and limitations of the data. The final section of the chapter discusses the implementation of the research design. The research design and methodology are based on multiple linear regression, Poisson regression, and fitting of a Poisson process, which are used to determine if a relationship exists between demands/maintenance actions, cumulative flying hours, and number of sorties at the work unit code level. The implementation of the research design attempts to develop a best regression model or Poisson process estimation technique to align demands, or maintenance actions, flying hours, and sorties at the work unit code level. The next chapter presents the results, and analysis of these results, which were obtained from implementing this research methodology and design.

IV. Results and Analysis

Introduction

This chapter presents the results obtained from implementing the three phases of the research methodology. The results obtained from multiple regression, Poisson regression, and fitting of a Poisson process will be initially reviewed and discussed. Following the review of the research results, an analysis/comparison of residuals obtained from multiple regression, Poisson regression, and fitting of a Poisson process is also presented to determine which method is most suitable in explaining any relationship between demands, flying hours, and number of sorties at the work unit code level.

Multiple Regression Results

Multiple regressions were performed against specific two, three, and five digit levels of the work unit code. The four digit level of the work unit code was not analyzed due to the close similarity with results obtained at the five digit work unit code level. Multiple regressions were performed on 24 different two digit work unit code levels, 74 different three digit work unit code levels, and 115 different five digit work unit code levels. Tables D-1, D-2, and D-3 in Appendix D list the work unit code levels along with the coefficient of multiple determination, R^2 , values and Durbin-Watson statistics obtained from the multiple regression analysis.

A majority of the multiple regression models produced very small R^2 values. The small R^2 values are believed to be caused by a lack of data points at lower levels of the work unit code. For example, at the two digit level, all work unit codes beginning with the same first two digits are used in the regression. However, at the five digit level, only those work unit codes with exactly the same five digits are used. The result is more data points at the two digit versus the five digit level of the work unit code.

Due to the lack of regression points and any significant correlation, only those two digit work unit code levels with an R^2 approximately equal to or greater than 0.200 were analyzed by hypotheses and residual testing. The two digit work unit codes which met this criterion are: 13, 23, 63, 74, and 76. Table 4-1 lists the results of the multiple regression analysis and hypotheses/residual testing.

Table 4-1. Multiple Regression Results

| Two-Digit WUC Level | R^2 | Reduced Model Hypoth. Test | Result | Reduced vs. Full Model Hypotheses Test | Result | Actual # of 1994 Maint. Actions | Expected # of Maint. Actions | Residual; Actual - Expected |
|---------------------|--------|---|---------------------------------|---|-------------------------------------|---------------------------------|------------------------------|-----------------------------|
| 13 | 0.2901 | $H_0: \beta_i = 0$ $H_a: \beta_i \neq 0$ | Enough evidence to reject H_0 | H_0 : Reduced Model H_a : Full Model | Not enough evidence to reject H_0 | 2,092 | 1,321 | 771 |
| 23 | 0.2134 | $H_0: \beta_i = 0$ $H_a: \beta_i \neq 0$ | Enough evidence to reject H_0 | H_0 : Reduced Model H_a : Full Model | Enough evidence to reject H_0 | 2,069 | Not Calculated | Not Calculated |
| 63 | 0.1999 | $H_0: \beta_i = 0$ $H_a: \beta_i \neq 0$ | Enough evidence to reject H_0 | H_0 : Reduced Model H_a : Full Model | Not enough evidence to reject H_0 | 1,057 | 976 | 81 |
| 74 | 0.3492 | $H_0: \beta_i = 0$ $H_a: \beta_i \neq 0$ | Enough evidence to reject H_0 | H_0 : Reduced Model H_a : Full Model | Not enough evidence to reject H_0 | 3,710 | 3,262 | 448 |
| 76 | 0.2522 | $H_0: \beta_i = 0$ $H_a: \beta_i \neq 0$ | Enough evidence to reject H_0 | H_0 : Reduced Model H_a : Full Model | Not enough evidence to reject H_0 | 1,842 | 1,083 | 759 |

For two-digit work unit code level 23, the expected number of maintenance actions and residual were not calculated due to the outcome of the reduced versus full hypotheses test. In this test, enough evidence existed to accept the full model. However, none of the variables in the full model passed the p-value comparison to $\alpha = 0.05$. Therefore, no predictive model was available for two digit work unit code level 23. Detailed

calculations of the multiple regression hypotheses testing and residual analysis are included in Appendix D.

Although multiple regression models were built and analyzed at the two digit work unit code level, the small R^2 values and the large residuals indicate a limited linear relationship exists between the criterion variable, demands/maintenance actions, and the predictor variables, cumulative flying hours, and number of sorties at the work unit code level. The lack of correlation is also portrayed in the multiple regression residual plots. Figure 4-1 shows a residual plot for two digit work unit code level 76. The residual plot appears to be stratified into layers because the demands/maintenance actions, used as the criterion variable in the multiple regression, are integer numbers representing the numbered counts of demands/maintenance actions.

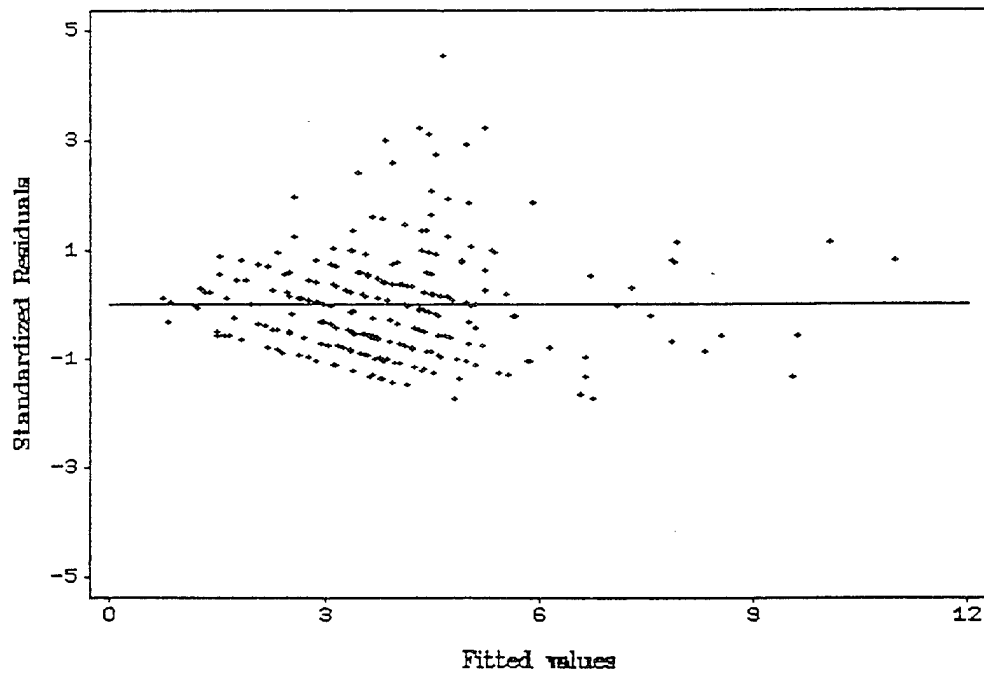


Figure 4-1. Residual Plot for Two Digit Work Unit Code Level 76, Electronic Countermeasures Components

To account for the “count” nature of the data represented by the criterion variable, demands/maintenance actions, Poisson regression was attempted to improve upon the multiple regression results. The results of the Poisson regression analysis are presented in the next section.

Poisson Regression Results

Three Poisson regressions were performed against the 24 different, two digit work unit code levels. The first Poisson regression included only cumulative flying hours (CUMFH) as a predictor variable, while the second Poisson regression included only the number of sorties (NSORT) as the predictor variable. The third Poisson regression contained both cumulative flying hours and number of sorties as predictor variables. The results of the Poisson regressions and residual analysis are presented in Table 4-2 on the next page. The Poisson regression calculations are included in Appendix E.

The measure of fit for a Poisson regression model is known as the deviance. The deviance is considered the “distance measure” between the fitted Poisson regression model and the actual data. The smaller the value of the deviance, the better the fit of the model (Statistix, 1985:183). As can be seen from Table 4-2, none of the three Poisson regressions models built for any of the two digit work unit code levels obtained small deviance values. For purposes of this study, a small deviance is considered 50 or less. Nevertheless, a residual was calculated for the Poisson regression model having the smallest deviance value for each two digit work unit code level.

Poisson regression is designed to handle data of a count nature. However, Poisson regression produced no better results than obtained through multiple linear regression. In reviewing simple scatter plots of the data, the absence of a linear relation is thought to be caused by a lack of data structure. Figures 4-2 and 4-3 on page 4-6 show two scatter plots of the study data, which exhibit the lack of data structure. Each point on

Table 4-2. Poisson Regression Results

| 2-Digit Work Unit Code Level | CUMFH Model Deviance | NSORT Model Deviance | CUMFH & NSORT Model Deviance | # of Actual 1994 Maint. Actions | # of Expected Maint. Actions | Residual; Actual - Expected |
|------------------------------------|----------------------------|----------------------------|---------------------------------------|---------------------------------------|---------------------------------------|-----------------------------------|
| 11 | 412.07 | 396.26 | 392.74 | 794 | 396 | 398 |
| 12 | 340.55 | 312.95 | 302.94 | 318 | 441 | -123 |
| 13 | 658.90 | 537.38 | 482.11 | 2,092 | 265 | 1,827 |
| 14 | 493.65 | 474.59 | 457.61 | 1,121 | 213 | 908 |
| 23 | 557.00 | 470.06 | 461.35 | 2,069 | 323 | 1,746 |
| 24 | 572.59 | 542.59 | 534.23 | 1,114 | 253 | 861 |
| 41 | 450.63 | 438.33 | 438.23 | 667 | 268 | 399 |
| 42 | 507.07 | 486.57 | 485.37 | 766 | 367 | 399 |
| 44 | 464.08 | 421.76 | 407.13 | 665 | 292 | 373 |
| 45 | 350.79 | 331.30 | 329.95 | 965 | 453 | 512 |
| 46 | 360.49 | 359.30 | 350.02 | 808 | 393 | 415 |
| 47 | 314.66 | 289.90 | 286.50 | 280 | 635 | -355 |
| 49 | 196.69 | 191.46 | 190.81 | 118 | 473 | -355 |
| 51 | 456.33 | 417.93 | 417.74 | 1,043 | 300 | 743 |
| 52 | 431.70 | 425.82 | 425.81 | 284 | 239 | 45 |
| 55 | 513.65 | 502.07 | 492.14 | 502 | 201 | 301 |
| 57 | 223.38 | 221.77 | 202.70 | 265 | 53 | 212 |
| 63 | 526.33 | 449.66 | 440.93 | 1,057 | 352 | 705 |
| 65 | 542.55 | 535.90 | 530.10 | 771 | 267 | 504 |
| 71 | 546.29 | 512.32 | 512.32 | 1,049 | 314 | 735 |
| 74 | 840.81 | 628.75 | 619.26 | 3,710 | 332 | 3,378 |
| 75 | 566.99 | 542.15 | 518.77 | 725 | 200 | 525 |
| 76 | 489.07 | 528.86 | 483.65 | 1,842 | 274 | 1,568 |
| 97 | 482.82 | 477.31 | 467.16 | 476 | 143 | 333 |

the scatter plots represents one aircraft from the 1993 data set, which flew a specified quantity of flying hours and number of sorties, while producing a given number of demands, or maintenance actions. The pattern exhibited on the scatter plots is similar to the stratified effect produced on the multiple regression residual plots (See Figure 4-1).

With data structured in this manner, the fitting of a linear relation is difficult at best. For example, consider a specified number of flying hours or sorties on either Figure 4-2 or 4-3. For any value of flying hours or sorties, the number of demands (or

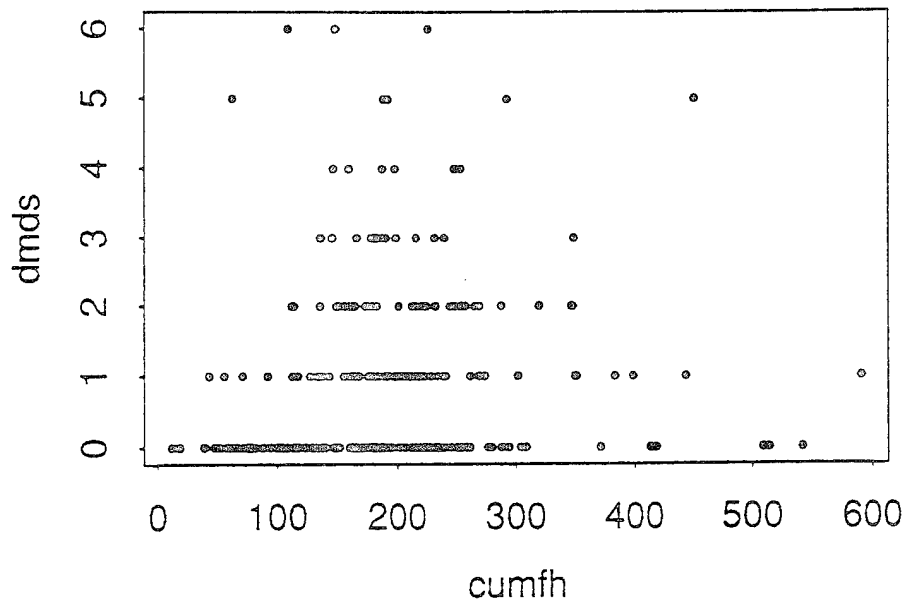


Figure 4-2. Scatter Plot of Demands versus Cumulative Flying Hours for Two Digit Work Unit Code Level 11, Airframe

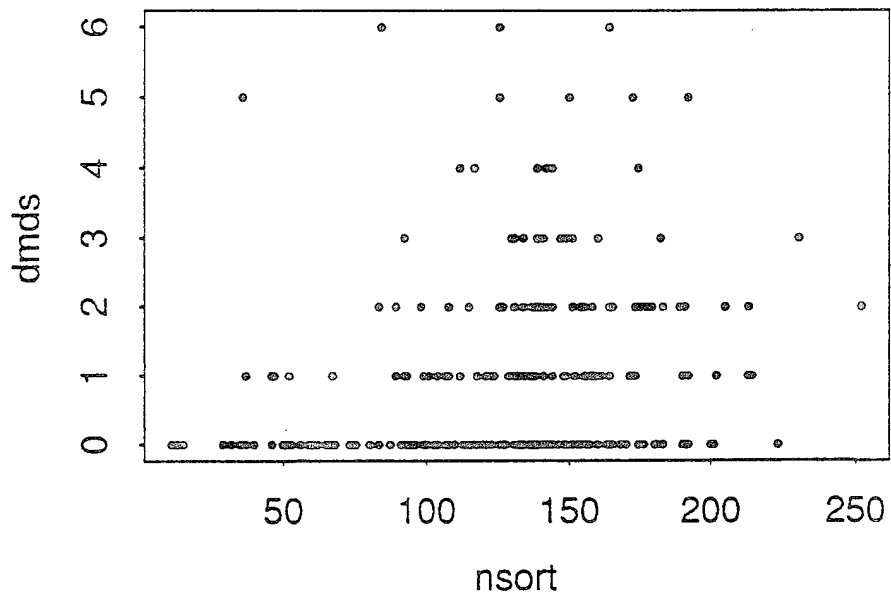


Figure 4-3. Scatter Plot of Demands versus Number of Sorties for Two Digit Work Unit Code Level 11, Airframe

maintenance actions) can range between 0 and 6. Therefore, attempting to fit a linear model produces small R^2 values in multiple linear regression and large deviance values with Poisson regression.

Based on the lack of correlation obtained with either multiple or Poisson regression, the data in the study were analyzed by fitting a Poisson process in an attempt to determine a relation between demands, or maintenance actions, cumulative flying hours, and number of sorties at the work unit code level. The results of the Poisson process analysis are covered in the next section.

Poisson Process Results

The Poisson process analysis was performed against each two digit work unit code level. The results of this analysis are contained in Tables 4-3 and 4-4, which are presented on the next two pages. Table 4-3 covers results for flying hour based lambda's while Table 4-4 covers results for sortie based lambda's. Detailed calculations for the Poisson process are included in Appendix F.

The Poisson process lambdas were calculated on the total demands, or maintenance actions, at each two digit work unit code level for all 247 aircraft in the 1993 data set. This process is a slightly different method than the regression procedures which considered each aircraft in the data set as a point for the regression. The probabilities calculated in Tables 4-3 and 4-4 show there is between a 4 and 5 percent probability that a value will fall outside the bounds of the computed confidence intervals. However, in 23 of the 24 cases, for both the flying hour and sortie based Poisson processes, the actual 1994 value fell outside the bounds of the computed confidence interval. Thus, only one two digit work unit code level, level 65, IFF, falls within the computed $\pm 2\sigma$ confidence interval. Also, the Poisson process residuals comparing actual to expected numbers of demands/maintenance actions are rather large. The confidence interval results and large

residual values indicate that a Poisson process may not explain the relationship between demands/maintenance actions, flying hours, and number of sorties at the work unit code level.

Table 4-3. Poisson Process Results (Flying Hour Based)

| Two-Digit WUC Level | 1993 Flying Hour Based Lambda | Expected Number of Maint. Actions | +/- 2 σ Confidence Interval; (Lower bound, Upper Bound) | Probability ($X \leq$ Lower bound or $X \geq$ Upper bound) | Actual # of 1994 Maint. Actions | Residual; Actual - Expected |
|---------------------|-------------------------------|-----------------------------------|--|---|---------------------------------|-----------------------------|
| 11 | 0.0049 | 295 | (260,329) | 0.04463 | 794 | 499 |
| 12 | 0.0028 | 169 | (143,195) | 0.04539 | 318 | 149 |
| 13 | 0.0323 | 1,946 | (1,858,2,034) | 0.04606 | 2,092 | 146 |
| 14 | 0.0065 | 392 | (352,431) | 0.04588 | 1,121 | 729 |
| 23 | 0.0212 | 1,277 | (1,206,1,349) | 0.04544 | 2,069 | 792 |
| 24 | 0.0097 | 584 | (536,663) | 0.04482 | 1,114 | 530 |
| 41 | 0.0062 | 374 | (335,412) | 0.04632 | 667 | 293 |
| 42 | 0.0062 | 374 | (335,412) | 0.04632 | 766 | 392 |
| 44 | 0.0098 | 590 | (542,639) | 0.04592 | 665 | 75 |
| 45 | 0.0038 | 229 | (199,259) | 0.04736 | 965 | 736 |
| 46 | 0.0058 | 349 | (312,387) | 0.04481 | 808 | 459 |
| 47 | 0.0028 | 169 | (143,195) | 0.04539 | 280 | 111 |
| 49 | 0.0010 | 60 | (45,76) | 0.04597 | 118 | 58 |
| 51 | 0.0155 | 934 | (873,995) | 0.04592 | 1,043 | 109 |
| 52 | 0.0037 | 223 | (193,253) | 0.04446 | 284 | 61 |
| 55 | 0.0041 | 247 | (216,278) | 0.0485 | 502 | 255 |
| 57 | 0.0013 | 78 | (61,96) | 0.04801 | 265 | 187 |
| 63 | 0.0155 | 934 | (873,995) | 0.04592 | 1,057 | 123 |
| 65 | 0.0132 | 795 | (739, 852) | 0.04513 | 771 | -24 |
| 71 | 0.0125 | 753 | (698,808) | 0.04503 | 1,049 | 296 |
| 74 | 0.0547 | 3,296 | (3,181,3,411) | 0.04516 | 3,710 | 414 |
| 75 | 0.0096 | 578 | (530,627) | 0.04372 | 725 | 147 |
| 76 | 0.0201 | 1,211 | (1,142,1,281) | 0.04584 | 1,842 | 631 |
| 97 | 0.0036 | 216 | (187,246) | 0.04505 | 476 | 260 |

Table 4-4. Poisson Process Results (Sortie Based)

| Two-Digit WUC Level | 1993 Sortie Based Lambda | Expected Number of Maint. Actions | +/- 2 σ Confidence Interval; (Lower bound, Upper Bound) | Probability ($X \leq$ Lower bound or $X \geq$ Upper bound) | Actual # of 1994 Maint. Actions | Residual; Actual - Expected |
|---------------------|--------------------------|-----------------------------------|--|---|---------------------------------|-----------------------------|
| 11 | 0.0074 | 286 | (252,320) | 0.04435 | 794 | 508 |
| 12 | 0.0042 | 162 | (137,188) | 0.04534 | 318 | 156 |
| 13 | 0.0487 | 1,883 | (1,796,1,970) | 0.04497 | 2,092 | 209 |
| 14 | 0.0098 | 379 | (340,418) | 0.0451 | 1,121 | 742 |
| 23 | 0.0319 | 1,233 | (1,163,1,304) | 0.0447 | 2,069 | 836 |
| 24 | 0.0147 | 568 | (521,616) | 0.04632 | 1,114 | 546 |
| 41 | 0.0094 | 363 | (325,402) | 0.0434 | 667 | 304 |
| 42 | 0.0094 | 363 | (325,402) | 0.0434 | 766 | 403 |
| 44 | 0.0148 | 572 | (524,620) | 0.04592 | 665 | 93 |
| 45 | 0.0058 | 224 | (194,254) | 0.04503 | 965 | 741 |
| 46 | 0.0088 | 340 | (303,377) | 0.04479 | 808 | 468 |
| 47 | 0.0041 | 158 | (133,184) | 0.04272 | 280 | 122 |
| 49 | 0.0015 | 58 | (43,73) | 0.04861 | 118 | 60 |
| 51 | 0.0233 | 901 | (841,961) | 0.0456 | 1,043 | 142 |
| 52 | 0.0056 | 217 | (187,246) | 0.04491 | 284 | 67 |
| 55 | 0.0061 | 236 | (205,267) | 0.04352 | 502 | 266 |
| 57 | 0.0019 | 73 | (56,91) | 0.04097 | 265 | 192 |
| 63 | 0.0233 | 901 | (841,961) | 0.0456 | 1,057 | 106 |
| 65 | 0.0199 | 769 | (714, 825) | 0.0454 | 771 | 2 |
| 71 | 0.0189 | 731 | (677,785) | 0.04578 | 1,049 | 318 |
| 74 | 0.0824 | 3,186 | (3,073,3,299) | 0.04529 | 3,710 | 524 |
| 75 | 0.0145 | 561 | (513,608) | 0.04481 | 725 | 164 |
| 76 | 0.0304 | 1,175 | (1,107,1,244) | 0.04571 | 1,842 | 667 |
| 97 | 0.0054 | 209 | (180,238) | 0.04477 | 476 | 267 |

A condition of fitting a Poisson process is that the conditions between different time periods must be the same. Otherwise, the main use of the Poisson process becomes descriptive and not prescriptive. Two formal hypotheses tests were conducted for each two digit work unit code level, one based on flying hours and the other based on sorties. These hypotheses tests determine whether the demand rates are the same or different for the 1993 and 1994 data sets used in this study. The results of these hypotheses tests are

included in Table 4-5 below. Detailed calculations for the hypotheses testing are included in Appendix F, starting on page F-27.

Table 4-5. Results of Poisson Process Hypotheses Testing

| Two Digit Work Unit Code Level | Flying Hour Based Hypotheses Test | Result | Sortie Based Hypotheses Test | Result |
|--------------------------------|---|--------------|---|--------------|
| 11 | $H_o: \lambda_{93} = \lambda_{94}$ $H_a: \lambda_{93} \neq \lambda_{94}$ | Reject H_o | $H_o: \lambda_{93} = \lambda_{94}$ $H_a: \lambda_{93} \neq \lambda_{94}$ | Reject H_o |
| 12 | “ “ | Reject H_o | “ “ | Reject H_o |
| 13 | “ “ | Reject H_o | “ “ | Reject H_o |
| 14 | “ “ | Reject H_o | “ “ | Reject H_o |
| 23 | “ “ | Reject H_o | “ “ | Reject H_o |
| 24 | “ “ | Reject H_o | “ “ | Reject H_o |
| 41 | “ “ | Reject H_o | “ “ | Reject H_o |
| 42 | “ “ | Reject H_o | “ “ | Reject H_o |
| 44 | “ “ | Reject H_o | “ “ | Reject H_o |
| 45 | “ “ | Reject H_o | “ “ | Reject H_o |
| 46 | “ “ | Reject H_o | “ “ | Reject H_o |
| 47 | “ “ | Reject H_o | “ “ | Reject H_o |
| 49 | “ “ | Reject H_o | “ “ | Reject H_o |
| 51 | “ “ | Reject H_o | “ “ | Reject H_o |
| 52 | “ “ | Reject H_o | “ “ | Reject H_o |
| 55 | “ “ | Reject H_o | “ “ | Reject H_o |
| 57 | “ “ | Reject H_o | “ “ | Reject H_o |
| 63 | “ “ | Reject H_o | “ “ | Reject H_o |
| 65 | “ “ | Accept H_o | “ “ | Accept H_o |
| 71 | “ “ | Reject H_o | “ “ | Reject H_o |
| 74 | “ “ | Reject H_o | “ “ | Reject H_o |
| 75 | “ “ | Reject H_o | “ “ | Reject H_o |
| 76 | “ “ | Reject H_o | “ “ | Reject H_o |
| 97 | $H_o: \lambda_{93} = \lambda_{94}$ $H_a: \lambda_{93} \neq \lambda_{94}$ | Reject H_o | $H_o: \lambda_{93} = \lambda_{94}$ $H_a: \lambda_{93} \neq \lambda_{94}$ | Reject H_o |

The results shown in Table 4-5 indicate that the rate of demand or maintenance action occurrence changed between 1993 and 1994, which explains why only one of the two digit work unit levels, level 65, IFF, fell within the computed confidence interval. Level 65 was

also the only two digit work unit code level which accepted the null hypothesis indicating the demand rates were the same between 1993 and 1994. Based on the data used in this study, this hypothesis testing shows that under a fitted Poisson process, 1994 demands cannot be accurately predicted from 1993 demand rates at the work unit code level.

To analyze which phase of the methodology produces the best model or estimation technique at the work unit code level, the following section compares the residual values from obtained from multiple regression, Poisson regression, and fitting of a Poisson process.

Analysis/Comparison of Residuals

Table 4-6 on the next page compares the residuals obtained from the Poisson regression and Poisson Process calculations. The multiple regression residual values are not included in the comparison analysis because residual values were only calculated for 4 of the 24 two digit work unit code levels, due to the poor R^2 values. The average residual value for the Poisson regression analysis is 668.92 ($16,054/24 = 668.92$). The Poisson process (flying hour based) average residual is 309.5 ($7,428/24 = 309.5$), while the Poisson process (sortie based) average residual is 329.29 ($7,903/24 = 329.29$).

Based on the comparison analysis of the Poisson regression and Poisson process residual and overall average residual values, the Poisson process provides a better fit to the data than the Poisson regression technique. The four multiple regression residual values calculated are also large in comparison to the Poisson process residuals. Although the Poisson process does provide better residuals than multiple or Poisson regression, the expected number of 1994 maintenance actions calculated from the Poisson process were still significantly different from the actual 1994 values. Also, hypotheses testing under the fitted Poisson process indicates demand rates between 1993 and 1994 were different, which reduced the estimation capability of any Poisson process models. Thus, as observed

with multiple and Poisson regression, a Poisson process did not prove to be a good estimator of expected demands or maintenance actions at the work unit code level.

Table 4-6. Comparison of Residuals

| Two-Digit Work Unit Code Level | Poisson Regression Residuals | Poisson Process Residuals (Flying Hour based) | Poisson Process Residuals (Sortie Based) |
|-----------------------------------|---------------------------------|---|--|
| 11 | 398 | 499 | 508 |
| 12 | -123 | 149 | 156 |
| 13 | 1,827 | 146 | 209 |
| 14 | 908 | 729 | 742 |
| 23 | 1,746 | 792 | 836 |
| 24 | 861 | 530 | 546 |
| 41 | 399 | 293 | 304 |
| 42 | 399 | 392 | 403 |
| 44 | 373 | 75 | 93 |
| 45 | 512 | 736 | 741 |
| 46 | 415 | 459 | 468 |
| 47 | -355 | 111 | 122 |
| 49 | -355 | 58 | 60 |
| 51 | 743 | 109 | 142 |
| 52 | 45 | 61 | 67 |
| 55 | 301 | 255 | 266 |
| 57 | 212 | 187 | 192 |
| 63 | 705 | 123 | 106 |
| 65 | 504 | -24 | 2 |
| 71 | 735 | 296 | 318 |
| 74 | 3,378 | 414 | 524 |
| 75 | 525 | 147 | 164 |
| 76 | 1,568 | 631 | 667 |
| 97 | 333 | 260 | 267 |

Summary

This chapter covered the results obtained from the three phases of the research methodology, which are multiple regression, Poisson regression, and a Poisson process. Current Air Force requirements computation programs use only flying hours to forecast demand. The intent of this study was to determine if a relationship exists between

demands, cumulative flying hours, and number of sorties at the work unit code level.

Despite evaluating the data from three different angles, a significant relationship between demands/maintenance actions, cumulative flying hours, and number of sorties could not be found at the work unit code level. The next chapter presents the conclusions and recommendations of this research study.

V. Conclusions and Recommendations

Introduction

The purpose of this chapter is to present conclusions and recommendations obtained from the research. Initially, the specific problem, purpose of the study, and research questions will be presented. Following this initial discussion, the results, conclusions, and important management implications obtained for each research question will be discussed. Recommendations for follow-on research will then be presented. To conclude the chapter, a research summary will be provided.

Specific Problem

Current Air Force requirements computation systems, for example, D041 and REALM, forecast peacetime and MRSP reparable requirements based solely on a flying hour basis. The specific problem is to determine whether demands or maintenance actions of reparable spare parts are correlated to operational characteristics of the weapon system, specifically flying hours and sorties. Because current requirements models assume only a direct, linear relationship to the number of flying hours, demands/maintenance actions could be driven by other factors, or a combination of factors, to include flying hours and/or number of sorties.

Purpose of the Study

The primary purpose of this study is to determine whether or not a relationship exists between reparable spares demands/maintenance actions, flying hours, and number of sorties at the work unit code level. A secondary purpose is to develop models and decision rules based on the existing demands, flying hours, sortie relationship, which can be used to improve forecasting of reparable peacetime and MRSP spare requirements.

Research Questions

To evaluate the extent of correlation between demands/maintenance actions, flying hours, and number of sorties at the work unit code level, the following research questions are developed:

1. Is there a relationship between demands/maintenance actions, flying hours, and number of sorties at the work unit code level?
2. Can decision rules be established to forecast demands/maintenance actions based on a spares work unit code alone?

The results, conclusions, and management implications for each research question will now be presented.

Research Question One: Results, Conclusions, and Management Implications

The following section discusses the results from the multiple regression, Poisson regression, and Poisson process analyses used to answer research question one, "Is there a relationship between demands/maintenance actions, flying hours, and number of sorties at the work unit code level?" Conclusions and important Air Force management implications, which could be derived from these results, are also covered.

Results. The multiple regression, Poisson regression, and Poisson process results show that there is a limited relationship between demands/maintenance actions, flying hours, and number of sorties at the work unit code level. The low R^2 values obtained with multiple regression, the large deviance values obtained from Poisson regression, and the single occurrence of meeting the bounds of a calculated Poisson process confidence interval indicate the data used in this study do not support a relationship between demands/maintenance actions, flying hours, and number of sorties at the work unit code level. Also, a strong relationship between demands/maintenance actions, flying hours, and

number of sorties was not found at any two, three, or five digit work unit code level analyzed.

The answer to investigative question one, "Are demands or maintenance actions, and number of sorties correlated at the two digit level of specific work unit codes?", is similar to the answer for research question one. Although demands/maintenance actions were aggregated at the two digit work unit code level, to increase the number of points available for analysis, there was limited correlation obtained at the two digit work unit code level by using either multiple regression, Poisson regression, or fitting of a Poisson process.

As for investigative question two, "Do demands or maintenance actions, flying hours, and number of sorties show more, or less, correlation at the three, four, or five digit level of specific work unit codes, as compared to the two digit level?", the answer is that there is less correlation at the three, four, or five digit level of the work unit code in comparison to the two digit level. The correlation obtained at the two digit work unit code level is limited. However, by moving to lower, more defined, levels of the work unit code, the number of positive, regression points available for analysis decreases. This decrease is due to the number of zero values for demands against specific work unit codes increasing by moving to lower levels of the work unit code.

At the five digit work unit code level, a majority of the aircraft may not experience a demand/maintenance action for the specific five digit work unit code being analyzed. If the aircraft experiences no demands/maintenance actions for this five digit work unit code, the criterion variable, demands/maintenance actions, enters the regression with a value of zero. As the number of zero values increases in the regression, the extent of correlation decreases.

Poisson regression analysis also exhibits a similar lack of correlation based on the large deviance values. Any models subsequently developed from either the multiple or

Poisson regression analysis exhibit poor predictive capability when used to estimate demands or maintenance actions based on a 1994 validation data set. The large residual values obtained from residual analysis on the multiple and Poisson regression models indicate the models perform poorly in estimating expected numbers of demands/maintenance actions.

Results obtained by fitting a Poisson process were similar to the results obtained from multiple and Poisson regression. The demand estimates and confidence intervals calculated with the Poisson process for the 1994 validation data set were normally much lower than the actual number of 1994 demands experienced. Also, only 1 of 24 two digit work unit code levels evaluated fell within the calculated confidence interval.

Also, by fitting a Poisson process to various data sets, similar conditions must be maintained between data sets. In other words, the lambda values, or rates of occurrence, must stay relatively constant between time periods. However, the lambda values calculated for the 1993 data set were much lower than the lambda values for 1994. Thus, the demand estimates and confidence intervals calculated from the 1993 lambda values were lower than the actual number of 1994 demands.

The Poisson process analysis further exhibits that the erratic nature of demands makes forecasting future demands at the work unit code level a difficult process. The calculated Poisson process work unit code level demand/maintenance action estimates, which are based on flying hours and sorties from past demands, are poor estimators of future demands/maintenance actions.

Conclusions. A conclusion is reached that aligning demands/maintenance actions with their underlying failure modes remains a complicated issue, despite analysis at the work unit code level. However, this research is unique in that it targets reparable demand forecasting at the work unit code level. Another conclusion the research supports is the limited correlation obtained across the F-15C weapon system at the work unit code level.

This limited correlation between demands/maintenance actions, flying hours, and number of sorties at the work unit code level supports previous demand forecasting research in that a significant relationship could not be determined to accurately predict reparable demands.

Management Implications. The current Air Force requirements computation systems calculate reparable requirements based solely on flying hours. However, some reparable spares may not fail on a strictly flying hour basis. Thus, based on the current Air Force systems, some flying hour driven spares are stocked with accurate quantities, while other non-flying hour driven spares are either over or under stocked. This study attempted to identify a relationship between demands, flying hours, and number of sorties at the work unit code level and align demands/maintenance actions with accurate spares requirements.

Despite the limited correlation obtained at the work unit code level, this research is important and has significant Air Force management implications. First, further research in the area of reparable demand forecasting is still required because an accurate, reliable relationship between demands, flying hours, and sorties could not be determined in this study. Second, this research supports the multitude of previous demand forecasting research in that a relationship was not found between demands, flying hours, and sorties. However, this study expands upon this previous research and uses greater insight by delving into the correlation between weapon system demands/maintenance actions, flying hours, and number of sorties at the work unit code level. This study is also a unique, first attempt at obtaining a correlation between demands, or maintenance actions, and spares requirements at the work unit code level. Third, the study uses large, REMIS maintenance and sortie data sets for limited time periods from 1993 and 1994. In spite of the short time periods, the combined size of these data sets is still nearly 100,000 maintenance and sortie images. Although these data sets are a large, representative

sample of REMIS data, a significant correlation could not be found between demands/maintenance actions, cumulative flying hours, and number of sorties at the work unit code level. In other words, this study was not small in scale and used one of the best available sources of data. Finally, the most important management implication is the benefit which could be gained by determining a demand, flying hour, sortie relationship at the work unit code level, particularly the five digit level. In determining a demand, flying hour, sortie relationship, the specific five digit work unit code level could eventually be matched to a national stock number. The significant benefit gained by the Air Force is that maintenance and supply technicians could use this five digit work unit code/national stock number match to predict accurate, on-hand quantities of reparable spares. By knowing the expected number of maintenance actions, the required amount of reparable spares could be maintained in peacetime inventories or configured in MRSPs. Thus, demand forecasting research at the work unit level is worthwhile and could possibly derive significant financial and operational benefits, if properly researched, developed, and deployed.

Research Question Two: Results, Conclusions, and Management Implications

This section discusses the results obtained to answer research question two, "Can decision rules be established to forecast demands/maintenance actions based on a spares work unit code alone?" Conclusions and important Air Force management implications are also covered.

Results. Based on the lack of fit obtained from the multiple or Poisson regression models and the Poisson process estimation techniques, decision rules cannot be established based solely on a spares work unit code. The optimal answer to research question two would have been to develop regression or estimation models for specific work unit codes,

which would require inputs of flying hour and sortie profiles to calculate the expected number of demands or maintenance actions. However, the models obtained from this study show limited correlation and would not be suitable for establishment of work unit code decision rules.

Conclusions. A primary conclusion is that the limited results obtained under research question one prohibit the development of work unit code decision rules, which could be used to estimate demands or maintenance actions based on operational factors of a weapon system. However, a secondary conclusion is that the limited number of demands/maintenance actions, particularly at the five digit work unit code level, did portray the erratic, uncertain nature of demands that is a common characteristic of reparable spares. The current DoD situation of tight budgets and slim force structures require accurate, reliable reparable inventories to compensate for the inherent problem of demand uncertainty. Accurate work unit code decision rules could possibly assist in irradiating this troublesome problem of erratic, uncertain demand.

Management Implications. Development of work unit code decision rules could have significant management implications in managing peacetime reparable inventories and configuring MRSPs. A major issue is that the work unit code decision rules must be established from analytical models which exhibit a significant amount of correlation. In other words, the operational factors used to generate the models must provide accurate predictive capability (low residuals) on the expected number versus actual number of demands/maintenance actions. However, the problem is determining the operational factors of the weapon system which drive the demands/maintenance actions. In this study, only the operational factors of flying hours and sorties were analyzed, which produced poor estimation models.

Accurate reparable inventories could be maintained to support specific peacetime flying profiles by aligning demands with the operational factors which cause the demands

and establishing work unit code decision rules. In the current DoD environment of Depot Level Repairables (DLRs), an operational wing is very interested in the quantity of repairable spares used because, instead of the Air Force stock fund paying for these repairables, the operational wing now pays for repairables out of operations and maintenance (O&M) funding. Reliable work unit code decision rules could ensure only required repairable inventories are maintained, while also saving operational wing O&M funding.

The work unit code decision rules could also significantly enhance MRSP configurations. If a given wartime scenario is known, the operational factors which drive the estimation models could be determined. These operational factors can be used to compute the number of expected demands/maintenance actions from the specific work unit code decision rules. The number of spares required to support the wartime scenario could then be determined by cross referencing the work unit code to the national stock number. By only deploying the required number of MRSP spares, the Air Force saves money and conserves vital mobility airlift capability, which can quickly become a limiting logistical resource given the areas of responsibility (AORs) of our most recent conflicts/wars. The recommendations for follow-on research are included in the next section.

Recommendations for Follow-on Research

There are two primary factors which constrained the analysis conducted in this study. These factors are the limited amount of data and the limited number of predictor variables available for regression/estimation analysis. Therefore, we make the following three recommendations for follow-on research. First, evaluate the demands/maintenance action, flying hour, sortie relationship at the work unit code level with a much larger data set. Second, conduct an exploratory study to determine what factors drive specific spares

to fail. Finally, using the same technique of this study, analyze a different weapon system and compare the results. Each of these recommendations will now be discussed.

The data sets used in this study were limited to eight months for 1993 and five months for 1994. Analysis at the work unit code level should be evaluated with a CAMS or REMIS data set containing at least two years worth of data. A larger data set is suggested because the current D041 system uses eight quarters, or two years, of data to compute Air Force reparable requirements. By using a larger data set, an in-depth evaluation at the five digit work unit code level could be performed. Due to the lack of specific five digit work unit code occurrences, this study aggregated work unit codes at the two, three, four, and five digit level to obtain enough points for meaningful regression analysis. A larger data set would provide sufficient occurrences of five digit work unit codes to allow for regression or Poisson process analysis at the five digit work unit code level. Also, any subsequent decision rules would estimate the number of expected five digit work unit code occurrences. This study would have only provided an estimate of the number of two digit work unit code level occurrences, if significant correlation could have been obtained.

Along with a larger data set, an exploratory study also needs to be conducted to determine what factors cause failures of specific spares. Based on the results presented in Chapter Four, the limited correlation between demands/maintenance actions, flying hours, and number of sorties at the work unit code level show that demands/maintenance actions are possibly driven by factors other than flying hours and number of sorties. The exploratory study needs to evaluate other operational factors, such as engine cycles, takeoffs/landings, and so on, to determine exactly which factors drive failures of reparable spares. If operational factors can be determined, reliable regression or estimation models can be developed based on the operational factors that drive generation of demands or maintenance actions for particular spares.

The results obtained in this study should be considered as only a single data point because similar studies have not been performed at the work unit code level. A follow-on study could be performed against another fighter weapon system, such as an F-16, or against a transport weapon system, such as a C-5 or C-141. The operational profiles of the F-15C analyzed in this study could have contributed to the lack of correlation obtained by analyzing the demands, flying hour, sortie relationship at the work unit code level. However, a comparable study performed against a transport aircraft, or another fighter aircraft, could produce similar or contradictory results. Research at the work unit code level is still in the preliminary stages and should not be abandoned after analysis of only a single weapon system, the F-15C.

This research study was extremely involved, required several attempts to obtain the phased methodology, and also required a significant amount of data manipulation. Nevertheless, the research is an initial study and has possibly opened the door to a significant amount of follow-on research at the work unit code level. Follow-on researchers should first of all obtain a reliable, "clean," data set to use for analysis. In referring to a "clean" data set, researchers should possibly obtain a reliable data set directly from CAMS, instead of attempting to use a REMIS data base that receives downloads from CAMS. By focusing on only CAMS data, the analysis may be limited to only a few bases instead of the worldwide REMIS sample used in this study. However, accurate downloads of a small amount of CAMS data may provide more concrete results than a large amount of questionable REMIS data. The bottom line to any research study is that the results will only be as good as the data which is analyzed.

Second, researchers should possibly consider a time between maintenance actions study, if sufficient occurrences of specific five digit work unit codes are present within the data. This study aggregated work unit codes and did not consider the time between maintenance actions. However, by focusing on the time between occurrences of the same

maintenance action, a researcher would obtain a more accurate representation of the flying hours, number of sorties, or other operational characteristics that actually transpire between occurrences of the same work unit code.

Third, follow-on researchers should thoroughly research all available statistical techniques in attempting to obtain a best "fit" to the data. Although this study used three techniques, multiple regression, Poisson regression, and fitting of a Poisson process, the use of REMIS data covering only F-15C worldwide flying profiles may or may not have been a contributing factor to the lack of correlation. CAMS or REMIS data on another weapon system, analyzed with a the same or a different statistical process, may provide entirely different results to those obtained in this study.

Finally, researchers must focus on developing some form of decision mechanism, which addresses the problem, and is suitable for use in the operational Air Force. The work unit code decision rules, which were an objective of this study, were not successfully generated. However, if these decision rules could have been developed, they would have required only simple inputs of flying hours or number of sorties to generate the expected number of demands/maintenance actions. Decision mechanisms developed from a research study must be applicable to the problem addressed and straight-forward enough to be used by those who need them the most.

By using a larger data set, obtaining "true" operational factors which drive failures, and analyzing a different fighter or transport weapon system, reliable models could be developed to determine the expected number of demands/maintenance actions at the five digit work unit code level. The establishment of a five digit work unit code level cross reference to the national stock number would then determine required inventory levels for reparable spares. However, the primary driving factor is determining what factors cause specific reparable spares to fail.

Research Summary

Current Air Force demand forecasting systems, D041 and REALM, assume reparable demand is solely flying hour driven. This research study presents the problem of determining whether demands/maintenance actions of aircraft reparable spares are correlated to flying hours and number of sorties at the work unit code level. A literature review examining the relationship between reparable failures and/or predictor variables is provided, to include a discussion on reparable spares management in the civilian sector.

The research focuses on worldwide F-15C REMIS data from May to December 1993 and February to June 1994. The data sets cover only a specified timeframe, however, nearly 100,000 F-15C sortie and maintenance images are analyzed. A three phase methodology is used to determine whether or not a relationship exists between demands/maintenance actions, flying hours, and number of sorties at the work unit code level. Phase One of the methodology uses multiple linear regression, while Phase Two employs Poisson regression. The third and final phase fits a Poisson process to the data. Despite evaluating the data by three different statistical techniques, a conclusion is reached that significant correlation could not be obtained between demands/maintenance actions, flying hours, and number of sorties at the work unit code level. The lack of correlation also prohibits development of work unit code decision rules, which could be used to estimate expected numbers of demands/maintenance actions given a specified flying hour and sortie profile. Recommendations for follow-on research are also provided.

This research study supports previous demand forecasting by not determining an accurate relationship between demands, flying hours, and sorties. However, by evaluating the relationship at the work unit code level, the study takes a leap forward into uncharted territory concerning reparable demand forecasting. Further research at the work unit code level may provide the elusive answer required to resolve the issue of matching erratic, uncertain, reparable demands with accurate spares requirements. By allocating "accurate

quantities of the right item, to the right place, at the right time," the Air Force can enhance peacetime operational readiness, while significantly improving wartime combat capability.

Appendix A: Work Unit Code Breakdown

A work unit code is a five digit alpha numeric code used by Air Force maintenance personnel to track specific maintenance actions at the system level and first/second level of assembly for major aircraft systems. Thirty-three separate levels may be identified with a single five-digit work unit code.

Construction of the work unit code designates the first two numeric characters in the sequence followed by three zeros as the system designator. The first level of assembly is designated by the first two numeric characters plus the third alpha character followed by two zeros. Two numeric characters, an alpha character followed by another alpha character or a numeric character and a zero designates the second level of assembly. Finally, the third level of assembly uses an alpha or a numeric character in the fifth character position. The number "99" used in the fourth and fifth positions indicate Not Otherwise Coded (NOC). This code provides a work unit code for components that do not have specific codes assigned. For example:

33000 - SYSTEM DESIGNATOR

33A00 - FIRST LEVEL OF ASSEMBLY

33AA0 - SECOND LEVEL OF ASSEMBLY

33AAA - THIRD LEVEL OF ASSEMBLY

33A99 - NOC

(Reference: MIL-M-38769C (USAF))

Table A-1 on the next page lists the two digit work unit code levels and their corresponding F-15C aircraft systems/components.

Table A-1. Two Digit Work Unit Code Level/System or Component Comparison

| Two Digit Work Unit Code Level | F-15C System/Component |
|--------------------------------|---|
| 11 | Airframe |
| 12 | Cockpit and Fuselage Compartments |
| 13 | Landing Gear |
| 14 | Flight Controls |
| 23 | Turbofan Power Plant |
| 24 | Auxiliary Power Plant |
| 41 | Air Conditioning, Pressurization, and Surface Ice Control |
| 42 | Electrical Power Supply |
| 44 | Lighting System |
| 45 | Hydraulic and Pneumatic Power Supply |
| 46 | Fuel System |
| 47 | Oxygen System |
| 49 | Miscellaneous Utilities |
| 51 | Instruments |
| 52 | Autopilot |
| 55 | Malfunction Analysis and Recording Equipment |
| 57 | Integrated Guidance and Flight Control |
| 63 | UHF Communication |
| 65 | IFF |
| 71 | Radio Navigation |
| 74 | Fire Control |
| 75 | Weapons Delivery |
| 76 | Electronic Countermeasures |
| 97 | Explosive Devices and Components Miscellaneous Series Aircraft Explosive Devices |

Appendix B: Statistical Analysis System (SAS) Routines

SAS Routine Used to Sort and Merge the 1993 Analysis Data Set

*** READING MAINTENANCE ACTION DATA INTO DATA FILE 'temp' ***

```
options linesize=78;
data temp missover;
  infile maintain;
    input tail $ 1-4 eventid $ 5-13 wuc $ 14-18
      at $ 19 wd $ 20 hm $ 21-23
      beg 26-29 nos $ 30-32 date $ 33-37
      sdate $ 38-42 sbeg 43-46 snos $ 47-49
      slen 51-53 smis $ 54-57 ck 59
      nsort 61-62 sran $ 63-66;
```

*** READING SORTIE DATA INTO DATA FILE 'temp1' ***

```
data temp1 missover;
  infile sortie;
    input tail $ 1-4 sdate $ 5-9 snos $ 10-12
      smis $ 13-16 sbeg 17-20 slen 22-24
      sran $ 25-28 nsort 30-31;
```

*** DELETING 160 AIRCRAFT FROM 1993 SORTIE DATA WHICH HAVE NO
CORRESPONDING MAINTENANCE ACTION DATA ***

```
if tail eq '2012' or tail eq '2013' or tail eq '2014' then delete;
if tail eq '2015' or tail eq '2016' or tail eq '2017' then delete;
if tail eq '2018' or tail eq '2019' or tail eq '2021' then delete;
if tail eq '2022' or tail eq '2023' or tail eq '2024' then delete;
if tail eq '2025' or tail eq '2026' or tail eq '2027' then delete;
if tail eq '2028' or tail eq '2029' or tail eq '2030' then delete;
if tail eq '2031' or tail eq '2032' or tail eq '2033' then delete;
if tail eq '2034' or tail eq '2035' or tail eq '2036' then delete;
if tail eq '2037' or tail eq '2038' or tail eq '3010' then delete;
if tail eq '3011' or tail eq '3012' or tail eq '3013' then delete;
if tail eq '3014' or tail eq '3015' or tail eq '3016' then delete;
if tail eq '3017' or tail eq '3018' or tail eq '3019' then delete;
if tail eq '3020' or tail eq '3022' or tail eq '3023' then delete;
if tail eq '3024' or tail eq '3025' or tail eq '3026' then delete;
if tail eq '3027' or tail eq '3028' or tail eq '3029' then delete;
if tail eq '3030' or tail eq '3031' or tail eq '3032' then delete;
if tail eq '3033' or tail eq '3034' or tail eq '3035' then delete;
if tail eq '3036' or tail eq '3037' or tail eq '3038' then delete;
if tail eq '3039' or tail eq '3040' or tail eq '3041' then delete;
```

```

if tail eq '3042' or tail eq '3043' or tail eq '4001' then delete;
if tail eq '4002' or tail eq '4003' or tail eq '4004' then delete;
if tail eq '4005' or tail eq '4006' or tail eq '4007' then delete;
if tail eq '4008' or tail eq '4009' or tail eq '4010' then delete;
if tail eq '4011' or tail eq '4012' or tail eq '4013' then delete;
if tail eq '4014' or tail eq '4015' or tail eq '4016' then delete;
if tail eq '4017' or tail eq '4018' or tail eq '4019' then delete;
if tail eq '4020' or tail eq '4021' or tail eq '4022' then delete;
if tail eq '4023' or tail eq '4024' or tail eq '4025' then delete;
if tail eq '4026' or tail eq '4027' or tail eq '4028' then delete;
if tail eq '4030' or tail eq '4031' or tail eq '5093' then delete;
if tail eq '5094' or tail eq '5095' or tail eq '5096' then delete;
if tail eq '5097' or tail eq '5098' or tail eq '5099' then delete;
if tail eq '5100' or tail eq '5101' or tail eq '5102' then delete;
if tail eq '5103' or tail eq '5104' or tail eq '5105' then delete;
if tail eq '5106' or tail eq '5107' or tail eq '5108' then delete;
if tail eq '5110' or tail eq '5111' or tail eq '5112' then delete;
if tail eq '5113' or tail eq '5114' or tail eq '5115' then delete;
if tail eq '5117' or tail eq '5118' or tail eq '5119' then delete;
if tail eq '5120' or tail eq '5121' or tail eq '5122' then delete;
if tail eq '5123' or tail eq '5124' or tail eq '5125' then delete;
if tail eq '5126' or tail eq '5127' or tail eq '5128' then delete;
if tail eq '6143' or tail eq '6144' or tail eq '6145' then delete;
if tail eq '6146' or tail eq '6147' or tail eq '6148' then delete;
if tail eq '6149' or tail eq '6150' or tail eq '6151' then delete;
if tail eq '6152' or tail eq '6154' or tail eq '6155' then delete;
if tail eq '6156' or tail eq '6157' or tail eq '6158' then delete;
if tail eq '6159' or tail eq '6160' or tail eq '6161' then delete;
if tail eq '6162' or tail eq '6163' or tail eq '6164' then delete;
if tail eq '6165' or tail eq '6166' or tail eq '6167' then delete;
if tail eq '6168' or tail eq '6169' or tail eq '6170' then delete;
if tail eq '6171' or tail eq '6172' or tail eq '6173' then delete;
if tail eq '6174' or tail eq '6175' or tail eq '6176' then delete;
if tail eq '6177' or tail eq '6178' or tail eq '6179' then delete;
if tail eq '6180' then delete;

```

*** SORTING temp DATA FILE BY TAIL NUMBER, SORTIE DATE, AND
NUMBER OF SORTIES ***

```

proc sort data=temp;
  by tail sdate nsort;
run;

```

*** SORTING temp1 DATA FILE BY TAIL NUMBER, SORTIE DATE, AND
NUMBER OF SORTIES ***

```
proc sort data=temp1;  
  by tail sdate nsort;  
run;
```

*** MERGING temp AND temp1 DATA FILES INTO DATA FILE 'master1' BY
MATCHING TAIL NUMBER, SORTIE DATE, AND NUMBER OF SORTIES ***

```
data master1;  
  merge temp temp1;  
  by tail sdate nsort;  
run;
```

*** SORTING master1 DATA FILE BY TAIL NUMBER AND SORTIE DATE ***

```
proc sort data=master1;  
  by tail sdate;  
run;
```

*** PRINTING master1 DATA FILE ***

```
proc print data=master1;  
  var tail eventid wuc sdate snos slen smis nsort sran;  
run;
```

SAS Routine Used to Sort and Merge the 1994 Validation Data Set

*** READING MAINTENANCE ACTION DATA INTO DATA FILE 'temp' ***

```
options linesize=78;
data temp missover;
  infile dremism;
    input tail $ 1-4 eventid $ 5-13 wuc $ 14-18
      at $ 19 wd $ 20 hm $ 21-23 up 24-25 beg 26-29
      nos $ 30-32 s $ 33 date $ 34-38 sdate $ 39-43
      sbeg 44-47 snos $ 48-50 slen 51-54 smis $ 55-58
      ck 59-60 p $ 61 nsorts 62-64 from $ 65-68;
  if ck eq '99' then delete;
```

*** READING SORTIE DATA INTO DATA FILE 'temp1' ***

```
data temp1 missover;
  infile dremiss;
    input tail $ 1-4 sdate $ 5-9 snos $ 10-12
      smis $ 13-16 sbeg 17-20 slen 21-24 nsorts 25-27
      loc 28-29 from $ 30-33;
```

*** SORTING temp DATA FILE BY TAIL NUMBER, SORTIE DATE, BEGINNING
TIME OF SORTIE, AND SORTIE LENGTH ***

```
proc sort data=temp;
  by tail sdate sbeg slen;
run;
```

*** SORTING temp1 DATA FILE BY TAIL NUMBER, SORTIE DATE, BEGINNING
TIME OF SORTIE, AND SORTIE LENGTH ***

```
proc sort data=temp1;
  by tail sdate sbeg slen;
run;
```

*** MERGING temp AND temp1 DATA FILES INTO DATA FILE 'master94' BY
MATCHING TAIL NUMBER, SORTIE DATE, BEGINNING TIME OF SORTIE,
AND SORTIE LENGTH ***

```
data master94;
  merge temp temp1;
  by tail sdate sbeg slen;
run;
```

*** SORTING master94 DATA FILE BY TAIL NUMBER AND SORTIE DATE ***

```
proc sort data=master94;
  by tail sdate;
run;
```

```
*** PRINTING master94 DATA FILE ***  
proc print data=master94;  
  var tail sdate wuc slen from;  
run;
```


Appendix C: FORTRAN Programs and 1993/1994 Data Output Samples

FORTRAN Program Used for 1993 Analysis Data

PROGRAM CALCULATES NUMBER OF WORK UNIT CODE OCCURRENCES, CUMULATIVE FLYING HOURS, AND NUMBER OF SORTIES FOR EACH TAIL NUMBER IN THE 1993 DATA SET. CHARACTERS IN **BOLD TYPE** CAN BE VARIED TO CALCULATE NUMBER OF WORK UNIT CODE OCCURRENCES, CUMULATIVE FLYING HOURS, AND NUMBER OF SORTIES AT DIFFERENT LEVELS OF THE WORK UNIT CODE.

PROGRAM MAIN

*** INITIALIZING VARIABLES ***

```
REAL*8 SLEN
INTEGER*4 NOBS, NTAIL, NDATE, NSNOS, NSORT
CHARACTER*4 SMIS
CHARACTER*5 SRAN
CHARACTER*9 EVENTID
CHARACTER*5 WUC
CHARACTER*5 WUCINT
INTEGER*4 CURTAL
REAL*8 HOURS
INTEGER*4 NLINE, ISORT, NHITS
CHARACTER*7 TRASH
REAL*4 HOURS2, HOURSORT
INTEGER*4 ISORT2
```

*** OPENING FILES ***

```
OPEN(1,FILE='USER2:[SKEPHART.GETDATA]MERGE2.LIS',STATUS='OLD')
OPEN(2,FILE='USER2:[SKEPHART.GETDATA.DATA] 11A99.MULTI',
1 STATUS='NEW')
```

*** FORMATS ***

```
1001 FORMAT(I6,3X,I4,3X,A9,3X,A5,3X,I5,3X,I3,3X,F5.1,3X,
1 A4,3X,I4,3X,A5)
1002 FORMAT(A7)
2001 FORMAT(' ',I4,3X,A5,3X,I4,3X,F8.1,3X,I5,3X,F10.2,3X,I8,F10.2)
```

*** SETTING VARIABLES TO 0 AND IDENTIFYING WORK UNIT CODE ***

```
CURTAL = 0
HOURS = 0.0
WUCINT = '11A99'
NLINE = 0
ISORT = 0
NHITS = 0
100 CONTINUE
```

*** INCREMENTING DATA LINES AND DISCARDING PAGE HEADERS ***

```
NLINE = NLINE + 1
IF (NLINE .EQ. 62) THEN
  NLINE = 1
ENDIF
```

```
IF (NLINE .LT. 7) THEN
  READ(1,1002)TRASH
  GO TO 100
ENDIF
```

*** READING DATA IMAGES ***

```
READ(1,1001,END=900) NOBS, NTAIL, EVENTID, WUC, NDATE, NSNOS,
1  SLEN, SMIS, NSORT, SRAN
```

*** ESTABLISHING TAIL NUMBER OF THE AIRCRAFT ***

```
IF (CURTAL .EQ. 0) THEN
  CURTAL = NTAIL
ENDIF
```

*** ACCUMULATING FLYING HOURS, SORTIES, AND NUMBER OF WORK
UNIT CODE HITS FOR EACH TAIL NUMBER ***

```
IF (NTAIL .EQ. CURTAL) THEN
  HOURS = HOURS + SLEN
  ISORT = ISORT + 1
  IF (WUC .EQ. WUCINT) THEN
    NHITS = NHITS + 1
  ENDIF
  GO TO 100
ENDIF
```

*** CALCULATING SQUARED AND INTERACTION PARAMETERS ***

HOURS2 = HOURS * HOURS
ISORT2 = ISORT * ISORT
HOURSORT = HOURS * REAL(ISORT)

*** WRITING RECORDS TO FILE ***

WRITE(2,2001) CURTAL, WUCINT, NHITS, HOURS, ISORT,
1 HOURS2, ISORT2, HOURSORT

*** RESETTING VARIABLES ***

CURTAL = NTAIL
HOURS = SLEN
ISORT = 1
NHITS = 0
IF (WUC.EQ. WUCINT) THEN
NHITS = 1
ENDIF
GO TO 100
900 CONTINUE

*** CLOSING DATA FILES AND TERMINATING PROGRAM ***

CLOSE(1)
CLOSE(2)

STOP
END

FORTTRAN Program Used for 1994 Validation Data

PROGRAM CALCULATES NUMBER OF WORK UNIT CODE OCCURRENCES, CUMULATIVE FLYING HOURS, AND NUMBER OF SORTIES FOR EACH TAIL NUMBER IN THE 1994 DATA SET. CHARACTERS IN **BOLD TYPE** CAN BE VARIED TO CALCULATE NUMBER OF WORK UNIT CODE OCCURRENCES, CUMULATIVE FLYING HOURS, AND NUMBER OF SORTIES AT DIFFERENT LEVELS OF THE WORK UNIT CODE.

PROGRAM VALID

*** INITIALIZING VARIABLES ***

```
REAL*8 SLEN
INTEGER*4 NOBS, NTAIL, NDATE
CHARACTER*5 SRAN
CHARACTER*2 WUC
CHARACTER*2 WUCINT
INTEGER*4 CURTAL
REAL*8 HOURS
INTEGER*4 NLINE, ISORT, NHITS
CHARACTER*7 TRASH
REAL*4 HOURS2, HOURSORT
INTEGER*4 ISORT2
```

*** OPENING FILES ***

```
OPEN(1,FILE='USER2:[SKEPHART.VALDATA]1994MAINT.LIS',STATUS='OLD')
OPEN(2,FILE='USER2:[SKEPHART.VALDATA.DATA]11.VAL',
1 STATUS='NEW')
```

*** FORMATS ***

```
1001 FORMAT(T15,I6,T25,I4,T33,I5,T42,A2,T50,F5.1,T59,A5)
1002 FORMAT(A7)
2001 FORMAT(' ',I4,3X,A2,6X,I4,3X,F8.1,3X,I5,3X,F10.2,3X,I8,F10.2)
```

*** SETTING VARIABLES TO 0 AND IDENTIFYING WORK UNIT CODE ***

```
CURTAL = 0
HOURS = 0.0
WUCINT = '11'
NLINE = 0
ISORT = 0
```

```

      NHITS = 0
100  CONTINUE

*** INCREMENTING DATA LINES AND DISCARDING PAGE HEADERS ***

      NLINE = NLINE + 1
      IF (NLINE .EQ. 62) THEN
        NLINE = 1
      ENDIF

      IF (NLINE .LT. 7) THEN
        READ(1,1002)TRASH
        GO TO 100
      ENDIF

*** READING DATA IMAGES ***

      READ(1,1001,END=900) NOBS, NTAIL, NDATE, WUC, SLEN, SRAN

*** ESTABLISHING TAIL NUMBER OF THE AIRCRAFT ***

      IF (CURTAL .EQ. 0) THEN
        CURTAL = NTAIL
      ENDIF

*** ACCUMULATING FLYING HOURS, SORTIES, AND NUMBER OF WORK
UNIT CODE HITS FOR EACH TAIL NUMBER ***

      IF (NTAIL .EQ. CURTAL) THEN
        HOURS = HOURS + SLEN
        ISORT = ISORT + 1
        IF (WUC .EQ. WUCINT) THEN
          NHITS = NHITS + 1
        ENDIF
        GO TO 100
      ENDIF

*** CALCULATING SQUARED AND INTERACTION PARAMETERS ***

      HOURS2 = HOURS * HOURS
      ISORT2 = ISORT * ISORT
      HOURSORT = HOURS * REAL(ISORT)

```

*** WRITING RECORDS TO FILE ***

```
WRITE(2,2001) CURTAL, WUCINT, NHITS, HOURS, ISORT,  
1      HOURS2, ISORT2, HOURSORT
```

*** RESETTING VARIABLES ***

```
CURTAL = NTAIL  
HOURS = SLEN  
ISORT = 1  
NHITS = 0  
IF (WUC .EQ. WUCINT) THEN  
    NHITS = 1  
ENDIF  
GO TO 100  
900 CONTINUE
```

*** CLOSING DATA FILES AND TERMINATING PROGRAM ***

```
CLOSE(1)  
CLOSE(2)
```

```
STOP  
END
```

Table C-1. 1993 FORTRAN Program Data Output Sample

| TAIL | WUC | DMDS | CUMFH | NSORT | CUMFH2 | NSORT2 | FHNSORT |
|------|-----|------|-------|-------|----------|--------|---------|
| 2 | 13 | 1 | 227.7 | 141 | 51847.29 | 19881 | 32105.7 |
| 3 | 13 | 3 | 234.4 | 93 | 54943.36 | 8649 | 21799.2 |
| 4 | 13 | 4 | 72 | 51 | 5184 | 2601 | 3672 |
| 5 | 13 | 6 | 187.1 | 139 | 35006.41 | 19321 | 26006.9 |
| 6 | 13 | 8 | 182.4 | 160 | 33269.76 | 25600 | 29184 |
| 9 | 13 | 4 | 288.1 | 165 | 83001.61 | 27225 | 47536.5 |
| 10 | 13 | 10 | 163.8 | 89 | 26830.44 | 7921 | 14578.2 |
| 11 | 13 | 2 | 225.1 | 117 | 50670.01 | 13689 | 26336.7 |
| 12 | 13 | 10 | 444 | 213 | 197136 | 45369 | 94572 |
| 13 | 13 | 5 | 190 | 149 | 36100 | 22201 | 28310 |
| 14 | 13 | 12 | 205 | 161 | 42025 | 25921 | 33005 |
| 15 | 13 | 9 | 163.6 | 134 | 26764.96 | 17956 | 21922.4 |
| 16 | 13 | 5 | 136.6 | 107 | 18659.56 | 11449 | 14616.2 |
| 18 | 13 | 0 | 62.5 | 36 | 3906.25 | 1296 | 2250 |
| 19 | 13 | 4 | 163.7 | 127 | 26797.69 | 16129 | 20789.9 |
| 20 | 13 | 2 | 140.6 | 96 | 19768.36 | 9216 | 13497.6 |
| 22 | 13 | 4 | 398.2 | 134 | 158563.2 | 17956 | 53358.8 |
| 24 | 13 | 4 | 70.9 | 47 | 5026.81 | 2209 | 3332.3 |
| 26 | 13 | 11 | 221 | 144 | 48841 | 20736 | 31824 |
| 27 | 13 | 6 | 201.4 | 127 | 40561.96 | 16129 | 25577.8 |
| 28 | 13 | 1 | 347.3 | 126 | 120617.3 | 15876 | 43759.8 |
| 29 | 13 | 5 | 223.3 | 138 | 49862.89 | 19044 | 30815.4 |
| 30 | 13 | 5 | 140.1 | 116 | 19628.01 | 13456 | 16251.6 |
| 31 | 13 | 4 | 413.4 | 127 | 170899.6 | 16129 | 52501.8 |
| 33 | 13 | 4 | 117.1 | 101 | 13712.41 | 10201 | 11827.1 |
| 34 | 13 | 2 | 155.8 | 142 | 24273.64 | 20164 | 22123.6 |
| 35 | 13 | 2 | 173.5 | 99 | 30102.25 | 9801 | 17176.5 |
| 38 | 13 | 2 | 224.1 | 144 | 50220.81 | 20736 | 32270.4 |
| 39 | 13 | 5 | 147.3 | 91 | 21697.29 | 8281 | 13404.3 |
| 40 | 13 | 8 | 151.3 | 132 | 22891.69 | 17424 | 19971.6 |
| 41 | 13 | 3 | 18.4 | 15 | 338.56 | 225 | 276 |
| 43 | 13 | 4 | 151.4 | 139 | 22921.96 | 19321 | 21044.6 |
| 44 | 13 | 2 | 150.4 | 140 | 22620.16 | 19600 | 21056 |
| 45 | 13 | 7 | 165.9 | 147 | 27522.81 | 21609 | 24387.3 |
| 46 | 13 | 4 | 159.8 | 98 | 25536.04 | 9604 | 15660.4 |
| 47 | 13 | 5 | 215.6 | 133 | 46483.36 | 17689 | 28674.8 |
| 48 | 13 | 7 | 161.3 | 99 | 26017.69 | 9801 | 15968.7 |
| 49 | 13 | 4 | 105.4 | 104 | 11109.16 | 10816 | 10961.6 |
| 50 | 13 | 1 | 116.4 | 73 | 13548.96 | 5329 | 8497.2 |
| 51 | 13 | 3 | 209.6 | 136 | 43932.16 | 18496 | 28505.6 |
| 53 | 13 | 1 | 168 | 118 | 28224 | 13924 | 19824 |

| | | | | | | | |
|------|----|----|-------|-----|----------|-------|----------|
| 54 | 13 | 6 | 210.8 | 148 | 44436.64 | 21904 | 31198.4 |
| 55 | 13 | 15 | 248.9 | 174 | 61951.21 | 30276 | 43308.6 |
| 57 | 13 | 8 | 139.6 | 87 | 19488.16 | 7569 | 12145.2 |
| 58 | 13 | 8 | 346.2 | 252 | 119854.4 | 63504 | 87242.4 |
| 60 | 13 | 13 | 193.6 | 157 | 37480.96 | 24649 | 30395.2 |
| 61 | 13 | 9 | 173.2 | 123 | 29998.24 | 15129 | 21303.6 |
| 1020 | 13 | 2 | 223.4 | 139 | 49907.56 | 19321 | 31052.6 |
| 1021 | 13 | 1 | 92 | 52 | 8464 | 2704 | 4784 |
| 1022 | 13 | 5 | 164.3 | 133 | 26994.49 | 17689 | 21851.9 |
| 1023 | 13 | 1 | 508.9 | 176 | 258979.2 | 30976 | 89566.4 |
| 1024 | 13 | 5 | 166.8 | 133 | 27822.24 | 17689 | 22184.4 |
| 1025 | 13 | 5 | 189.3 | 139 | 35834.49 | 19321 | 26312.7 |
| 1026 | 13 | 5 | 198.6 | 144 | 39441.96 | 20736 | 28598.4 |
| 1027 | 13 | 9 | 143.3 | 108 | 20534.89 | 11664 | 15476.4 |
| 1028 | 13 | 7 | 175.1 | 143 | 30660.01 | 20449 | 25039.3 |
| 1029 | 13 | 7 | 189.7 | 154 | 35986.09 | 23716 | 29213.8 |
| 1030 | 13 | 1 | 160.7 | 127 | 25824.49 | 16129 | 20408.9 |
| 1031 | 13 | 8 | 240.3 | 183 | 57744.09 | 33489 | 43974.9 |
| 1032 | 13 | 3 | 590.4 | 214 | 348572.2 | 45796 | 126345.6 |
| 1033 | 13 | 2 | 513.7 | 181 | 263887.7 | 32761 | 92979.7 |
| 1034 | 13 | 3 | 174 | 134 | 30276 | 17956 | 23316 |
| 1035 | 13 | 4 | 173.4 | 138 | 30067.56 | 19044 | 23929.2 |
| 1036 | 13 | 5 | 382.8 | 124 | 146535.8 | 15376 | 47467.2 |
| 1037 | 13 | 5 | 540.6 | 201 | 292248.4 | 40401 | 108660.6 |
| 1038 | 13 | 9 | 231.7 | 182 | 53684.89 | 33124 | 42169.4 |
| 1039 | 13 | 6 | 203.2 | 158 | 41290.24 | 24964 | 32105.6 |
| 1040 | 13 | 7 | 189.3 | 149 | 35834.49 | 22201 | 28205.7 |
| 1041 | 13 | 11 | 232.3 | 177 | 53963.29 | 31329 | 41117.1 |
| 1042 | 13 | 8 | 220.4 | 155 | 48576.16 | 24025 | 34162 |
| 1043 | 13 | 0 | 59.1 | 29 | 3492.81 | 841 | 1713.9 |
| 1044 | 13 | 2 | 183.6 | 131 | 33708.96 | 17161 | 24051.6 |
| 1045 | 13 | 6 | 179 | 137 | 32041 | 18769 | 24523 |
| 1046 | 13 | 13 | 176.4 | 137 | 31116.96 | 18769 | 24166.8 |
| 1047 | 13 | 9 | 193.7 | 126 | 37519.69 | 15876 | 24406.2 |
| 1048 | 13 | 9 | 223.1 | 168 | 49773.61 | 28224 | 37480.8 |
| 1050 | 13 | 5 | 161.1 | 123 | 25953.21 | 15129 | 19815.3 |
| 1051 | 13 | 6 | 181 | 141 | 32761 | 19881 | 25521 |
| 1053 | 13 | 8 | 122.5 | 113 | 15006.25 | 12769 | 13842.5 |
| 1054 | 13 | 0 | 78.8 | 50 | 6209.44 | 2500 | 3940 |
| 1055 | 13 | 3 | 151 | 142 | 22801 | 20164 | 21442 |
| 1061 | 13 | 7 | 259.8 | 170 | 67496.04 | 28900 | 44166 |
| 1062 | 13 | 3 | 123.8 | 100 | 15326.44 | 10000 | 12380 |
| 1063 | 13 | 6 | 202.8 | 157 | 41127.84 | 24649 | 31839.6 |
| 1064 | 13 | 10 | 149.1 | 118 | 22230.81 | 13924 | 17593.8 |

| | | | | | | | |
|------|----|----|-------|-----|----------|-------|---------|
| 1065 | 13 | 14 | 251.5 | 191 | 63252.25 | 36481 | 48036.5 |
| 2008 | 13 | 6 | 135.9 | 108 | 18468.81 | 11664 | 14677.2 |
| 2009 | 13 | 6 | 198.8 | 152 | 39521.44 | 23104 | 30217.6 |
| 2010 | 13 | 8 | 226.4 | 164 | 51256.96 | 26896 | 37129.6 |
| 2011 | 13 | 1 | 165.5 | 128 | 27390.25 | 16384 | 21184 |
| 2044 | 13 | 14 | 178.2 | 141 | 31755.24 | 19881 | 25126.2 |
| 2045 | 13 | 1 | 17.2 | 13 | 295.84 | 169 | 223.6 |
| 2046 | 13 | 7 | 254.2 | 205 | 64617.64 | 42025 | 52111 |
| 2047 | 13 | 5 | 75.6 | 62 | 5715.36 | 3844 | 4687.2 |
| 2048 | 13 | 10 | 189.8 | 152 | 36024.04 | 23104 | 28849.6 |
| 3046 | 13 | 4 | 123.4 | 96 | 15227.56 | 9216 | 11846.4 |
| 3047 | 13 | 8 | 221.3 | 183 | 48973.69 | 33489 | 40497.9 |
| 3048 | 13 | 6 | 206.1 | 170 | 42477.21 | 28900 | 35037 |
| 3049 | 13 | 8 | 288.3 | 223 | 83116.89 | 49729 | 64290.9 |
| 3050 | 13 | 7 | 201.1 | 151 | 40441.21 | 22801 | 30366.1 |
| 4043 | 13 | 10 | 164.2 | 139 | 26961.64 | 19321 | 22823.8 |
| 4044 | 13 | 1 | 42.6 | 37 | 1814.76 | 1369 | 1576.2 |
| 4045 | 13 | 10 | 188.2 | 126 | 35419.24 | 15876 | 23713.2 |
| 4046 | 13 | 7 | 104.7 | 53 | 10962.09 | 2809 | 5549.1 |
| 5129 | 13 | 5 | 229.1 | 190 | 52486.81 | 36100 | 43529 |
| 5130 | 13 | 6 | 236.2 | 176 | 55790.44 | 30976 | 41571.2 |
| 5131 | 13 | 1 | 175.5 | 155 | 30800.25 | 24025 | 27202.5 |
| 5132 | 13 | 3 | 211.5 | 164 | 44732.25 | 26896 | 34686 |
| 5133 | 13 | 0 | 112.8 | 92 | 12723.84 | 8464 | 10377.6 |
| 5134 | 13 | 2 | 269.8 | 191 | 72792.04 | 36481 | 51531.8 |
| 6181 | 13 | 7 | 212.8 | 137 | 45283.84 | 18769 | 29153.6 |
| 6182 | 13 | 1 | 134.2 | 104 | 18009.64 | 10816 | 13956.8 |
| 8468 | 13 | 2 | 62.5 | 52 | 3906.25 | 2704 | 3250 |
| 8469 | 13 | 13 | 227.9 | 164 | 51938.41 | 26896 | 37375.6 |
| 8470 | 13 | 6 | 179.3 | 117 | 32148.49 | 13689 | 20978.1 |
| 8471 | 13 | 10 | 185.6 | 143 | 34447.36 | 20449 | 26540.8 |
| 8473 | 13 | 12 | 174.2 | 132 | 30345.64 | 17424 | 22994.4 |
| 8474 | 13 | 14 | 197.6 | 144 | 39045.76 | 20736 | 28454.4 |
| 8475 | 13 | 6 | 128.8 | 107 | 16589.44 | 11449 | 13781.6 |
| 8476 | 13 | 20 | 254.9 | 190 | 64974.01 | 36100 | 48431 |
| 8477 | 13 | 3 | 100 | 56 | 10000 | 3136 | 5600 |
| 8478 | 13 | 7 | 199 | 134 | 39601 | 17956 | 26666 |
| 8479 | 13 | 7 | 265.9 | 158 | 70702.81 | 24964 | 42012.2 |
| 8480 | 13 | 2 | 104.2 | 74 | 10857.64 | 5476 | 7710.8 |
| 8482 | 13 | 9 | 178.9 | 110 | 32005.21 | 12100 | 19679 |
| 8483 | 13 | 5 | 102.2 | 53 | 10444.84 | 2809 | 5416.6 |
| 8484 | 13 | 3 | 66.7 | 51 | 4448.89 | 2601 | 3401.7 |
| 8485 | 13 | 6 | 118.1 | 66 | 13947.61 | 4356 | 7794.6 |
| 8486 | 13 | 5 | 214.1 | 148 | 45838.81 | 21904 | 31686.8 |

| | | | | | | | |
|------|----|----|-------|-----|----------|-------|---------|
| 8487 | 13 | 13 | 218 | 156 | 47524 | 24336 | 34008 |
| 8488 | 13 | 11 | 219.1 | 173 | 48004.81 | 29929 | 37904.3 |
| 8489 | 13 | 8 | 167.7 | 118 | 28123.29 | 13924 | 19788.6 |
| 8490 | 13 | 7 | 304.6 | 191 | 92781.16 | 36481 | 58178.6 |
| 8491 | 13 | 12 | 181 | 134 | 32761 | 17956 | 24254 |
| 8492 | 13 | 1 | 51.9 | 37 | 2693.61 | 1369 | 1920.3 |
| 8493 | 13 | 11 | 244.9 | 177 | 59976.01 | 31329 | 43347.3 |
| 8494 | 13 | 9 | 182.5 | 138 | 33306.25 | 19044 | 25185 |
| 8496 | 13 | 9 | 261.8 | 157 | 68539.24 | 24649 | 41102.6 |
| 8497 | 13 | 8 | 214.9 | 136 | 46182.01 | 18496 | 29226.4 |
| 8498 | 13 | 11 | 234.7 | 168 | 55084.09 | 28224 | 39429.6 |
| 8499 | 13 | 11 | 215.6 | 141 | 46483.36 | 19881 | 30399.6 |
| 8500 | 13 | 3 | 77.1 | 59 | 5944.41 | 3481 | 4548.9 |
| 8501 | 13 | 9 | 254.2 | 142 | 64617.64 | 20164 | 36096.4 |
| 8502 | 13 | 6 | 176.9 | 127 | 31293.61 | 16129 | 22466.3 |
| 8503 | 13 | 6 | 214.4 | 135 | 45967.36 | 18225 | 28944 |
| 8504 | 13 | 5 | 269.6 | 171 | 72684.16 | 29241 | 46101.6 |
| 8505 | 13 | 7 | 216.5 | 164 | 46872.25 | 26896 | 35506 |
| 8506 | 13 | 11 | 113.3 | 83 | 12836.89 | 6889 | 9403.9 |
| 8507 | 13 | 5 | 270.1 | 172 | 72954.01 | 29584 | 46457.2 |
| 8508 | 13 | 9 | 222.3 | 132 | 49417.29 | 17424 | 29343.6 |
| 8509 | 13 | 3 | 191.6 | 112 | 36710.56 | 12544 | 21459.2 |
| 8510 | 13 | 5 | 74 | 62 | 5476 | 3844 | 4588 |
| 8511 | 13 | 7 | 212 | 156 | 44944 | 24336 | 33072 |
| 8512 | 13 | 6 | 109.3 | 84 | 11946.49 | 7056 | 9181.2 |
| 8513 | 13 | 5 | 239.2 | 161 | 57216.64 | 25921 | 38511.2 |
| 8514 | 13 | 8 | 149.5 | 115 | 22350.25 | 13225 | 17192.5 |
| 8515 | 13 | 10 | 188.1 | 131 | 35381.61 | 17161 | 24641.1 |
| 8516 | 13 | 5 | 94.6 | 65 | 8949.16 | 4225 | 6149 |
| 8517 | 13 | 10 | 122.6 | 83 | 15030.76 | 6889 | 10175.8 |
| 8518 | 13 | 5 | 101.4 | 68 | 10281.96 | 4624 | 6895.2 |
| 8519 | 13 | 0 | 11.7 | 11 | 136.89 | 121 | 128.7 |
| 8520 | 13 | 7 | 240.1 | 159 | 57648.01 | 25281 | 38175.9 |
| 8521 | 13 | 3 | 81.9 | 60 | 6707.61 | 3600 | 4914 |
| 8522 | 13 | 8 | 234.4 | 148 | 54943.36 | 21904 | 34691.2 |
| 8523 | 13 | 7 | 164.9 | 129 | 27192.01 | 16641 | 21272.1 |
| 8525 | 13 | 4 | 159.6 | 117 | 25472.16 | 13689 | 18673.2 |
| 8527 | 13 | 6 | 137.1 | 94 | 18796.41 | 8836 | 12887.4 |
| 8528 | 13 | 8 | 308.8 | 192 | 95357.44 | 36864 | 59289.6 |
| 8529 | 13 | 9 | 173.5 | 138 | 30102.25 | 19044 | 23943 |
| 8530 | 13 | 5 | 55.3 | 46 | 3058.09 | 2116 | 2543.8 |
| 8531 | 13 | 10 | 220.2 | 139 | 48488.04 | 19321 | 30607.8 |
| 8532 | 13 | 3 | 173.6 | 103 | 30136.96 | 10609 | 17880.8 |
| 8533 | 13 | 9 | 301.6 | 192 | 90962.56 | 36864 | 57907.2 |

| | | | | | | | |
|------|----|----|-------|-----|----------|-------|---------|
| 8535 | 13 | 8 | 162.8 | 127 | 26503.84 | 16129 | 20675.6 |
| 8536 | 13 | 17 | 233.9 | 146 | 54709.21 | 21316 | 34149.4 |
| 8537 | 13 | 6 | 108.5 | 75 | 11772.25 | 5625 | 8137.5 |
| 8538 | 13 | 2 | 38.5 | 32 | 1482.25 | 1024 | 1232 |
| 8539 | 13 | 11 | 182.6 | 131 | 33342.76 | 17161 | 23920.6 |
| 8541 | 13 | 5 | 166.7 | 110 | 27788.89 | 12100 | 18337 |
| 8542 | 13 | 5 | 58.9 | 46 | 3469.21 | 2116 | 2709.4 |
| 8543 | 13 | 19 | 216.6 | 123 | 46915.56 | 15129 | 26641.8 |
| 8544 | 13 | 8 | 261.8 | 157 | 68539.24 | 24649 | 41102.6 |
| 8545 | 13 | 8 | 220.3 | 153 | 48532.09 | 23409 | 33705.9 |
| 8546 | 13 | 10 | 348.8 | 230 | 121661.4 | 52900 | 80224 |
| 8547 | 13 | 17 | 253.8 | 168 | 64414.44 | 28224 | 42638.4 |
| 8548 | 13 | 6 | 133.3 | 80 | 17768.89 | 6400 | 10664 |
| 8549 | 13 | 0 | 226.4 | 149 | 51256.96 | 22201 | 33733.6 |
| 8561 | 13 | 2 | 113.9 | 89 | 12973.21 | 7921 | 10137.1 |
| 8562 | 13 | 4 | 88.8 | 66 | 7885.44 | 4356 | 5860.8 |
| 8563 | 13 | 3 | 127.9 | 93 | 16358.41 | 8649 | 11894.7 |
| 8564 | 13 | 7 | 257.3 | 175 | 66203.29 | 30625 | 45027.5 |
| 8565 | 13 | 12 | 180.1 | 135 | 32436.01 | 18225 | 24313.5 |
| 8566 | 13 | 8 | 192.7 | 148 | 37133.29 | 21904 | 28519.6 |
| 8567 | 13 | 9 | 245.8 | 168 | 60417.64 | 28224 | 41294.4 |
| 8568 | 13 | 19 | 225.1 | 179 | 50670.01 | 32041 | 40292.9 |
| 8569 | 13 | 5 | 195.4 | 115 | 38181.16 | 13225 | 22471 |
| 8570 | 13 | 16 | 274.4 | 202 | 75295.36 | 40804 | 55428.8 |
| 8571 | 13 | 3 | 39.6 | 37 | 1568.16 | 1369 | 1465.2 |
| 8572 | 13 | 17 | 182.2 | 151 | 33196.84 | 22801 | 27512.2 |
| 8573 | 13 | 14 | 225 | 174 | 50625 | 30276 | 39150 |
| 8574 | 13 | 7 | 129.3 | 108 | 16718.49 | 11664 | 13964.4 |
| 9007 | 13 | 3 | 170.8 | 120 | 29172.64 | 14400 | 20496 |
| 9008 | 13 | 5 | 94.9 | 67 | 9006.01 | 4489 | 6358.3 |
| 9009 | 13 | 1 | 205 | 138 | 42025 | 19044 | 28290 |
| 9011 | 13 | 10 | 191.7 | 172 | 36748.89 | 29584 | 32972.4 |
| 9012 | 13 | 9 | 146 | 130 | 21316 | 16900 | 18980 |
| 9013 | 13 | 1 | 57.4 | 35 | 3294.76 | 1225 | 2009 |
| 9014 | 13 | 14 | 162.4 | 126 | 26373.76 | 15876 | 20462.4 |
| 9016 | 13 | 10 | 197.1 | 158 | 38848.41 | 24964 | 31141.8 |
| 9020 | 13 | 5 | 232.1 | 153 | 53870.41 | 23409 | 35511.3 |
| 9021 | 13 | 3 | 185.2 | 121 | 34299.04 | 14641 | 22409.2 |
| 9022 | 13 | 2 | 418.8 | 153 | 175393.4 | 23409 | 64076.4 |
| 9025 | 13 | 2 | 155.9 | 67 | 24304.81 | 4489 | 10445.3 |
| 9026 | 13 | 18 | 258.7 | 200 | 66925.69 | 40000 | 51740 |
| 9029 | 13 | 12 | 232.6 | 189 | 54102.76 | 35721 | 43961.4 |
| 9030 | 13 | 12 | 189.5 | 151 | 35910.25 | 22801 | 28614.5 |
| 9034 | 13 | 8 | 205.2 | 161 | 42107.04 | 25921 | 33037.2 |

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|------|-------|------|---------|-------|----------|-------|---------|
| 9035 | 13 | 2 | 114.3 | 107 | 13064.49 | 11449 | 12230.1 |
| 9036 | 13 | 1 | 370.6 | 128 | 137344.4 | 16384 | 47436.8 |
| 9037 | 13 | 0 | 47.8 | 40 | 2284.84 | 1600 | 1912 |
| 9041 | 13 | 5 | 136 | 92 | 18496 | 8464 | 12512 |
| 9042 | 13 | 2 | 254.9 | 154 | 64974.01 | 23716 | 39254.6 |
| 9046 | 13 | 0 | 112.1 | 83 | 12566.41 | 6889 | 9304.3 |
| 9048 | 13 | 5 | 176 | 112 | 30976 | 12544 | 19712 |
| 9049 | 13 | 3 | 150.9 | 95 | 22770.81 | 9025 | 14335.5 |
| 9050 | 13 | 1 | 258.3 | 155 | 66718.89 | 24025 | 40036.5 |
| 9053 | 13 | 2 | 277 | 180 | 76729 | 32400 | 49860 |
| 9054 | 13 | 6 | 241.5 | 157 | 58322.25 | 24649 | 37915.5 |
| 9056 | 13 | 3 | 207 | 141 | 42849 | 19881 | 29187 |
| 9057 | 13 | 5 | 178 | 139 | 31684 | 19321 | 24742 |
| 9058 | 13 | 6 | 414.6 | 163 | 171893.2 | 26569 | 67579.8 |
| 9059 | 13 | 6 | 185.8 | 131 | 34521.64 | 17161 | 24339.8 |
| 9064 | 13 | 12 | 220.5 | 191 | 48620.25 | 36481 | 42115.5 |
| 9065 | 13 | 5 | 280 | 181 | 78400 | 32761 | 50680 |
| 9066 | 13 | 2 | 240.4 | 160 | 57792.16 | 25600 | 38464 |
| 9068 | 13 | 9 | 148.4 | 126 | 22022.56 | 15876 | 18698.4 |
| 9069 | 13 | 9 | 349.7 | 156 | 122290.1 | 24336 | 54553.2 |
| 9070 | 13 | 6 | 186.8 | 117 | 34894.24 | 13689 | 21855.6 |
| 9072 | 13 | 2 | 414.7 | 165 | 171976.1 | 27225 | 68425.5 |
| 9073 | 13 | 1 | 451.6 | 150 | 203942.6 | 22500 | 67740 |
| 9074 | 13 | 9 | 292.8 | 192 | 85731.84 | 36864 | 56217.6 |
| 9075 | 13 | 11 | 319.4 | 213 | 102016.4 | 45369 | 68032.2 |
| 9076 | 13 | 0 | 350.9 | 121 | 123130.8 | 14641 | 42458.9 |
| 9077 | 13 | 7 | 146.6 | 112 | 21491.56 | 12544 | 16419.2 |
| 9078 | 13 | 4 | 88.6 | 62 | 7849.96 | 3844 | 5493.2 |
| 9079 | 13 | 8 | 294.6 | 200 | 86789.16 | 40000 | 58920 |
| 9080 | 13 | 8 | 248.3 | 173 | 61652.89 | 29929 | 42955.9 |
| | TOTAL | 1560 | 48337.5 | 32057 | | | |

Table C-2. 1994 FORTRAN Program Data Output Sample

| TAIL | WUC | DMDS | CUMFH | NSORT | CUMFH2 | NSORT2 | FHNSORT |
|------|-----|------|-------|-------|----------|--------|---------|
| 2 | 11 | 0 | 178.2 | 133 | 31755.24 | 17689 | 23700.6 |
| 3 | 11 | 1 | 142.5 | 115 | 20306.25 | 13225 | 16387.5 |
| 4 | 11 | 0 | 205 | 95 | 42025 | 9025 | 19475 |
| 5 | 11 | 1 | 180.6 | 149 | 32616.36 | 22201 | 26909.4 |
| 6 | 11 | 7 | 131.1 | 133 | 17187.21 | 17689 | 17436.3 |
| 9 | 11 | 3 | 175.7 | 120 | 30870.49 | 14400 | 21084 |
| 10 | 11 | 3 | 186.9 | 161 | 34931.61 | 25921 | 30090.9 |
| 11 | 11 | 2 | 185.7 | 130 | 34484.49 | 16900 | 24141 |
| 12 | 11 | 4 | 318.8 | 106 | 101633.4 | 11236 | 33792.8 |
| 13 | 11 | 2 | 166.8 | 118 | 27822.24 | 13924 | 19682.4 |
| 14 | 11 | 1 | 147.5 | 113 | 21756.25 | 12769 | 16667.5 |
| 15 | 11 | 10 | 169.5 | 141 | 28730.25 | 19881 | 23899.5 |
| 16 | 11 | 11 | 168.2 | 134 | 28291.24 | 17956 | 22538.8 |
| 18 | 11 | 0 | 130.9 | 92 | 17134.81 | 8464 | 12042.8 |
| 19 | 11 | 6 | 142.6 | 121 | 20334.76 | 14641 | 17254.6 |
| 20 | 11 | 8 | 260.9 | 139 | 68068.81 | 19321 | 36265.1 |
| 21 | 11 | 0 | 131.3 | 83 | 17239.69 | 6889 | 10897.9 |
| 22 | 11 | 0 | 148.2 | 105 | 21963.24 | 11025 | 15561 |
| 24 | 11 | 0 | 150.1 | 90 | 22530.01 | 8100 | 13509 |
| 26 | 11 | 7 | 170.4 | 134 | 29036.16 | 17956 | 22833.6 |
| 27 | 11 | 5 | 112.2 | 79 | 12588.84 | 6241 | 8863.8 |
| 28 | 11 | 2 | 164.4 | 156 | 27027.36 | 24336 | 25646.4 |
| 29 | 11 | 0 | 168.8 | 106 | 28493.44 | 11236 | 17892.8 |
| 30 | 11 | 3 | 90 | 82 | 8100 | 6724 | 7380 |
| 31 | 11 | 3 | 172.4 | 157 | 29721.76 | 24649 | 27066.8 |
| 33 | 11 | 13 | 186.1 | 180 | 34633.21 | 32400 | 33498 |
| 34 | 11 | 7 | 127.3 | 136 | 16205.29 | 18496 | 17312.8 |
| 35 | 11 | 1 | 115 | 75 | 13225 | 5625 | 8625 |
| 38 | 11 | 0 | 77 | 52 | 5929 | 2704 | 4004 |
| 39 | 11 | 2 | 159.1 | 108 | 25312.81 | 11664 | 17182.8 |
| 40 | 11 | 0 | 157.9 | 121 | 24932.41 | 14641 | 19105.9 |
| 42 | 11 | 0 | 168.4 | 114 | 28358.56 | 12996 | 19197.6 |
| 43 | 11 | 6 | 141 | 129 | 19881 | 16641 | 18189 |
| 44 | 11 | 8 | 160 | 154 | 25600 | 23716 | 24640 |
| 45 | 11 | 4 | 131.4 | 128 | 17265.96 | 16384 | 16819.2 |
| 46 | 11 | 0 | 9.8 | 2 | 96.04 | 4 | 19.6 |
| 47 | 11 | 0 | 52.7 | 28 | 2777.29 | 784 | 1475.6 |
| 48 | 11 | 1 | 158.4 | 84 | 25090.56 | 7056 | 13305.6 |
| 49 | 11 | 3 | 132.1 | 124 | 17450.41 | 15376 | 16380.4 |
| 50 | 11 | 2 | 129.2 | 79 | 16692.64 | 6241 | 10206.8 |
| 51 | 11 | 1 | 184.3 | 107 | 33966.49 | 11449 | 19720.1 |

| | | | | | | | |
|------|----|----|-------|-----|----------|-------|---------|
| 52 | 11 | 0 | 160.9 | 105 | 25888.81 | 11025 | 16894.5 |
| 53 | 11 | 4 | 197.2 | 124 | 38887.84 | 15376 | 24452.8 |
| 1020 | 11 | 0 | 172.6 | 108 | 29790.76 | 11664 | 18640.8 |
| 1021 | 11 | 1 | 182.7 | 110 | 33379.29 | 12100 | 20097 |
| 1022 | 11 | 5 | 136.4 | 96 | 18604.96 | 9216 | 13094.4 |
| 1023 | 11 | 0 | 196.8 | 85 | 38730.24 | 7225 | 16728 |
| 1024 | 11 | 3 | 174.7 | 143 | 30520.09 | 20449 | 24982.1 |
| 1025 | 11 | 4 | 205.3 | 146 | 42148.09 | 21316 | 29973.8 |
| 1026 | 11 | 4 | 165 | 113 | 27225 | 12769 | 18645 |
| 1027 | 11 | 0 | 39.9 | 31 | 1592.01 | 961 | 1236.9 |
| 1028 | 11 | 1 | 169.8 | 122 | 28832.04 | 14884 | 20715.6 |
| 1029 | 11 | 10 | 271 | 209 | 73441 | 43681 | 56639 |
| 1030 | 11 | 1 | 187.8 | 134 | 35268.84 | 17956 | 25165.2 |
| 1031 | 11 | 5 | 153.8 | 114 | 23654.44 | 12996 | 17533.2 |
| 1032 | 11 | 3 | 225.9 | 133 | 51030.81 | 17689 | 30044.7 |
| 1033 | 11 | 2 | 146 | 76 | 21316 | 5776 | 11096 |
| 1034 | 11 | 0 | 202 | 147 | 40804 | 21609 | 29694 |
| 1035 | 11 | 0 | 215.9 | 142 | 46612.81 | 20164 | 30657.8 |
| 1036 | 11 | 0 | 187.5 | 119 | 35156.25 | 14161 | 22312.5 |
| 1037 | 11 | 2 | 219.1 | 127 | 48004.81 | 16129 | 27825.7 |
| 1038 | 11 | 1 | 68.2 | 39 | 4651.24 | 1521 | 2659.8 |
| 1039 | 11 | 1 | 165.7 | 122 | 27456.49 | 14884 | 20215.4 |
| 1040 | 11 | 3 | 222.3 | 156 | 49417.29 | 24336 | 34678.8 |
| 1041 | 11 | 2 | 201.9 | 140 | 40763.61 | 19600 | 28266 |
| 1042 | 11 | 3 | 214.6 | 144 | 46053.16 | 20736 | 30902.4 |
| 1043 | 11 | 4 | 239.9 | 148 | 57552.01 | 21904 | 35505.2 |
| 1044 | 11 | 0 | 164.6 | 116 | 27093.16 | 13456 | 19093.6 |
| 1045 | 11 | 0 | 159.2 | 115 | 25344.64 | 13225 | 18308 |
| 1046 | 11 | 11 | 191.2 | 144 | 36557.44 | 20736 | 27532.8 |
| 1047 | 11 | 3 | 220.9 | 133 | 48796.81 | 17689 | 29379.7 |
| 1048 | 11 | 0 | 148.4 | 126 | 22022.56 | 15876 | 18698.4 |
| 1050 | 11 | 2 | 191.8 | 145 | 36787.24 | 21025 | 27811 |
| 1051 | 11 | 2 | 191.5 | 127 | 36672.25 | 16129 | 24320.5 |
| 1053 | 11 | 6 | 162.2 | 167 | 26308.84 | 27889 | 27087.4 |
| 1054 | 11 | 0 | 210.5 | 143 | 44310.25 | 20449 | 30101.5 |
| 1055 | 11 | 8 | 119.5 | 120 | 14280.25 | 14400 | 14340 |
| 2008 | 11 | 5 | 199.4 | 165 | 39760.36 | 27225 | 32901 |
| 2009 | 11 | 0 | 144 | 54 | 20736 | 2916 | 7776 |
| 2010 | 11 | 6 | 184.5 | 144 | 34040.25 | 20736 | 26568 |
| 2011 | 11 | 0 | 54 | 40 | 2916 | 1600 | 2160 |
| 2012 | 11 | 1 | 155.6 | 125 | 24211.36 | 15625 | 19450 |
| 2013 | 11 | 1 | 140.2 | 101 | 19656.04 | 10201 | 14160.2 |
| 2014 | 11 | 0 | 14 | 14 | 196 | 196 | 196 |
| 2015 | 11 | 0 | 158.5 | 114 | 25122.25 | 12996 | 18069 |

| | | | | | | | |
|------|----|---|-------|-----|----------|-------|---------|
| 2016 | 11 | 3 | 145.2 | 116 | 21083.04 | 13456 | 16843.2 |
| 2017 | 11 | 2 | 243.5 | 116 | 59292.25 | 13456 | 28246 |
| 2018 | 11 | 1 | 263.6 | 153 | 69484.96 | 23409 | 40330.8 |
| 2019 | 11 | 0 | 183.4 | 125 | 33635.56 | 15625 | 22925 |
| 2021 | 11 | 2 | 209.8 | 145 | 44016.04 | 21025 | 30421 |
| 2022 | 11 | 1 | 132.9 | 96 | 17662.41 | 9216 | 12758.4 |
| 2023 | 11 | 1 | 151.2 | 118 | 22861.44 | 13924 | 17841.6 |
| 2024 | 11 | 1 | 173.3 | 127 | 30032.89 | 16129 | 22009.1 |
| 2025 | 11 | 1 | 263.1 | 138 | 69221.61 | 19044 | 36307.8 |
| 2026 | 11 | 0 | 44.7 | 37 | 1998.09 | 1369 | 1653.9 |
| 2027 | 11 | 0 | 80.6 | 62 | 6496.36 | 3844 | 4997.2 |
| 2028 | 11 | 4 | 160.8 | 101 | 25856.64 | 10201 | 16240.8 |
| 2029 | 11 | 1 | 261.4 | 138 | 68329.96 | 19044 | 36073.2 |
| 2030 | 11 | 1 | 153.7 | 125 | 23623.69 | 15625 | 19212.5 |
| 2031 | 11 | 0 | 104.8 | 83 | 10983.04 | 6889 | 8698.4 |
| 2032 | 11 | 6 | 145.1 | 100 | 21054.01 | 10000 | 14510 |
| 2033 | 11 | 0 | 170 | 155 | 28900 | 24025 | 26350 |
| 2034 | 11 | 8 | 144.6 | 113 | 20909.16 | 12769 | 16339.8 |
| 2035 | 11 | 4 | 154.9 | 123 | 23994.01 | 15129 | 19052.7 |
| 2036 | 11 | 1 | 168.2 | 122 | 28291.24 | 14884 | 20520.4 |
| 2037 | 11 | 3 | 173.4 | 103 | 30067.56 | 10609 | 17860.2 |
| 2038 | 11 | 1 | 276.9 | 142 | 76673.61 | 20164 | 39319.8 |
| 3010 | 11 | 6 | 228.1 | 164 | 52029.61 | 26896 | 37408.4 |
| 3011 | 11 | 1 | 184.9 | 127 | 34188.01 | 16129 | 23482.3 |
| 3012 | 11 | 8 | 260.5 | 150 | 67860.25 | 22500 | 39075 |
| 3013 | 11 | 0 | 236.6 | 111 | 55979.56 | 12321 | 26262.6 |
| 3014 | 11 | 1 | 226.5 | 135 | 51302.25 | 18225 | 30577.5 |
| 3015 | 11 | 0 | 190.7 | 128 | 36366.49 | 16384 | 24409.6 |
| 3016 | 11 | 7 | 196.2 | 143 | 38494.44 | 20449 | 28056.6 |
| 3017 | 11 | 1 | 71.4 | 86 | 5097.96 | 7396 | 6140.4 |
| 3018 | 11 | 3 | 115.2 | 90 | 13271.04 | 8100 | 10368 |
| 3019 | 11 | 3 | 153.5 | 114 | 23562.25 | 12996 | 17499 |
| 3020 | 11 | 4 | 174.9 | 137 | 30590.01 | 18769 | 23961.3 |
| 3022 | 11 | 4 | 150.9 | 116 | 22770.81 | 13456 | 17504.4 |
| 3023 | 11 | 2 | 104.5 | 91 | 10920.25 | 8281 | 9509.5 |
| 3024 | 11 | 2 | 249.5 | 133 | 62250.25 | 17689 | 33183.5 |
| 3025 | 11 | 2 | 242.9 | 151 | 59000.41 | 22801 | 36677.9 |
| 3026 | 11 | 2 | 258.3 | 166 | 66718.89 | 27556 | 42877.8 |
| 3027 | 11 | 3 | 194.5 | 141 | 37830.25 | 19881 | 27424.5 |
| 3028 | 11 | 1 | 84.4 | 65 | 7123.36 | 4225 | 5486 |
| 3029 | 11 | 1 | 37.8 | 17 | 1428.84 | 289 | 642.6 |
| 3030 | 11 | 1 | 17.7 | 18 | 313.29 | 324 | 318.6 |
| 3031 | 11 | 6 | 111.7 | 84 | 12476.89 | 7056 | 9382.8 |
| 3032 | 11 | 1 | 189.4 | 146 | 35872.36 | 21316 | 27652.4 |

| | | | | | | | |
|------|----|----|-------|-----|----------|-------|---------|
| 3033 | 11 | 2 | 186.2 | 151 | 34670.44 | 22801 | 28116.2 |
| 3034 | 11 | 4 | 188.7 | 142 | 35607.69 | 20164 | 26795.4 |
| 3035 | 11 | 0 | 124.8 | 106 | 15575.04 | 11236 | 13228.8 |
| 3036 | 11 | 4 | 225.8 | 166 | 50985.64 | 27556 | 37482.8 |
| 3037 | 11 | 2 | 169.5 | 115 | 28730.25 | 13225 | 19492.5 |
| 3038 | 11 | 1 | 195.5 | 152 | 38220.25 | 23104 | 29716 |
| 3039 | 11 | 4 | 271.1 | 153 | 73495.21 | 23409 | 41478.3 |
| 3040 | 11 | 0 | 155.9 | 138 | 24304.81 | 19044 | 21514.2 |
| 3041 | 11 | 0 | 185.2 | 114 | 34299.04 | 12996 | 21112.8 |
| 3042 | 11 | 4 | 229.9 | 171 | 52854.01 | 29241 | 39312.9 |
| 3043 | 11 | 3 | 131 | 104 | 17161 | 10816 | 13624 |
| 4001 | 11 | 2 | 180.1 | 114 | 32436.01 | 12996 | 20531.4 |
| 4002 | 11 | 8 | 161.6 | 119 | 26114.56 | 14161 | 19230.4 |
| 4003 | 11 | 0 | 76.7 | 28 | 5882.89 | 784 | 2147.6 |
| 4004 | 11 | 2 | 155 | 117 | 24025 | 13689 | 18135 |
| 4005 | 11 | 4 | 201.8 | 131 | 40723.24 | 17161 | 26435.8 |
| 4006 | 11 | 11 | 279.4 | 148 | 78064.36 | 21904 | 41351.2 |
| 4007 | 11 | 13 | 153.4 | 115 | 23531.56 | 13225 | 17641 |
| 4008 | 11 | 1 | 274.8 | 148 | 75515.04 | 21904 | 40670.4 |
| 4009 | 11 | 6 | 270.2 | 120 | 73008.04 | 14400 | 32424 |
| 4010 | 11 | 3 | 157.3 | 124 | 24743.29 | 15376 | 19505.2 |
| 4011 | 11 | 10 | 238.1 | 182 | 56691.61 | 33124 | 43334.2 |
| 4012 | 11 | 2 | 136.3 | 114 | 18577.69 | 12996 | 15538.2 |
| 4013 | 11 | 2 | 143 | 97 | 20449 | 9409 | 13871 |
| 4014 | 11 | 2 | 198.4 | 85 | 39362.56 | 7225 | 16864 |
| 4015 | 11 | 4 | 316 | 126 | 99856 | 15876 | 39816 |
| 4016 | 11 | 5 | 180.1 | 143 | 32436.01 | 20449 | 25754.3 |
| 4017 | 11 | 0 | 156.8 | 100 | 24586.24 | 10000 | 15680 |
| 4018 | 11 | 12 | 199.3 | 154 | 39720.49 | 23716 | 30692.2 |
| 4019 | 11 | 9 | 158.3 | 126 | 25058.89 | 15876 | 19945.8 |
| 4020 | 11 | 2 | 142.1 | 116 | 20192.41 | 13456 | 16483.6 |
| 5094 | 11 | 1 | 219.5 | 154 | 48180.25 | 23716 | 33803 |
| 5095 | 11 | 3 | 55.6 | 44 | 3091.36 | 1936 | 2446.4 |
| 5096 | 11 | 0 | 328.5 | 91 | 107912.3 | 8281 | 29893.5 |
| 5097 | 11 | 0 | 514 | 129 | 264196 | 16641 | 66306 |
| 5098 | 11 | 2 | 151.6 | 128 | 22982.56 | 16384 | 19404.8 |
| 5099 | 11 | 4 | 200.3 | 137 | 40120.09 | 18769 | 27441.1 |
| 5100 | 11 | 0 | 489.9 | 129 | 240002 | 16641 | 63197.1 |
| 5101 | 11 | 0 | 117 | 97 | 13689 | 9409 | 11349 |
| 5102 | 11 | 2 | 154.5 | 124 | 23870.25 | 15376 | 19158 |
| 5103 | 11 | 2 | 201.3 | 128 | 40521.69 | 16384 | 25766.4 |
| 5104 | 11 | 8 | 118.8 | 119 | 14113.44 | 14161 | 14137.2 |
| 5105 | 11 | 0 | 627.2 | 155 | 393379.8 | 24025 | 97216 |
| 5106 | 11 | 1 | 539.7 | 130 | 291276.1 | 16900 | 70161 |

| | | | | | | | |
|------|----|---|-------|-----|----------|-------|---------|
| 5107 | 11 | 0 | 521 | 132 | 271441 | 17424 | 68772 |
| 5108 | 11 | 2 | 324.8 | 113 | 105495 | 12769 | 36702.4 |
| 5110 | 11 | 2 | 205.1 | 140 | 42066.01 | 19600 | 28714 |
| 5111 | 11 | 1 | 177.2 | 134 | 31399.84 | 17956 | 23744.8 |
| 5112 | 11 | 0 | 288.3 | 146 | 83116.89 | 21316 | 42091.8 |
| 5113 | 11 | 6 | 276.7 | 203 | 76562.89 | 41209 | 56170.1 |
| 5114 | 11 | 4 | 127.3 | 94 | 16205.29 | 8836 | 11966.2 |
| 5115 | 11 | 3 | 140.9 | 118 | 19852.81 | 13924 | 16626.2 |
| 5117 | 11 | 0 | 324 | 108 | 104976 | 11664 | 34992 |
| 5118 | 11 | 0 | 504.6 | 122 | 254621.2 | 14884 | 61561.2 |
| 5119 | 11 | 2 | 236.5 | 101 | 55932.25 | 10201 | 23886.5 |
| 5120 | 11 | 5 | 148 | 120 | 21904 | 14400 | 17760 |
| 5121 | 11 | 1 | 280.7 | 183 | 78792.49 | 33489 | 51368.1 |
| 5122 | 11 | 2 | 157.6 | 112 | 24837.76 | 12544 | 17651.2 |
| 5123 | 11 | 6 | 223.3 | 153 | 49862.89 | 23409 | 34164.9 |
| 5124 | 11 | 3 | 373.2 | 115 | 139278.2 | 13225 | 42918 |
| 5125 | 11 | 0 | 384.2 | 143 | 147609.6 | 20449 | 54940.6 |
| 5126 | 11 | 3 | 176.9 | 135 | 31293.61 | 18225 | 23881.5 |
| 5127 | 11 | 2 | 131.4 | 108 | 17265.96 | 11664 | 14191.2 |
| 5128 | 11 | 2 | 140.5 | 129 | 19740.25 | 16641 | 18124.5 |
| 6143 | 11 | 0 | 126.2 | 100 | 15926.44 | 10000 | 12620 |
| 6144 | 11 | 1 | 135.8 | 98 | 18441.64 | 9604 | 13308.4 |
| 6145 | 11 | 4 | 179.8 | 104 | 32328.04 | 10816 | 18699.2 |
| 6146 | 11 | 0 | 174 | 123 | 30276 | 15129 | 21402 |
| 6147 | 11 | 4 | 147.5 | 122 | 21756.25 | 14884 | 17995 |
| 6148 | 11 | 3 | 145.6 | 104 | 21199.36 | 10816 | 15142.4 |
| 6149 | 11 | 3 | 206.3 | 126 | 42559.69 | 15876 | 25993.8 |
| 6150 | 11 | 4 | 263.6 | 173 | 69484.96 | 29929 | 45602.8 |
| 6151 | 11 | 4 | 219.2 | 139 | 48048.64 | 19321 | 30468.8 |
| 6152 | 11 | 1 | 199.6 | 131 | 39840.16 | 17161 | 26147.6 |
| 6154 | 11 | 0 | 92.1 | 64 | 8482.41 | 4096 | 5894.4 |
| 6155 | 11 | 2 | 246.5 | 145 | 60762.25 | 21025 | 35742.5 |
| 6156 | 11 | 1 | 150.7 | 92 | 22710.49 | 8464 | 13864.4 |
| 6157 | 11 | 0 | 54.9 | 46 | 3014.01 | 2116 | 2525.4 |
| 6158 | 11 | 4 | 208 | 126 | 43264 | 15876 | 26208 |
| 6159 | 11 | 2 | 101.4 | 47 | 10281.96 | 2209 | 4765.8 |
| 6160 | 11 | 0 | 28.6 | 24 | 817.96 | 576 | 686.4 |
| 6161 | 11 | 0 | 64.2 | 39 | 4121.64 | 1521 | 2503.8 |
| 6162 | 11 | 4 | 190.5 | 126 | 36290.25 | 15876 | 24003 |
| 6163 | 11 | 4 | 158.5 | 121 | 25122.25 | 14641 | 19178.5 |
| 6164 | 11 | 2 | 92.2 | 70 | 8500.84 | 4900 | 6454 |
| 6165 | 11 | 0 | 113.6 | 89 | 12904.96 | 7921 | 10110.4 |
| 6166 | 11 | 1 | 102.7 | 80 | 10547.29 | 6400 | 8216 |
| 6167 | 11 | 0 | 188 | 75 | 35344 | 5625 | 14100 |

| | | | | | | | |
|------|----|---|-------|-----|----------|-------|---------|
| 6168 | 11 | 7 | 111.2 | 100 | 12365.44 | 10000 | 11120 |
| 6169 | 11 | 0 | 153.3 | 109 | 23500.89 | 11881 | 16709.7 |
| 6170 | 11 | 2 | 126.7 | 84 | 16052.89 | 7056 | 10642.8 |
| 6171 | 11 | 4 | 125.8 | 70 | 15825.64 | 4900 | 8806 |
| 6172 | 11 | 0 | 88.5 | 73 | 7832.25 | 5329 | 6460.5 |
| 6173 | 11 | 1 | 104.4 | 63 | 10899.36 | 3969 | 6577.2 |
| 6174 | 11 | 0 | 150.9 | 104 | 22770.81 | 10816 | 15693.6 |
| 6175 | 11 | 0 | 200.5 | 26 | 40200.25 | 676 | 5213 |
| 6176 | 11 | 2 | 133 | 70 | 17689 | 4900 | 9310 |
| 6177 | 11 | 1 | 139 | 103 | 19321 | 10609 | 14317 |
| 6178 | 11 | 1 | 102 | 97 | 10404 | 9409 | 9894 |
| 6179 | 11 | 0 | 167.4 | 156 | 28022.76 | 24336 | 26114.4 |
| 6180 | 11 | 2 | 108.4 | 99 | 11750.56 | 9801 | 10731.6 |
| 8468 | 11 | 1 | 179.2 | 129 | 32112.64 | 16641 | 23116.8 |
| 8469 | 11 | 0 | 97.9 | 88 | 9584.41 | 7744 | 8615.2 |
| 8470 | 11 | 7 | 131.1 | 115 | 17187.21 | 13225 | 15076.5 |
| 8471 | 11 | 4 | 170.5 | 125 | 29070.25 | 15625 | 21312.5 |
| 8473 | 11 | 0 | 175.4 | 130 | 30765.16 | 16900 | 22802 |
| 8474 | 11 | 0 | 168.9 | 128 | 28527.21 | 16384 | 21619.2 |
| 8475 | 11 | 0 | 108.1 | 94 | 11685.61 | 8836 | 10161.4 |
| 8476 | 11 | 0 | 199.8 | 153 | 39920.04 | 23409 | 30569.4 |
| 8478 | 11 | 0 | 124 | 87 | 15376 | 7569 | 10788 |
| 8479 | 11 | 0 | 31.7 | 24 | 1004.89 | 576 | 760.8 |
| 8480 | 11 | 3 | 182.1 | 139 | 33160.41 | 19321 | 25311.9 |
| 8483 | 11 | 0 | 178.3 | 110 | 31790.89 | 12100 | 19613 |
| 8485 | 11 | 0 | 72 | 47 | 5184 | 2209 | 3384 |
| 8486 | 11 | 1 | 138.5 | 106 | 19182.25 | 11236 | 14681 |
| 8487 | 11 | 7 | 170 | 139 | 28900 | 19321 | 23630 |
| 8488 | 11 | 0 | 152.1 | 119 | 23134.41 | 14161 | 18099.9 |
| 8489 | 11 | 2 | 188.9 | 138 | 35683.21 | 19044 | 26068.2 |
| 8490 | 11 | 5 | 227.2 | 141 | 51619.84 | 19881 | 32035.2 |
| 8491 | 11 | 1 | 73.6 | 52 | 5416.96 | 2704 | 3827.2 |
| 8492 | 11 | 1 | 212.9 | 154 | 45326.41 | 23716 | 32786.6 |
| 8493 | 11 | 1 | 129.9 | 90 | 16874.01 | 8100 | 11691 |
| 8494 | 11 | 0 | 86.8 | 67 | 7534.24 | 4489 | 5815.6 |
| 8496 | 11 | 0 | 182.6 | 139 | 33342.76 | 19321 | 25381.4 |
| 8497 | 11 | 0 | 89.8 | 71 | 8064.04 | 5041 | 6375.8 |
| 8498 | 11 | 0 | 109.6 | 84 | 12012.16 | 7056 | 9206.4 |
| 8499 | 11 | 2 | 141.1 | 113 | 19909.21 | 12769 | 15944.3 |
| 8500 | 11 | 0 | 142.7 | 105 | 20363.29 | 11025 | 14983.5 |
| 8502 | 11 | 0 | 109.4 | 75 | 11968.36 | 5625 | 8205 |
| 8503 | 11 | 5 | 148.9 | 109 | 22171.21 | 11881 | 16230.1 |
| 8504 | 11 | 1 | 122.4 | 101 | 14981.76 | 10201 | 12362.4 |
| 8505 | 11 | 9 | 152.7 | 122 | 23317.29 | 14884 | 18629.4 |

| | | | | | | | |
|------|----|---|-------|-----|----------|-------|---------|
| 8506 | 11 | 1 | 251.8 | 105 | 63403.24 | 11025 | 26439 |
| 8507 | 11 | 6 | 220.3 | 141 | 48532.09 | 19881 | 31062.3 |
| 8508 | 11 | 4 | 161.1 | 124 | 25953.21 | 15376 | 19976.4 |
| 8509 | 11 | 1 | 138.9 | 119 | 19293.21 | 14161 | 16529.1 |
| 8510 | 11 | 0 | 147.5 | 114 | 21756.25 | 12996 | 16815 |
| 8511 | 11 | 0 | 119.9 | 97 | 14376.01 | 9409 | 11630.3 |
| 8512 | 11 | 2 | 163.3 | 131 | 26666.89 | 17161 | 21392.3 |
| 8513 | 11 | 0 | 195.3 | 122 | 38142.09 | 14884 | 23826.6 |
| 8514 | 11 | 3 | 144.1 | 129 | 20764.81 | 16641 | 18588.9 |
| 8515 | 11 | 0 | 155.4 | 108 | 24149.16 | 11664 | 16783.2 |
| 8516 | 11 | 3 | 199.4 | 146 | 39760.36 | 21316 | 29112.4 |
| 8517 | 11 | 4 | 202.1 | 131 | 40844.41 | 17161 | 26475.1 |
| 8518 | 11 | 1 | 247.7 | 154 | 61355.29 | 23716 | 38145.8 |
| 8519 | 11 | 1 | 178.1 | 132 | 31719.61 | 17424 | 23509.2 |
| 8520 | 11 | 0 | 116.4 | 79 | 13548.96 | 6241 | 9195.6 |
| 8521 | 11 | 0 | 204.5 | 137 | 41820.25 | 18769 | 28016.5 |
| 8522 | 11 | 0 | 133 | 104 | 17689 | 10816 | 13832 |
| 8523 | 11 | 0 | 171.3 | 127 | 29343.69 | 16129 | 21755.1 |
| 8525 | 11 | 0 | 167.2 | 127 | 27955.84 | 16129 | 21234.4 |
| 8527 | 11 | 0 | 56.6 | 41 | 3203.56 | 1681 | 2320.6 |
| 8528 | 11 | 1 | 230.2 | 152 | 52992.04 | 23104 | 34990.4 |
| 8529 | 11 | 1 | 186.5 | 144 | 34782.25 | 20736 | 26856 |
| 8530 | 11 | 2 | 128.1 | 88 | 16409.61 | 7744 | 11272.8 |
| 8531 | 11 | 0 | 212.4 | 134 | 45113.76 | 17956 | 28461.6 |
| 8532 | 11 | 0 | 68 | 43 | 4624 | 1849 | 2924 |
| 8533 | 11 | 2 | 189 | 125 | 35721 | 15625 | 23625 |
| 8535 | 11 | 4 | 122.2 | 85 | 14932.84 | 7225 | 10387 |
| 8536 | 11 | 0 | 201.7 | 127 | 40682.89 | 16129 | 25615.9 |
| 8537 | 11 | 0 | 74.5 | 52 | 5550.25 | 2704 | 3874 |
| 8538 | 11 | 2 | 164.3 | 125 | 26994.49 | 15625 | 20537.5 |
| 8539 | 11 | 0 | 164.9 | 124 | 27192.01 | 15376 | 20447.6 |
| 8541 | 11 | 0 | 179 | 113 | 32041 | 12769 | 20227 |
| 8542 | 11 | 2 | 140.6 | 126 | 19768.36 | 15876 | 17715.6 |
| 8543 | 11 | 1 | 90.3 | 70 | 8154.09 | 4900 | 6321 |
| 8544 | 11 | 3 | 145.4 | 105 | 21141.16 | 11025 | 15267 |
| 8545 | 11 | 1 | 194.4 | 118 | 37791.36 | 13924 | 22939.2 |
| 8546 | 11 | 5 | 249.5 | 137 | 62250.25 | 18769 | 34181.5 |
| 8547 | 11 | 0 | 32.6 | 43 | 1062.76 | 1849 | 1401.8 |
| 8548 | 11 | 0 | 63.6 | 48 | 4044.96 | 2304 | 3052.8 |
| 8549 | 11 | 9 | 847 | 151 | 717409 | 22801 | 127897 |
| 8550 | 11 | 1 | 98.2 | 50 | 9643.24 | 2500 | 4910 |
| 9016 | 11 | 4 | 101.8 | 82 | 10363.24 | 6724 | 8347.6 |
| 9020 | 11 | 0 | 187.6 | 128 | 35193.76 | 16384 | 24012.8 |
| 9021 | 11 | 3 | 90.2 | 55 | 8136.04 | 3025 | 4961 |

| | | | | | | | |
|------|--------|-----|---------|-------|----------|-------|----------|
| 9022 | 11 | 3 | 210.8 | 119 | 44436.64 | 14161 | 25085.2 |
| 9025 | 11 | 4 | 346.2 | 123 | 119854.4 | 15129 | 42582.6 |
| 9026 | 11 | 2 | 154.9 | 136 | 23994.01 | 18496 | 21066.4 |
| 9029 | 11 | 6 | 139 | 121 | 19321 | 14641 | 16819 |
| 9030 | 11 | 2 | 116.4 | 110 | 13548.96 | 12100 | 12804 |
| 9034 | 11 | 5 | 173.5 | 120 | 30102.25 | 14400 | 20820 |
| 9035 | 11 | 2 | 201.5 | 167 | 40602.25 | 27889 | 33650.5 |
| 9036 | 11 | 1 | 116.1 | 89 | 13479.21 | 7921 | 10332.9 |
| 9037 | 11 | 2 | 165.3 | 126 | 27324.09 | 15876 | 20827.8 |
| 9041 | 11 | 1 | 82.9 | 48 | 6872.41 | 2304 | 3979.2 |
| 9042 | 11 | 0 | 143.7 | 96 | 20649.69 | 9216 | 13795.2 |
| 9046 | 11 | 2 | 173.7 | 147 | 30171.69 | 21609 | 25533.9 |
| 9049 | 11 | 1 | 1229.9 | 173 | 1512654 | 29929 | 212772.7 |
| 9050 | 11 | 2 | 267 | 172 | 71289 | 29584 | 45924 |
| 9053 | 11 | 3 | 24.6 | 10 | 605.16 | 100 | 246 |
| 9054 | 11 | 0 | 50.9 | 33 | 2590.81 | 1089 | 1679.7 |
| 9056 | 11 | 2 | 152.7 | 102 | 23317.29 | 10404 | 15575.4 |
| 9057 | 11 | 2 | 191.8 | 112 | 36787.24 | 12544 | 21481.6 |
| 9058 | 11 | 2 | 126.5 | 91 | 16002.25 | 8281 | 11511.5 |
| 9059 | 11 | 4 | 194.5 | 131 | 37830.25 | 17161 | 25479.5 |
| 9064 | 11 | 6 | 286.3 | 158 | 81967.69 | 24964 | 45235.4 |
| 9065 | 11 | 3 | 258.2 | 160 | 66667.24 | 25600 | 41312 |
| 9066 | 11 | 1 | 216.6 | 130 | 46915.56 | 16900 | 28158 |
| 9068 | 11 | 0 | 173.7 | 134 | 30171.69 | 17956 | 23275.8 |
| 9069 | 11 | 5 | 195.4 | 163 | 38181.16 | 26569 | 31850.2 |
| 9070 | 11 | 0 | 6.4 | 4 | 40.96 | 16 | 25.6 |
| 9072 | 11 | 5 | 179.5 | 133 | 32220.25 | 17689 | 23873.5 |
| 9073 | 11 | 2 | 139 | 93 | 19321 | 8649 | 12927 |
| 9074 | 11 | 1 | 227.9 | 152 | 51938.41 | 23104 | 34640.8 |
| 9075 | 11 | 7 | 185.4 | 132 | 34373.16 | 17424 | 24472.8 |
| 9076 | 11 | 5 | 166.7 | 124 | 27788.89 | 15376 | 20670.8 |
| 9077 | 11 | 0 | 135.2 | 112 | 18279.04 | 12544 | 15142.4 |
| 9078 | 11 | 2 | 174.3 | 97 | 30380.49 | 9409 | 16907.1 |
| 9079 | 11 | 0 | 132.1 | 85 | 17450.41 | 7225 | 11228.5 |
| 9080 | 11 | 10 | 272.4 | 175 | 74201.76 | 30625 | 47670 |
| | TOTALS | 794 | 60254.5 | 38666 | | | |

**Appendix D: Multiple Regression Results, Hypothesis Testing, and
Residual Calculations**

Table D-1. Multiple Regression Results at the Two-Digit Work Unit Code Level

| Two-Digit Work Unit Code Level | R-Squared | Durbin-Watson Statistic |
|---------------------------------------|------------------|--------------------------------|
| 11 | 0.0525 | 1.9574 |
| 12 | 0.0929 | 1.9420 |
| 13 | 0.2901 | 1.6189 |
| 14 | 0.0615 | 2.0383 |
| 23 | 0.2134 | 2.1051 |
| 24 | 0.0756 | 1.7635 |
| 41 | 0.0614 | 2.0368 |
| 42 | 0.0561 | 2.0909 |
| 44 | 0.1310 | 1.9953 |
| 45 | 0.0698 | 1.7640 |
| 46 | 0.1594 | 1.9672 |
| 47 | 0.0914 | 1.9924 |
| 49 | 0.0205 | 2.0391 |
| 51 | 0.1789 | 1.6510 |
| 52 | 0.0229 | 2.0665 |
| 55 | 0.0295 | 2.1670 |
| 57 | 0.0388 | 1.6796 |
| 63 | 0.1999 | 1.9403 |
| 65 | 0.1074 | 1.7577 |
| 71 | 0.1163 | 1.9307 |
| 74 | 0.3492 | 2.1225 |
| 75 | 0.0759 | 1.9376 |
| 76 | 0.2522 | 1.9698 |
| 97 | 0.0182 | 1.4665 |

Table D-2. Multiple Regression Results at the Three-Digit Work Unit Code Level

| Three-Digit Work Unit Code Level | R-Squared | Durbin-Watson Statistic |
|----------------------------------|-----------|-------------------------|
| 11A | 0.0230 | 2.1474 |
| 11G | 0.0061 | 1.7838 |
| 11K | 0.0206 | 1.9703 |
| 11P | 0.0234 | 1.9081 |
| 12C | 0.0220 | 2.2048 |
| 12E | 0.0728 | 2.1067 |
| 13A | 0.2412 | 1.6609 |
| 13B | 0.0724 | 2.0657 |
| 13D | 0.0260 | 2.1315 |
| 13F | 0.0084 | 1.8672 |
| 13H | 0.0058 | 1.9034 |
| 14A | 0.0426 | 2.0302 |
| 14C | 0.0131 | 2.0907 |
| 14D | 0.0336 | 1.9797 |
| 14G | 0.0067 | 2.0680 |
| 23I | 0.0946 | 1.8876 |
| 23F | 0.0586 | 2.2327 |
| 23H | 0.0446 | 2.2055 |
| 23K | 0.0373 | 2.1012 |
| 23P | 0.0309 | 1.8887 |
| 23Q | 0.0168 | 2.0020 |
| 23Z | 0.0430 | 2.0665 |
| 24A | 0.0249 | 1.7457 |
| 24B | 0.0433 | 1.8595 |
| 24D | 0.0247 | 1.9201 |
| 41A | 0.0608 | 2.2055 |
| 42A | 0.0501 | 2.0501 |
| 42C | 0.0002 | 2.0325 |
| 44A | 0.1307 | 1.9989 |
| 44B | 0.0245 | 2.1242 |
| 44E | 0.0048 | 2.0092 |
| 45A | 0.0390 | 1.8893 |
| 45B | 0.0078 | 1.8874 |
| 45C | 0.0447 | 1.8974 |
| 46A | 0.0694 | 2.0272 |
| 46B | 0.2429 | 2.1105 |
| 46E | 0.0768 | 2.0991 |
| 47A | 0.0914 | 1.9924 |
| 49A | 0.0084 | 2.0310 |
| 51A | 0.1307 | 1.9001 |

| | | |
|-----|--------|--------|
| 51E | 0.0396 | 1.9314 |
| 51M | 0.0583 | 1.8195 |
| 51N | 0.0492 | 1.5113 |
| 52A | 0.0229 | 2.0665 |
| 55A | 0.0131 | 2.0782 |
| 55B | 0.0133 | 2.2440 |
| 55C | 0.0125 | 1.9785 |
| 57A | 0.0364 | 1.6815 |
| 63A | 0.1480 | 1.9293 |
| 63B | 0.0961 | 2.0761 |
| 63E | 0.0156 | 2.1430 |
| 65A | 0.0572 | 2.0053 |
| 65B | 0.0717 | 1.7196 |
| 71A | 0.0126 | 2.0112 |
| 71C | 0.0244 | 1.9800 |
| 71F | 0.0619 | 1.8191 |
| 71M | 0.0367 | 1.9671 |
| 71Z | 0.0277 | 1.8990 |
| 74E | 0.0202 | 2.0803 |
| 74F | 0.2720 | 1.9823 |
| 74J | 0.0578 | 2.1457 |
| 74K | 0.0595 | 2.1790 |
| 74L | 0.0521 | 2.0844 |
| 75B | 0.0728 | 1.6988 |
| 75E | 0.0180 | 1.8724 |
| 75G | 0.0019 | 1.5684 |
| 75M | 0.0054 | 2.1472 |
| 75P | 0.0210 | 2.0404 |
| 76A | 0.1039 | 1.8724 |
| 76B | 0.0454 | 1.9337 |
| 76G | 0.1384 | 2.0169 |
| 76H | 0.2050 | 1.8456 |
| 76K | 0.0324 | 1.9537 |
| 97A | 0.0182 | 1.4665 |

Table D-3. Multiple Regression Results at the Five-Digit Work Unit Code Level

| Five-Digit Work Unit Code Level | R-Squared | Durbin-Watson Statistic |
|---------------------------------|-----------|-------------------------|
| 11A99 | 0.0045 | 2.0258 |
| 11AB0 | 0.0381 | 2.0695 |
| 13AK0 | 0.0664 | 1.3208 |
| 13AKA | 0.1344 | 1.6445 |
| 13AKB | 0.0412 | 1.1493 |
| 13BJ0 | 0.0253 | 1.7670 |
| 13BJA | 0.0188 | 2.0721 |
| 13BJB | 0.0610 | 1.6475 |
| 13DC0 | 0.0028 | 1.8629 |
| 14AAA | 0.0119 | 1.7964 |
| 14DDA | 0.0114 | 1.8753 |
| 231AA | 0.0073 | 2.0465 |
| 231AB | 0.0158 | 2.0510 |
| 231AM | 0.0389 | 1.9780 |
| 231FN | 0.0153 | 1.7092 |
| 23FBA | 0.0222 | 2.1064 |
| 23HAA | 0.0157 | 1.9451 |
| 23HAB | 0.0387 | 1.9737 |
| 23PAB | 0.0312 | 1.8756 |
| 23QAN | 0.0878 | 2.0846 |
| 23Z00 | 0.0430 | 2.0665 |
| 24AD0 | 0.0188 | 1.8656 |
| 24AN0 | 0.0251 | 1.8597 |
| 24BA0 | 0.0118 | 2.1282 |
| 24BAC | 0.0070 | 1.8751 |
| 24DAD | 0.0193 | 2.0084 |
| 41AEH | 0.0141 | 1.9702 |
| 42AD0 | 0.0248 | 1.9303 |
| 42ADA | 0.0220 | 1.8653 |
| 42ADB | 0.0252 | 1.9858 |
| 42AF0 | 0.0219 | 1.6833 |
| 44A99 | 0.0249 | 2.0473 |
| 44AAA | 0.0338 | 1.9766 |
| 44AAC | 0.0401 | 1.8672 |
| 44AAL | 0.0170 | 1.7640 |
| 44AAY | 0.0246 | 1.9899 |
| 46EBB | 0.0109 | 1.9957 |
| 47AAH | 0.0456 | 2.0320 |
| 47AAX | 0.0306 | 2.0348 |

| | | |
|-------|--------|--------|
| 51AD0 | 0.0059 | 2.0632 |
| 51AE0 | 0.0148 | 2.1236 |
| 51AF0 | 0.1185 | 1.8077 |
| 51AJ0 | 0.0342 | 1.9295 |
| 51AK0 | 0.0371 | 2.1234 |
| 51EA0 | 0.0123 | 1.8769 |
| 51ED0 | 0.0189 | 1.8679 |
| 51EF0 | 0.0395 | 2.0289 |
| 51MA0 | 0.0655 | 1.7841 |
| 51NA0 | 0.0385 | 1.6626 |
| 51NB0 | 0.0173 | 2.0315 |
| 52AA0 | 0.0244 | 2.0084 |
| 52AB0 | 0.0132 | 2.0423 |
| 52AC0 | 0.0066 | 1.9469 |
| 55AE0 | 0.0095 | 2.0332 |
| 55BC0 | 0.0111 | 2.1136 |
| 55BE0 | 0.0101 | 2.2148 |
| 55CB0 | 0.0117 | 1.9678 |
| 57AC0 | 0.0236 | 1.4634 |
| 63AD0 | 0.0651 | 1.9364 |
| 63AN0 | 0.0138 | 2.0816 |
| 63AT0 | 0.0127 | 2.1176 |
| 63AV0 | 0.0765 | 1.8931 |
| 63BH0 | 0.0266 | 1.9878 |
| 63BJ0 | 0.0749 | 1.8576 |
| 65AA0 | 0.0493 | 1.9606 |
| 65AB0 | 0.0379 | 2.0633 |
| 65BA0 | 0.0239 | 2.0292 |
| 65BB0 | 0.0449 | 1.9590 |
| 65BC0 | 0.0203 | 1.8960 |
| 65BH0 | 0.0391 | 1.5730 |
| 71AK0 | 0.0105 | 2.0557 |
| 71CA0 | 0.0265 | 1.9402 |
| 71FA0 | 0.0626 | 1.9594 |
| 71FB0 | 0.0465 | 1.7997 |
| 71FE0 | 0.0043 | 2.0911 |
| 71MA0 | 0.0375 | 2.0677 |
| 71ZA0 | 0.0447 | 2.0573 |
| 71ZF0 | 0.0026 | 1.7985 |
| 74EB0 | 0.0202 | 2.0803 |
| 74FA0 | 0.1127 | 2.0015 |
| 74FC0 | 0.0762 | 2.0209 |

| | | |
|-------|--------|--------|
| 74FH0 | 0.0578 | 1.9339 |
| 74FJ0 | 0.0973 | 2.0292 |
| 74FQ0 | 0.0468 | 2.1193 |
| 74FS0 | 0.1247 | 1.8172 |
| 74FU0 | 0.0728 | 1.9853 |
| 74FY0 | 0.0452 | 2.0396 |
| 74JA0 | 0.0451 | 2.0370 |
| 74JE0 | 0.0009 | 2.2456 |
| 74KA0 | 0.0647 | 2.1417 |
| 74KC0 | 0.0226 | 2.1947 |
| 74LB0 | 0.0558 | 1.9746 |
| 75BB0 | 0.0263 | 1.8835 |
| 75BD0 | 0.0369 | 1.7991 |
| 75BH0 | 0.0178 | 1.7788 |
| 75EB0 | 0.0180 | 1.8724 |
| 75GA0 | 0.0018 | 1.5485 |
| 75MA0 | 0.0033 | 2.1269 |
| 75MC0 | 0.0057 | 2.1082 |
| 75PA0 | 0.0210 | 2.0404 |
| 76AA0 | 0.0374 | 1.8559 |
| 76AC0 | 0.0733 | 1.9720 |
| 76AD0 | 0.0412 | 2.0224 |
| 76AG0 | 0.0153 | 1.9021 |
| 76BA0 | 0.0364 | 1.9692 |
| 76BB0 | 0.0307 | 1.9290 |
| 76BD0 | 0.0087 | 2.0252 |
| 76GF0 | 0.1131 | 2.0127 |
| 76HA0 | 0.0251 | 1.8808 |
| 76HB0 | 0.2288 | 1.9382 |
| 76HF0 | 0.0101 | 1.8524 |
| 76HG0 | 0.1929 | 2.1541 |
| 76KA0 | 0.0051 | 2.0164 |
| 76KC0 | 0.0254 | 1.7922 |
| 97ABD | 0.0160 | 1.8016 |

Reduced and Full Model Multiple Regressions and Hypothesis Testing for Two Digit Work Unit Code 13:

Unweighted Least Squares Linear Regression of Demands (Reduced Model)

| Predictor Variables | Coefficient | P-Value |
|---------------------|-------------|---------|
| CONSTANT | 0.80062 | 0.2375 |
| CUMFH | -0.02052 | 0.0000 |
| NSORT | 0.07344 | 0.0000 |

R- Squared: 0.2901; Adjusted R-Squared: 0.2843; Sum of Squares Error (SSE1): 2875.87

Hypothesis Testing: Ho: $\beta_1 = \beta_2 = \dots = \beta_k = 0$

Ha: at least one $\beta_i \neq 0$ ($i = 1, \dots, k$)

$$R^2 = 0.2901$$

n = number of data points = 247

k = number of carriers = 2

$$\text{Test Statistic Value: } f = \frac{R^2 / k}{(1 - R^2) / [n - (k + 1)]} = \frac{0.2901 / 2}{(1 - 0.2901) / [247 - (2 + 1)]} = 49.86$$

Rejection region for a level $\alpha = 0.05$ test = $f \geq F_{\alpha, k, n - (k + 1)} = f \geq F_{0.05, 2, 244} = 3.00$ (Table A-4, Neter and Wasserman, 1974:812)

Since $49.86 \gg 3.00$, the null hypothesis is rejected and the conclusion is demands/ maintenance actions is linearly related to at least one of the predictor variables.

Now, the test to determine if additional carriers can improve the reduced model:

Unweighted Least Squares Linear Regression of Demands (Full Model)

| Predictor Variables | Coefficient | P-Value |
|---------------------|-------------|---------|
| CONSTANT | 0.47398 | 0.6997 |
| CUMFH | -3.915E-04 | 0.9832 |
| NSORT | 0.04954 | 0.1658 |
| CUMFH2 | -3.519E-05 | 0.4683 |
| NSORT2 | 3.838E-05 | 0.8839 |
| CUMFHNSORT | 1.398E-05 | 0.9496 |

R-Squared: 0.2954; Adjusted R-Squared: 0.2808

Sum of Squares Error (SSE2): 2854.50; Residual Mean Square (MSE2): 11.8444

Hypothesis Testing:

Ho: model is $Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \epsilon$ (reduced model)

Ha: model is $Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \beta_3 x_1^2 + \beta_4 x_2^2 + \beta_5 x_1 x_2 + \epsilon$ (full model)

The number of points, n, = 247

The number of regressors in full model, k, = 5; therefore $(n - (k + 1)) = 247 - 6 = 241$

The number of additional regressor variables, $p, = 3$

$$\text{Test Statistic Value: } f = \frac{(SSE1 - SSE2) / p}{MSE2} = \frac{(2875.87 - 2854.50) / 3}{11.8444} = 0.6014$$

Rejection region for a level $\alpha = 0.05$ test = $f \geq F_{\alpha, k, n-(k+1)} = f \geq F_{0.05, 2, 241} = 2.60$ (Table A-4, Neter and Wasserman, 1974:812)

Since $0.6014 < 2.60$, the decision is to accept the null hypothesis and conclude additional variables do not contribute to the model. Therefore, the reduced model will be used to calculate a residual.

Based on p-values for the reduced model, CUMFH and NSORT contribute to the reduced model (The p-values are less than $\alpha = 0.05$). Thus, the reduced model takes the following form: $Y = -0.02052x_1 + 0.07344x_2$. Y represents the expected number of demands or maintenance actions. x_1 represents cumulative flying hours and x_2 represents the number of sorties.

Residual computation: The cumulative flying hour total from the 1994 validation data set is 60,254.5 and the total number of sorties is 38,666. Substituting these values into the reduced model yields $Y = -0.02052(60,254.5) + 0.07344(38,666) = 1,321.22$ or approximately 1,321 expected demands or maintenance actions for 60,254.5 flying hours and 38,666 sorties. The actual number of 13 demands or maintenance actions from the 1994 data is 2,092. Therefore, the residual on the reduced multiple regression model is $2,092 - 1,321 = 771$.

Reduced and Full Model Multiple Regressions and Hypothesis Testing for Two Digit Work Unit Code 23:

Unweighted Least Squares Linear Regression of Demands (Reduced Model)

| Predictor Variables | Coefficient | P-Value |
|---------------------|-------------|---------|
| CONSTANT | -0.04772 | 0.9320 |
| CUMFH | -0.00683 | 0.0317 |
| NSORT | 0.04257 | 0.0000 |

R- Squared: 0.2134; Adjusted R-Squared: 0.2070; Sum of Squares Error (SSE1): 1964.91

Hypothesis Testing: Ho: $\beta_1 = \beta_2 = \dots = \beta_k = 0$

Ha: at least one $\beta_i \neq 0$ ($i = 1, \dots, k$)

$$R^2 = 0.2134$$

n = number of data points = 247

k = number of carriers = 2

$$\text{Test Statistic Value: } f = \frac{R^2 / k}{(1 - R^2) / [n - (k + 1)]} = \frac{0.2134 / 2}{(1 - 0.2134) / [247 - (2 + 1)]} = 33.10$$

Rejection region for a level $\alpha = 0.05$ test = $f \geq F_{\alpha, k, n-(k+1)} = f \geq F_{0.05, 2, 244} = 3.00$ (Table A-4, Neter and Wasserman, 1974:812)

Since $33.10 \gg 3.00$, the null hypothesis is rejected and the conclusion is demands/ maintenance actions is linearly related to at least one of the predictor variables.

Now, the test to determine if additional carriers can improve the reduced model:

Unweighted Least Squares Linear Regression of Demands (Full Model)

| Predictor Variables | Coefficient | P-Value |
|---------------------|-------------|---------|
| CONSTANT | 1.48347 | 0.1392 |
| CUMFH | 0.01328 | 0.3811 |
| NSORT | -0.01725 | 0.5530 |
| CUMFH2 | -7.609E-05 | 0.0551 |
| NSORT2 | -4.375E-05 | 0.8382 |
| CUMFHNSORT | 1.757E-05 | 0.3304 |

R-Squared: 0.2415; Adjusted R-Squared: 0.2258

Sum of Squares Error (SSE2): 1894.71; Residual Mean Square (MSE2): 7.86188

Hypothesis Testing:

Ho: model is $Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \epsilon$ (reduced model)

Ha: model is $Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \beta_3 x_1^2 + \beta_4 x_2^2 + \beta_5 x_1 x_2 + \epsilon$ (full model)

The number of points, n, = 247

The number of regressors in full model, k, = 5; therefore $(n-(k+1)) = 247 - 6 = 241$

The number of additional regressor variables, p , = 3

$$\text{Test Statistic Value: } f = \frac{(SSE1 - SSE2) / p}{MSE2} = \frac{(1964.91 - 1894.71) / 3}{7.86188} = 2.97$$

Rejection region for a level $\alpha = 0.05$ test = $f \geq F_{\alpha, k, n-(k+1)} = f \geq F_{0.05, 2, 241} = 2.60$ (Table A-4, Neter and Wasserman, 1974:812)

Since $2.97 > 2.60$, the decision is to reject the null hypothesis and conclude additional variables do contribute to the model. Therefore, the full model will be used to calculate a residual.

However, based on p-values for the full model, none of the predictor variables contribute to the full model (None of the p-values are less than $\alpha = 0.05$). Thus, a residual will not be calculated on the full model.

Reduced and Full Model Multiple Regressions and Hypothesis Testing for Two Digit Work Unit Code 63:

Unweighted Least Squares Linear Regression of Demands (Reduced Model)

| Predictor Variables | Coefficient | P-Value |
|---------------------|-------------|---------|
| CONSTANT | -0.29133 | 0.5301 |
| CUMFH | -0.00584 | 0.0266 |
| NSORT | 0.03435 | 0.0000 |

R- Squared: 0.1999; Adjusted R-Squared: 0.1933; Sum of Squares Error (SSE1): 1350.48

Hypothesis Testing: Ho: $\beta_1 = \beta_2 = \dots = \beta_k = 0$

Ha: at least one $\beta_i \neq 0$ ($i = 1, \dots, k$)

$$R^2 = 0.1999$$

n = number of data points = 247

k = number of carriers = 2

$$\text{Test Statistic Value: } f = \frac{R^2 / k}{(1 - R^2) / [n - (k + 1)]} = \frac{0.1999 / 2}{(1 - 0.1999) / [247 - (2 + 1)]} = 30.48$$

Rejection region for a level $\alpha = 0.05$ test = $f \geq F_{\alpha, k, n - (k + 1)} = f \geq F_{0.05, 2, 244} = 3.00$ (Table A-4, Neter and Wasserman, 1974:812)

Since $30.48 \gg 3.00$, the null hypothesis is rejected and the conclusion is demands/ maintenance actions is linearly related to at least one of the predictor variables.

Now, the test to determine if additional carriers can improve the reduced model:

Unweighted Least Squares Linear Regression of Demands (Full Model)

| Predictor Variables | Coefficient | P-Value |
|---------------------|-------------|---------|
| CONSTANT | 0.47760 | 0.5711 |
| CUMFH | -0.00105 | 0.9344 |
| NSORT | 0.01220 | 0.6183 |
| CUMFH2 | -3.863E-08 | 0.9991 |
| NSORT2 | 1.091E-05 | 0.5454 |
| CUMFHNSORT | -3.096E-05 | 0.8385 |

R-Squared: 0.2042; Adjusted R-Squared: 0.1877

Sum of Squares Error (SSE2): 1343.20; Residual Mean Square (MSE2): 5.57344

Hypothesis Testing:

Ho: model is $Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \epsilon$ (reduced model)

Ha: model is $Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \beta_3 x_1^2 + \beta_4 x_2^2 + \beta_5 x_1 x_2 + \epsilon$ (full model)

The number of points, n, = 247

The number of regressors in full model, k, = 5; therefore $(n - (k + 1)) = 247 - 6 = 241$

The number of additional regressor variables, p , = 3

$$\text{Test Statistic Value: } f = \frac{(SSE1 - SSE2) / p}{MSE2} = \frac{(1350.48 - 1343.20) / 3}{5.57344} = 1.306$$

Rejection region for a level $\alpha = 0.05$ test = $f \geq F_{\alpha, k, n-(k+1)} = f \geq F_{0.05, 2, 241} = 2.60$ (Table A-4, Neter and Wasserman, 1974:812)

Since $1.306 < 2.60$, the decision is to accept the null hypothesis and conclude additional variables do not contribute to the model. Therefore, the reduced model will be used to calculate a residual.

Based on p-values for the reduced model, the CUMFH and NSORT predictor variables contribute to the reduced model (the p-values are less than $\alpha = 0.05$). The reduced model takes the following form: $Y = -0.00584x_1 + 0.03435x_2$. Y represents the expected number of demands or maintenance actions. x_1 represents cumulative flying hours and x_2 represents the number of sorties.

Residual computation: The cumulative flying hour total from the 1994 validation data set is 60,254.5 and the total number of sorties is 38,666. Substituting these values into the reduced model yields $Y = -0.00584(60,254.5) + 0.03435(38,666) = 976.29$ or approximately 976 expected demands or maintenance actions for 60,254.5 flying hours and 38,666 sorties. The actual number of 63 demands or maintenance actions from the 1994 data is 1,057. Therefore, the residual on the reduced multiple regression model is $1,057 - 976 = 81$.

Reduced and Full Model Multiple Regressions and Hypothesis Testing for Two Digit Work Unit Code 74:

Unweighted Least Squares Linear Regression of Demands (Reduced Model)

| Predictor Variables | Coefficient | P-Value |
|---------------------|-------------|---------|
| CONSTANT | -0.32329 | 0.7532 |
| CUMFH | -0.01133 | 0.0521 |
| NSORT | 0.10203 | 0.0000 |

R- Squared: 0.3492; Adjusted R-Squared: 0.3439; Sum of Squares Error (SSE1): 6636.40

Hypothesis Testing: Ho: $\beta_1 = \beta_2 = \dots = \beta_k = 0$

Ha: at least one $\beta_i \neq 0$ ($i = 1, \dots, k$)

$$R^2 = 0.3492$$

n = number of data points = 247

k = number of carriers = 2

$$\text{Test Statistic Value: } f = \frac{R^2 / k}{(1 - R^2) / [n - (k + 1)]} = \frac{0.3492 / 2}{(1 - 0.3492) / [247 - (2 + 1)]} = 65.47$$

Rejection region for a level $\alpha = 0.05$ test = $f \geq F_{\alpha, k, n - (k + 1)} = f \geq F_{0.05, 2, 244} = 3.00$ (Table A-4, Neter and Wasserman, 1974:812)

Since $65.47 \gg 3.00$, the null hypothesis is rejected and the conclusion is demands/maintenance actions is linearly related to at least one of the predictor variables.

Now, the test to determine if additional carriers can improve the reduced model:

Unweighted Least Squares Linear Regression of Demands (Full Model)

| Predictor Variables | Coefficient | P-Value |
|---------------------|-------------|---------|
| CONSTANT | 1.59691 | 0.3924 |
| CUMFH | -0.02601 | 0.3572 |
| NSORT | 0.08691 | 0.1095 |
| CUMFH2 | -1.743E-05 | 0.8130 |
| NSORT2 | -4.475E-05 | 0.9108 |
| CUMFHNSORT | 1.603E-05 | 0.6335 |

R-Squared: 0.3546; Adjusted R-Squared: 0.3412

Sum of Squares Error (SSE2): 6581.36; Residual Mean Square (MSE2): 27.3086

Hypothesis Testing:

Ho: model is $Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \epsilon$ (reduced model)

Ha: model is $Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \beta_3 x_1^2 + \beta_4 x_2^2 + \beta_5 x_1 x_2 + \epsilon$ (full model)

The number of points, n, = 247

The number of regressors in full model, k, = 5; therefore $(n - (k + 1)) = 247 - 6 = 241$

The number of additional regressor variables, p , = 3

$$\text{Test Statistic Value: } f = \frac{(SSE1 - SSE2) / p}{MSE2} = \frac{(6636.40 - 6581.36) / 3}{27.3086} = 0.672$$

Rejection region for a level $\alpha = 0.05$ test = $f \geq F_{\alpha, k, n-(k+1)} = f \geq F_{0.05, 2, 241} = 2.60$ (Table A-4, Neter and Wasserman, 1974:812)

Since $0.672 < 2.60$, the decision is to accept the null hypothesis and conclude additional variables do not contribute to the model. Therefore, the reduced model will be used to calculate a residual.

Based on p-values for the reduced model, the CUMFH and NSORT predictor variables contribute to the reduced model (the p-values are approximately equal to or less than $\alpha = 0.05$). The reduced model takes the following form: $Y = -0.01133x_1 + 0.10203x_2$. Y represents the expected number of demands or maintenance actions. x_1 represents cumulative flying hours and x_2 represents the number of sorties.

Residual computation: The cumulative flying hour total from the 1994 validation data set is 60,254.5 and the total number of sorties is 38,666. Substituting these values into the reduced model yields $Y = -0.01133(60,254.5) + 0.10203(38,666) = 3,262.41$ or approximately 3,262 expected demands or maintenance actions for 60,254.5 flying hours and 38,666 sorties. The actual number of 74 demands or maintenance actions from the 1994 data is 3,710. Therefore, the residual on the reduced multiple regression model is $3,710 - 3,262 = 448$.

Reduced and Full Model Multiple Regressions and Hypothesis Testing for Two Digit Work Unit Code 76:

Unweighted Least Squares Linear Regression of Demands (Reduced Model)

| Predictor Variables | Coefficient | P-Value |
|---------------------|-------------|---------|
| CONSTANT | 0.52078 | 0.3368 |
| CUMFH | 0.01797 | 0.0000 |
| NSORT | -7.182E-04 | 0.9076 |

R- Squared: 0.2522; Adjusted R-Squared: 0.2461; Sum of Squares Error (SSE1): 1841.95

Hypothesis Testing: Ho: $\beta_1 = \beta_2 = \dots = \beta_k = 0$

Ha: at least one $\beta_i \neq 0$ ($i = 1, \dots, k$)

$$R^2 = 0.2522$$

n = number of data points = 247

k = number of carriers = 2

$$\text{Test Statistic Value: } f = \frac{R^2 / k}{(1 - R^2) / [n - (k + 1)]} = \frac{0.2522 / 2}{(1 - 0.2522) / [247 - (2 + 1)]} = 41.15$$

Rejection region for a level $\alpha = 0.05$ test = $f \geq F_{\alpha, k, n - (k + 1)} = f \geq F_{0.05, 2, 244} = 3.00$ (Table A-4, Neter and Wasserman, 1974:812)

Since $41.15 \gg 3.00$, the null hypothesis is rejected and the conclusion is demands/maintenance actions is linearly related to at least one of the predictor variables.

Now, the test to determine if additional carriers can improve the reduced model:

Unweighted Least Squares Linear Regression of Demands (Full Model)

| Predictor Variables | Coefficient | P-Value |
|---------------------|-------------|---------|
| CONSTANT | 0.20846 | 0.8323 |
| CUMFH | 0.01202 | 0.4202 |
| NSORT | 0.01442 | 0.6141 |
| CUMFH2 | -3.514E-05 | 0.3664 |
| NSORT2 | -2.095E-04 | 0.3207 |
| CUMFHNSORT | 1.785E-04 | 0.3149 |

R-Squared: 0.2556; Adjusted R-Squared: 0.2402

Sum of Squares Error (SSE2): 1833.54; Residual Mean Square (MSE2): 7.60804

Hypothesis Testing:

Ho: model is $Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \epsilon$ (reduced model)

Ha: model is $Y = \beta_0 + \beta_1 x_1 + \beta_2 x_2 + \beta_3 x_1^2 + \beta_4 x_2^2 + \beta_5 x_1 x_2 + \epsilon$ (full model)

The number of points, n, = 247

The number of regressors in full model, k, = 5; therefore $(n - (k + 1)) = 247 - 6 = 241$

The number of additional regressor variables, $p, = 3$

$$\text{Test Statistic Value: } f = \frac{(SSE1 - SSE2) / p}{MSE2} = \frac{(1841.95 - 1833.54) / 3}{7.60804} = 0.36$$

Rejection region for a level $\alpha = 0.05$ test = $f \geq F_{\alpha, k, n-(k+1)} = f \geq F_{0.05, 2, 241} = 2.60$ (Table A-4, Neter and Wasserman, 1974:812)

Since $0.36 < 2.60$, the decision is to accept the null hypothesis and conclude additional variables do not contribute to the model. Therefore, the reduced model will be used to calculate a residual.

Based on p-values for the reduced model, only the CUMFH predictor variable contributes to the reduced model (the p-value is less than $\alpha = 0.05$). The reduced model takes the following form: $Y = 0.01797x_1$. Y represents the expected number of demands or maintenance actions and x_1 represents cumulative flying hours.

Residual computation: The cumulative flying hour total from the 1994 validation data set is 60,254.5 and the total number of sorties is 38,666. Substituting these values into the reduced model yields $Y = 0.01797(60,254.5) = 1,082.77$ or approximately 1,083 expected demands or maintenance actions for 60,254.5 flying hours and 38,666 sorties. The actual number of 76 demands or maintenance actions from the 1994 data is 1,842. Therefore, the residual on the reduced multiple regression model is $1,842 - 1,083 = 759$.

Appendix E: Poisson Regression Results and Residual Calculations

Results for Two-Digit Work Unit Code Level 11:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.33647 | 0.0282 |
| CUMFH | 0.00146 | 0.0283 |

Deviance: 412.07; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.98256 | 0.0000 |
| NSORT | 0.00691 | 0.0000 |

Deviance: 396.26; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.99760 | 0.0000 |
| CUMFH | -0.00219 | 0.0766 |
| NSORT | 0.01027 | 0.0000 |

Deviance: 392.74; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, both CONSTANT and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is: $\hat{\mu} = -0.99760 + 0.01027(38,666) = 396.12$ or approximately 396 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 11 demands/maintenance actions is 396. The actual number of 11 demands/maintenance actions from the 1994 data set is 794. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $794 - 396 = 398$.

Poisson Regression Results for Two-Digit Work Unit Code Level 12:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.98685 | 0.0000 |
| CUMFH | 0.00191 | 0.0250 |

Deviance: 340.55; P-Value: 0.0001

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -2.256 | 0.0000 |
| NSORT | 0.01180 | 0.0000 |

Deviance: 312.95; P-Value: 0.0022

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -2.28723 | 0.0000 |
| CUMFH | -0.00546 | 0.0066 |
| NSORT | 0.01998 | 0.0000 |

Deviance: 302.94; P-Value: 0.0061

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT, CUMFH, and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is: $\hat{\mu} = -2.28723 - 0.00546(60,254.5) + 0.01998(38,666) = 441.27$ or approximately 441 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 12 demands/maintenance actions is 441. The actual number of 12 demands/maintenance actions from the 1994 data set is 318. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $318 - 441 = -123$.

Poisson Regression Results for Two-Digit Work Unit Code Level 13:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 1.62039 | 0.0000 |
| CUMFH | 0.00111 | 0.0000 |

Deviance: 658.90; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.88507 | 0.0000 |
| NSORT | 0.00702 | 0.0000 |

Deviance: 537.38; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.86609 | 0.0000 |
| CUMFH | -0.00363 | 0.0000 |
| NSORT | 0.01250 | 0.0000 |

Deviance: 482.11; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT, CUMFH, and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is: $\hat{\mu} = 0.86609 - 0.00363(60,254.5) + 0.01250(38,666) = 265.47$ or approximately 265 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 13 demands/maintenance actions is 265. The actual number of 13 demands/maintenance actions from the 1994 data set is 2,092. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $2,092 - 265 = 1,827$.

Poisson Regression Results for Two-Digit Work Unit Code Level 14:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.13736 | 0.3081 |
| CUMFH | 5.189E-04 | 0.3966 |

Deviance: 493.65; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.55521 | 0.0049 |
| NSORT | 0.00588 | 0.0000 |

Deviance: 474.59; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.57246 | 0.0040 |
| CUMFH | -0.00481 | 0.0003 |
| NSORT | 0.01303 | 0.0000 |

Deviance: 457.61; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT, CUMFH, and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is: $\hat{\mu} = -0.57246 - 0.00481(60,254.5) + 0.01303(38,666) = 213.42$ or approximately 213 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 14 demands/maintenance actions is 213. The actual number of 14 demands/maintenance actions from the 1994 data set is 1,121. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $1,121 - 213 = 908$.

Poisson Regression Results for Two-Digit Work Unit Code Level 23:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 1.00816 | 0.0000 |
| CUMFH | 0.00202 | 0.0000 |

Deviance: 557.00; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.27268 | 0.0176 |
| NSORT | 0.00835 | 0.0000 |

Deviance: 470.06; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.25923 | 0.0244 |
| CUMFH | -0.00158 | 0.0047 |
| NSORT | 0.01080 | 0.0000 |

Deviance: 461.35; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT, CUMFH, and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is: $\hat{\mu} = 0.25923 - 0.00158(60,254.5) + 0.01080(38,666) = 322.65$ or approximately 323 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 23 demands/maintenance actions is 323. The actual number of 23 demands/maintenance actions from the 1994 data set is 2,069. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $2,069 - 323 = 1,746$.

Poisson Regression Results for Two-Digit Work Unit Code Level 24:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.38121 | 0.0005 |
| CUMFH | 0.00131 | 0.0059 |

Deviance: 572.59; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.25395 | 0.1210 |
| NSORT | 0.00661 | 0.0000 |

Deviance: 542.59; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.26952 | 0.1013 |
| CUMFH | -0.00243 | 0.0068 |
| NSORT | 0.01033 | 0.0000 |

Deviance: 534.23; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CUMFH and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is:
 $\hat{\mu} = -0.00243(60,254.5) + 0.01033(38,666) = 253.00$ or approximately 253 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 24 demands/maintenance actions is 253. The actual number of 24 demands/maintenance actions from the 1994 data set is 1,114. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $1,114 - 253 = 861$.

Poisson Regression Results for Two-Digit Work Unit Code Level 41:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.30292 | 0.0243 |
| CUMFH | 0.00243 | 0.0000 |

Deviance: 450.63; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.81418 | 0.0001 |
| NSORT | 0.00740 | 0.0000 |

Deviance: 438.33; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.81099 | 0.0001 |
| CUMFH | 2.867E-04 | 0.7492 |
| NSORT | 0.00694 | 0.0005 |

Deviance: 438.23; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is: $\hat{\mu} = -0.81099 + 0.00694(38,666) = 267.53$ or approximately 268 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 41 demands/maintenance actions is 268. The actual number of 41 demands/maintenance actions from the 1994 data set is 667. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $667 - 268 = 399$.

Poisson Regression Results for Two-Digit Work Unit Code Level 42:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.22441 | 0.0966 |
| CUMFH | 0.00206 | 0.0003 |

Deviance: 507.57; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.87985 | 0.0000 |
| NSORT | 0.00786 | 0.0000 |

Deviance: 486.57; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.88941 | 0.0000 |
| CUMFH | -0.00106 | 0.2874 |
| NSORT | 0.00951 | 0.0000 |

Deviance: 485.37; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is:
 $\hat{\mu} = -0.88941 + 0.00951(38,666) = 366.82$ or approximately 367 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 42 demands/maintenance actions is 367. The actual number of 42 demands/maintenance actions from the 1994 data set is 766. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $766 - 367 = 399$.

Poisson Regression Results for Two-Digit Work Unit Code Level 44:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.37666 | 0.0005 |
| CUMFH | 0.00138 | 0.0037 |

Deviance: 464.08; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.40250 | 0.0157 |
| NSORT | 0.00771 | 0.0000 |

Deviance: 421.76; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.42207 | 0.0117 |
| CUMFH | -0.00331 | 0.0005 |
| NSORT | 0.01273 | 0.0000 |

Deviance: 407.13; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT, CUMFH, and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is: $\hat{\mu} = -0.42207 - 0.00331(60,254.5) + 0.01273(38,666) = 292.35$ or approximately 292 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 44 demands/maintenance actions is 292. The actual number of 44 demands/maintenance actions from the 1994 data set is 665. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $665 - 292 = 373$.

Poisson Regression Results for Two-Digit Work Unit Code Level 45:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.76796 | 0.0000 |
| CUMFH | 0.00235 | 0.0010 |

Deviance: 350.79; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -1.60279 | 0.0000 |
| NSORT | 0.00952 | 0.0000 |

Deviance: 331.30; P-Value: 0.0002

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -1.61624 | 0.0000 |
| CUMFH | -0.00143 | 0.2622 |
| NSORT | 0.01175 | 0.0000 |

Deviance: 329.95; P-Value: 0.0002

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is:
 $\hat{\mu} = -1.61624 + 0.01175(38,666) = 452.71$ or approximately 453 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 45 demands/maintenance actions is 453. The actual number of 45 demands/maintenance actions from the 1994 data set is 965. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $965 - 453 = 512$.

Poisson Regression Results for Two-Digit Work Unit Code Level 46:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.73428 | 0.0000 |
| CUMFH | 0.00407 | 0.0000 |

Deviance: 360.49; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -1.32260 | 0.0000 |
| NSORT | 0.01049 | 0.0000 |

Deviance: 359.30; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -1.27776 | 0.0000 |
| CUMFH | 0.00245 | 0.0012 |
| NSORT | 0.00638 | 0.0013 |

Deviance: 350.02; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT, CUMFH, and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is: $\hat{\mu} = -1.27776 + 0.00245(60,254.5) + 0.00638(38,666) = 393.03$ or approximately 393 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 46 demands/maintenance actions is 393. The actual number of 46 demands/maintenance actions from the 1994 data set is 808. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $808 - 393 = 415$.

Poisson Regression Results for Two-Digit Work Unit Code Level 47:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -1.15019 | 0.0000 |
| CUMFH | 0.00257 | 0.0020 |

Deviance: 314.66; P-Value: 0.0018

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -2.33310 | 0.0000 |
| NSORT | 0.01217 | 0.0000 |

Deviance: 289.9; P-Value: 0.0258

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -2.35765 | 0.0000 |
| CUMFH | -0.00280 | 0.0887 |
| NSORT | 0.01649 | 0.0000 |

Deviance: 286.50; P-Value: 0.0320

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is:
 $\hat{\mu} = -2.35765 + 0.01649(38,666) = 635.24$ or approximately 635 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 47 demands/maintenance actions is 635. The actual number of 47 demands/maintenance actions from the 1994 data set is 280. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $280 - 635 = -355$.

Poisson Regression Results for Two-Digit Work Unit Code Level 49:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -2.05009 | 0.0000 |
| CUMFH | 0.00211 | 0.1333 |

Deviance: 196.69; P-Value: 0.9897

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -2.89330 | 0.0000 |
| NSORT | 0.00922 | 0.0086 |

Deviance: 191.46; P-Value: 0.9952

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -2.91000 | 0.0000 |
| CUMFH | -0.00199 | 0.4460 |
| NSORT | 0.01231 | 0.0198 |

Deviance: 190.81; P-Value: 0.9950

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is:
 $\hat{\mu} = -2.91000 + 0.01231(38,666) = 473.07$ or approximately 473 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 49 demands/maintenance actions is 473. The actual number of 49 demands/maintenance actions from the 1994 data set is 118. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $118 - 473 = -355$.

Poisson Regression Results for Two-Digit Work Unit Code Level 51:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.56724 | 0.0000 |
| CUMFH | 0.00260 | 0.0000 |

Deviance: 456.33; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.01529 | 0.9091 |
| NSORT | 0.00817 | 0.0000 |

Deviance: 417.93; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.01243 | 0.9261 |
| CUMFH | 2.477E-04 | 0.6616 |
| NSORT | 0.00777 | 0.0000 |

Deviance: 417.74; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, only NSORT contributes to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is:
 $\hat{\mu} = 0.00777(38,666) = 300.43$ or approximately 300 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 51 demands/maintenance actions is 300. The actual number of 51 demands/maintenance actions from the 1994 data set is 1,043. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $1,043 - 300 = 743$.

Poisson Regression Results for Two-Digit Work Unit Code Level 52:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.73840 | 0.0000 |
| CUMFH | 0.00209 | 0.0044 |

Deviance: 431.70; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -1.17689 | 0.0000 |
| NSORT | 0.00638 | 0.0003 |

Deviance: 425.82; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -1.17567 | 0.0000 |
| CUMFH | 1.069E-04 | 0.9281 |
| NSORT | 0.00621 | 0.0164 |

Deviance: 425.81; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is: $\hat{\mu} = -1.17567 + 0.00621(38,666) = 238.94$ or approximately 239 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 52 demands/maintenance actions is 239. The actual number of 52 demands/maintenance actions from the 1994 data set is 284. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $284 - 239 = 45$.

Poisson Regression Results for Two-Digit Work Unit Code Level 55:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.31020 | 0.0686 |
| CUMFH | 4.255E-04 | 0.5838 |

Deviance: 513.65; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.97823 | 0.0001 |
| NSORT | 0.00557 | 0.0010 |

Deviance: 502.07; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.99483 | 0.0001 |
| CUMFH | -0.00483 | 0.0042 |
| NSORT | 0.01274 | 0.0000 |

Deviance: 492.14; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT, CUMFH, and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is: $\hat{\mu} = -0.99483 - 0.00483(60,254.5) + 0.01274(38,666) = 200.58$ or approximately 201 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 55 demands/maintenance actions is 201. The actual number of 55 demands/maintenance actions from the 1994 data set is 502. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $502 - 201 = 301$.

Poisson Regression Results for Two-Digit Work Unit Code Level 57:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -1.11911 | 0.0003 |
| CUMFH | -0.00147 | 0.3369 |

Deviance: 223.38; P-Value: 0.8356

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -2.04007 | 0.0000 |
| NSORT | 0.00478 | 0.1133 |

Deviance: 221.77; P-Value: 0.8541

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -2.01050 | 0.0000 |
| CUMFH | -0.01882 | 0.0008 |
| NSORT | 0.03074 | 0.0001 |

Deviance: 202.70; P-Value: 0.9748

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT, CUMFH, and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is: $\hat{\mu} = -2.01050 - 0.01882(60,254.5) + 0.03074(38,666) = 52.59$ or approximately 53 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 57 demands/maintenance actions is 53. The actual number of 57 demands/maintenance actions from the 1994 data set is 265. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $265 - 53 = 212$.

Poisson Regression Results for Two-Digit Work Unit Code Level 63:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.67209 | 0.0000 |
| CUMFH | 0.00212 | 0.0000 |

Deviance: 526.33; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.15134 | 0.2663 |
| NSORT | 0.00910 | 0.0000 |

Deviance: 449.66; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.16718 | 0.2211 |
| CUMFH | -0.00186 | 0.0050 |
| NSORT | 0.01199 | 0.0000 |

Deviance: 440.93; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CUMFH and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is:
 $\hat{\mu} = -0.00186(60,254.5) + 0.01199(38,666) = 351.53$ or approximately 352 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 63 demands/maintenance actions is 352. The actual number of 63 demands/maintenance actions from the 1994 data set is 1,057. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $1,057 - 352 = 705$.

Poisson Regression Results for Two-Digit Work Unit Code Level 65:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.37084 | 0.0001 |
| CUMFH | 0.00278 | 0.0000 |

Deviance: 542.55; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.00384 | 0.9784 |
| NSORT | 0.00700 | 0.0000 |

Deviance: 535.90; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.01481 | 0.9167 |
| CUMFH | 0.00141 | 0.0126 |
| NSORT | 0.00470 | 0.0005 |

Deviance: 530.10; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CUMFH and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is:
 $\hat{\mu} = 0.00141(60,254.5) + 0.00470(38,666) = 266.69$ or approximately 267 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 65 demands/maintenance actions is 267. The actual number of 65 demands/maintenance actions from the 1994 data set is 771. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $771 - 267 = 504$.

Poisson Regression Results for Two-Digit Work Unit Code Level 71:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.38012 | 0.0001 |
| CUMFH | 0.00249 | 0.0000 |

Deviance: 546.29; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.21951 | 0.1400 |
| NSORT | 0.00812 | 0.0000 |

Deviance: 512.32; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.21961 | 0.1403 |
| CUMFH | -9.668E-06 | 0.9880 |
| NSORT | 0.00813 | 0.0000 |

Deviance: 512.32; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, only NSORT contributes to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is:
 $\hat{\mu} = 0.00813(38,666) = 314.35$ or approximately 314 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 71 demands/maintenance actions is 314. The actual number of 71 demands/maintenance actions from the 1994 data set is 1,049. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $1,049 - 314 = 735$.

Poisson Regression Results for Two-Digit Work Unit Code Level 74:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 1.90619 | 0.0000 |
| CUMFH | 0.00226 | 0.0000 |

Deviance: 840.81; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 1.19240 | 0.0000 |
| NSORT | 0.00855 | 0.0000 |

Deviance: 628.75; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 1.18292 | 0.0000 |
| CUMFH | -9.924E-04 | 0.0027 |
| NSORT | 0.001011 | 0.0000 |

Deviance: 619.26; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT, CUMFH, and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is: $\hat{\mu} = 1.18292 - 0.0009924(60,254.5) + 0.001011(38,666) = 332.29$ or approximately 332 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 74 demands/maintenance actions is 332. The actual number of 74 demands/maintenance actions from the 1994 data set is 3,710. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $3,710 - 332 = 3,378$.

Poisson Regression Results for Two-Digit Work Unit Code Level 75:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.54251 | 0.0000 |
| CUMFH | 4.454E-04 | 0.3783 |

Deviance: 566.99; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.11038 | 0.4940 |
| NSORT | 0.00549 | 0.0000 |

Deviance: 542.15; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.12693 | 0.4345 |
| CUMFH | -0.00462 | 0.0000 |
| NSORT | 0.01237 | 0.0000 |

Deviance: 518.77; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CUMFH, and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is:
 $\hat{\mu} = -0.00462(60,254.5) + 0.01237(38,666) = 199.92$ or approximately 200 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 75 demands/maintenance actions is 200. The actual number of 75 demands/maintenance actions from the 1994 data set is 725. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $725 - 200 = 525$.

Poisson Regression Results for Two-Digit Work Unit Code Level 76:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.60785 | 0.0000 |
| CUMFH | 0.00360 | 0.0000 |

Deviance: 489.07; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.35741 | 0.0020 |
| NSORT | 0.00742 | 0.0000 |

Deviance: 528.86; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | 0.40819 | 0.0004 |
| CUMFH | 0.00295 | 0.0000 |
| NSORT | 0.00247 | 0.0203 |

Deviance: 483.65; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT, CUMFH, and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is: $\hat{\mu} = 0.40819 + 0.00295(60,254.5) + 0.00247(38,666) = 273.66$ or approximately 274 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 76 demands/maintenance actions is 274. The actual number of 76 demands/maintenance actions from the 1994 data set is 1,842. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $1,842 - 274 = 1,568$.

Poisson Regression Results for Two-Digit Work Unit Code Level 97:

CUMFH Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.34314 | 0.06011 |
| CUMFH | -9.601E-05 | 0.9106 |

Deviance: 482.82; P-Value: 0.0000

NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.91677 | 0.0004 |
| NSORT | 0.00415 | 0.0205 |

Deviance: 477.31; P-Value: 0.0000

CUMFH and NSORT Model Results:

| Predictor Variables | Coefficient | p-value |
|---------------------|-------------|---------|
| CONSTANT | -0.93006 | 0.0003 |
| CUMFH | -0.00552 | 0.0056 |
| NSORT | 0.01186 | 0.0002 |

Deviance: 467.16; P-Value: 0.0000

Estimation of Demands/Maintenance Actions: The general form of the Poisson regression model used in this study is: $\hat{\mu} = \hat{\beta}_0 + \hat{\beta}_1 x_1 + \hat{\beta}_2 x_2$. Based on the deviance values, the CUMFH and NSORT model has the smallest deviance. Predictor variable p-values will be compared to an $\alpha = 0.05$ to determine which variables contribute to the model. In the CUMFH and NSORT model, CONSTANT, CUMFH, and NSORT contribute to the model. This model will be used to estimate the number of demands/maintenance actions from the 1994 validation data set. The 1994 validation data set is based on 60,254.5 flying hours and 38,666 sorties. The expected number of maintenance actions is: $\hat{\mu} = -0.93006 - 0.00552(60,254.5) + 0.01186(38,666) = 143.12$ or approximately 143 expected demands/maintenance actions.

Calculation of the Poisson Regression Residual: The expected number of 97 demands/maintenance actions is 143. The actual number of 97 demands/maintenance actions from the 1994 data set is 476. The residual is calculated by taking the actual value minus the expected value. The value of the residual is $476 - 143 = 333$.

Appendix F: Data Tables and Poisson Process Calculations

Table F-1: 1993 Data Used to Determine Poisson Process Lambdas

| Two-Digit Work Unit Code | Number of Maint. Actions | Cum. Flying Hours | Cum. # of Sorties | Lambda, λ, (# of Maint. Actions/ Cum Flying Hours) | Lambda, λ, (# of Maint. Actions/ Cum # of Sorties) |
|---|---|----------------------------------|------------------------------|--|--|
| 11 | 237 | 48,337.5 | 32,057 | 0.0049 | 0.0074 |
| 12 | 136 | 48,337.5 | 32,057 | 0.0028 | 0.0042 |
| 13 | 1,560 | 48,337.5 | 32,057 | 0.0323 | 0.0487 |
| 14 | 314 | 48,337.5 | 32,057 | 0.0065 | 0.0098 |
| 23 | 1,023 | 48,337.5 | 32,057 | 0.0212 | 0.0319 |
| 24 | 471 | 48,337.5 | 32,057 | 0.0097 | 0.0147 |
| 41 | 301 | 48,337.5 | 32,057 | 0.0062 | 0.0094 |
| 42 | 301 | 48,337.5 | 32,057 | 0.0062 | 0.0094 |
| 44 | 475 | 48,337.5 | 32,057 | 0.0098 | 0.0148 |
| 45 | 186 | 48,337.5 | 32,057 | 0.0038 | 0.0058 |
| 46 | 284 | 48,337.5 | 32,057 | 0.0058 | 0.0088 |
| 47 | 133 | 48,337.5 | 32,057 | 0.0028 | 0.0041 |
| 49 | 49 | 48,337.5 | 32,057 | 0.0010 | 0.0015 |
| 51 | 747 | 48,337.5 | 32,057 | 0.0155 | 0.0233 |
| 52 | 181 | 48,337.5 | 32,057 | 0.0037 | 0.0056 |
| 55 | 197 | 48,337.5 | 32,057 | 0.0041 | 0.0061 |
| 57 | 61 | 48,337.5 | 32,057 | 0.0013 | 0.0019 |
| 63 | 747 | 48,337.5 | 32,057 | 0.0155 | 0.0233 |
| 65 | 639 | 48,337.5 | 32,057 | 0.0132 | 0.0199 |
| 71 | 605 | 48,337.5 | 32,057 | 0.0125 | 0.0189 |
| 74 | 2,643 | 48,337.5 | 32,057 | 0.0547 | 0.0824 |
| 75 | 464 | 48,337.5 | 32,057 | 0.0096 | 0.0145 |
| 76 | 974 | 48,337.5 | 32,057 | 0.0201 | 0.0304 |
| 97 | 172 | 48,337.5 | 32,057 | 0.0036 | 0.0054 |

Note: Data used to comprise the above table was for F-15Cs covering the period of May to December 1993. Database contained 247 aircraft which flew a total of 32,057 sorties and accumulated 48,337.5 flying hours over the eight month period.

How to interpret table: Consider two digit work unit code, 11. The 247 aircraft in the database had 237 maintenance actions at the 11 work unit code level, while flying 48,337.5 hours or 32,057 sorties.

Table F-2: 1994 Validation Data

| Two-Digit Work Unit Code | Number of Actual Maintenance Actions | Cum. Flying Hours | Cum. # of Sorties |
|--------------------------|--------------------------------------|-------------------|-------------------|
| 11 | 794 | 60,254.5 | 38,666 |
| 12 | 318 | 60,254.5 | 38,666 |
| 13 | 2,092 | 60,254.5 | 38,666 |
| 14 | 1,121 | 60,254.5 | 38,666 |
| 23 | 2,069 | 60,254.5 | 38,666 |
| 24 | 1,114 | 60,254.5 | 38,666 |
| 41 | 667 | 60,254.5 | 38,666 |
| 42 | 766 | 60,254.5 | 38,666 |
| 44 | 665 | 60,254.5 | 38,666 |
| 45 | 965 | 60,254.5 | 38,666 |
| 46 | 808 | 60,254.5 | 38,666 |
| 47 | 280 | 60,254.5 | 38,666 |
| 49 | 118 | 60,254.5 | 38,666 |
| 51 | 1,043 | 60,254.5 | 38,666 |
| 52 | 284 | 60,254.5 | 38,666 |
| 55 | 502 | 60,254.5 | 38,666 |
| 57 | 265 | 60,254.5 | 38,666 |
| 63 | 1,057 | 60,254.5 | 38,666 |
| 65 | 771 | 60,254.5 | 38,666 |
| 71 | 1,049 | 60,254.5 | 38,666 |
| 74 | 3,710 | 60,254.5 | 38,666 |
| 75 | 725 | 60,254.5 | 38,666 |
| 76 | 1,842 | 60,254.5 | 38,666 |
| 97 | 476 | 60,254.5 | 38,666 |

Note: Data used to comprise the above table was for F-15Cs covering the period from of February to June 1994. Database contained 340 aircraft which flew a total of 38,666 sorties and accumulated 60,254.5 flying hours over the five month period.

How to interpret table: Consider work unit code 11. The 340 aircraft in the database had 794 maintenance actions at the 11 work unit code level, while flying 60,254.5 flying hours or 38,666 sorties.

Poisson Process Calculations for the 11 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0049) based on flying hours [$237/48,337.5 = 0.0049$].

Z is Poisson (0.0074) based on sorties [$237/32,057 = 0.0074$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0049 * 60,254.5$), the expected value of X is $(0.0049)(60,254.5) = 295.247$. Also, the variance of X is 295.247, while the standard deviation, σ , is $\sqrt{295.247} = 17.18$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (260, 329).

The probability that $(X \leq 260 \text{ or } X \geq 329) = 0.01999 + (1 - 0.97536) = 0.04463$.

The actual number of 11 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 794. The expected number of 11 maintenance actions is approximately 295. Thus, the residual on the Poisson process based on flying hours is $794 - 295 = 499$.

Sortie Based:

If X is approximately Poisson ($0.0074 * 38,666$), the expected value of X is $(0.0074)(38,666) = 286.128$. Also, the variance of X is 286.128, while the standard deviation, σ , is $\sqrt{286.128} = 16.91$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (252, 320).

The probability that $(X \leq 252 \text{ or } X \geq 320) = 0.02175 + (1 - 0.9774) = 0.04435$.

The actual number of 11 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 794. The expected number of 11 maintenance actions is approximately 286. Thus, the residual on the Poisson process based on sorties is $794 - 286 = 508$.

Poisson Process Calculations for the 12 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0028) based on flying hours [$136/48,337.5 = 0.0028$].

Z is Poisson (0.0042) based on sorties [$136/32,057 = 0.0042$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0028 * 60,254.5$), the expected value of X is $(0.0028)(60,254.5) = 168.713$. Also, the variance of X is 168.713, while the standard deviation, σ , is $\sqrt{168.713} = 12.99$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (143, 195).

The probability that $(X \leq 143 \text{ or } X \geq 195) = 0.02388 + (1 - 0.97849) = 0.04539$.

The actual number of 12 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 318. The expected number of 11 maintenance actions is approximately 169. Thus, the residual on the Poisson process based on flying hours is $318 - 169 = 149$.

Sortie Based:

If X is approximately Poisson ($0.0042 * 38,666$), the expected value of X is $(0.0042)(38,666) = 162.397$. Also, the variance of X is 162.397, while the standard deviation, σ , is $\sqrt{162.397} = 12.74$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (137, 188).

The probability that $(X \leq 137 \text{ or } X \geq 188) = 0.02310 + (1 - 0.97776) = 0.04534$.

The actual number of 12 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 318. The expected number of 11 maintenance actions is approximately 162. Thus, the residual on the Poisson process based on sorties is $318 - 162 = 156$.

Poisson Process Calculations for the 13 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0323) based on flying hours [$1,560/48,337.5 = 0.0323$].

Z is Poisson (0.0487) based on sorties [$1,560/32,057 = 0.0487$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0323 * 60,254.5$), the expected value of X is $(0.0323)(60,254.5) = 1,946.220$. Also, the variance of X is 1,946.220, while the standard deviation, σ , is $\sqrt{1,946.220} = 44.12$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (1,858, 2,034).

The probability that $(X \leq 1,858 \text{ or } X \geq 2,034) = 0.02276 + (1 - 0.9767) = 0.04606$.

The actual number of 13 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 2,092. The expected number of 13 maintenance actions is approximately 1,946. Thus, the residual on the Poisson process based on flying hours is $2,092 - 1,946 = 146$.

Sortie Based:

If X is approximately Poisson ($0.0487 * 38,666$), the expected value of X is $(0.0487)(38,666) = 1,883.034$. Also, the variance of X is 1,883.034, while the standard deviation, σ , is $\sqrt{1,883.034} = 43.39$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (1,796, 1,970).

The probability that $(X \leq 1,796 \text{ or } X \geq 1,970) = 0.02244 + (1 - 0.97747) = 0.04497$.

The actual number of 13 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 2,092. The expected number of 13 maintenance actions is approximately 1,883. Thus, the residual on the Poisson process based on sorties is $2,092 - 1,883 = 209$.

Poisson Process Calculations for the 14 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0065) based on flying hours $[314/48,337.5 = 0.0065]$.

Z is Poisson (0.0098) based on sorties $[314/32,057 = 0.0098]$.

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0065 * 60,254.5$), the expected value of X is $(0.0065)(60,254.5) = 391.654$. Also, the variance of X is 391.654, while the standard deviation, σ , is $\sqrt{391.654} = 19.79$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (352,431).

The probability that $(X \leq 352 \text{ or } X \geq 431) = 0.02252 + (1 - 0.97664) = 0.04588$.

The actual number of 14 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 1,121. The expected number of 14 maintenance actions is approximately 392. Thus, the residual on the Poisson process based on flying hours is $1,121 - 392 = 729$.

Sortie Based:

If X is approximately Poisson ($0.0098 * 38,666$), the expected value of X is $(0.0098)(38,666) = 378.927$. Also, the variance of X is 378.927, while the standard deviation, σ , is $\sqrt{378.927} = 19.47$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (340,418).

The probability that $(X \leq 340 \text{ or } X \geq 418) = 0.02274 + (1 - 0.97764) = 0.0451$.

The actual number of 14 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 1,121. The expected number of 14 maintenance actions is approximately 379. Thus, the residual on the Poisson process based on sorties is $1,121 - 379 = 742$.

Poisson Process Calculations for the 23 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0212) based on flying hours [$1,023/48,337.5 = 0.0212$].

Z is Poisson (0.0319) based on sorties [$1,023/32,057 = 0.0319$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0212 * 60,254.5$), the expected value of X is $(0.0212)(60,254.5) = 1,277.395$. Also, the variance of X is 1,277.395, while the standard deviation, σ , is $\sqrt{1,277.395} = 35.74$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (1,206, 1,349).

The probability that $(X \leq 1,206 \text{ or } X \geq 1,349) = 0.02288 + (1 - 0.97744) = 0.04544$.

The actual number of 23 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 2,069. The expected number of 23 maintenance actions is approximately 1,277. Thus, the residual on the Poisson process based on flying hours is $2,069 - 1,277 = 792$.

Sortie Based:

If X is approximately Poisson ($0.0319 * 38,666$), the expected value of X is $(0.0319)(38,666) = 1,233.445$. Also, the variance of X is 1,233.445, while the standard deviation, σ , is $\sqrt{1,233.445} = 35.12$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (1,163, 1,304).

The probability that $(X \leq 1,163 \text{ or } X \geq 1,304) = 0.02243 + (1 - 0.97773) = 0.0447$.

The actual number of 23 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 2,069. The expected number of 23 maintenance actions is approximately 1,233. Thus, the residual on the Poisson process based on sorties is $2,069 - 1,233 = 836$.

Poisson Process Calculations for the 24 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties.
Thus, Z is Poisson (0.0097) based on flying hours $[471/48,337.5 = 0.0097]$.

Z is Poisson (0.0147) based on sorties $[471/32,057 = 0.0147]$.

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0097 * 60,254.5$), the expected value of X is $(0.0097)(60,254.5) = 584.469$. Also, the variance of X is 584.469, while the standard deviation, σ , is $\sqrt{584.469} = 24.18$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (536,633).

The probability that $(X \leq 536 \text{ or } X \geq 633) = 0.02247 + (1 - 0.97765) = 0.04482$.

The actual number of 24 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 1,114. The expected number of 24 maintenance actions is approximately 584. Thus, the residual on the Poisson process based on flying hours is $1,114 - 584 = 530$.

Sortie Based:

If X is approximately Poisson ($0.0147 * 38,666$), the expected value of X is $(0.0147)(38,666) = 568.390$. Also, the variance of X is 568.390, while the standard deviation, σ , is $\sqrt{568.390} = 23.84$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (521,616).

The probability that $(X \leq 521 \text{ or } X \geq 616) = 0.02342 + (1 - 0.97710) = 0.04632$.

The actual number of 24 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 1,114. The expected number of 24 maintenance actions is approximately 568. Thus, the residual on the Poisson process based on sorties is $1,114 - 568 = 546$.

Poisson Process Calculations for the 41 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0062) based on flying hours [$301/48,337.5 = 0.0062$].

Z is Poisson (0.0094) based on sorties [$301/32,057 = 0.0094$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0062 * 60,254.5$), the expected value of X is $(0.0062)(60,254.5) = 373.578$. Also, the variance of X is 373.578, while the standard deviation, σ , is $\sqrt{373.578} = 19.33$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (335,412).

The probability that $(X \leq 335 \text{ or } X \geq 412) = 0.02295 + (1 - 0.97663) = 0.04632$.

The actual number of 41 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 667. The expected number of 41 maintenance actions is approximately 374. Thus, the residual on the Poisson process based on flying hours is $667 - 374 = 293$.

Sortie Based:

If X is approximately Poisson ($0.0094 * 38,666$), the expected value of X is $(0.0094)(38,666) = 363.460$. Also, the variance of X is 363.460, while the standard deviation, σ , is $\sqrt{363.460} = 19.06$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (325,402).

The probability that $(X \leq 325 \text{ or } X \geq 402) = 0.02177 + (1 - 0.97837) = 0.0434$.

The actual number of 41 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 667. The expected number of 41 maintenance actions is approximately 363. Thus, the residual on the Poisson process based on sorties is $667 - 363 = 304$.

Poisson Process Calculations for the 42 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0062) based on flying hours [$301/48,337.5 = 0.0062$].

Z is Poisson (0.0094) based on sorties [$301/32,057 = 0.0094$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0062 * 60,254.5$), the expected value of X is $(0.0062)(60,254.5) = 373.578$. Also, the variance of X is 373.578, while the standard deviation, σ , is $\sqrt{373.578} = 19.33$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (335,412).

The probability that $(X \leq 335 \text{ or } X \geq 412) = 0.02295 + (1 - 0.97663) = 0.04632$.

The actual number of 42 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 766. The expected number of 42 maintenance actions is approximately 374. Thus, the residual on the Poisson process based on flying hours is $766 - 374 = 392$.

Sortie Based:

If X is approximately Poisson ($0.0094 * 38,666$), the expected value of X is $(0.0094)(38,666) = 363.460$. Also, the variance of X is 363.460, while the standard deviation, σ , is $\sqrt{363.460} = 19.06$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (325,402).

The probability that $(X \leq 325 \text{ or } X \geq 402) = 0.02177 + (1 - 0.97837) = 0.0434$.

The actual number of 42 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 766. The expected number of 42 maintenance actions is approximately 363. Thus, the residual on the Poisson process based on sorties is $766 - 363 = 403$.

Poisson Process Calculations for the 44 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0098) based on flying hours [$475/48,337.5 = 0.0098$].

Z is Poisson (0.0148) based on sorties [$475/32,057 = 0.0148$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0098 * 60,254.5$), the expected value of X is $(0.0098)(60,254.5) = 590.494$. Also, the variance of X is 590.494, while the standard deviation, σ , is $\sqrt{590.494} = 24.30$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (542,639).

The probability that $(X \leq 542 \text{ or } X \geq 639) = 0.02298 + (1 - 0.97706) = 0.04592$.

The actual number of 44 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 665. The expected number of 44 maintenance actions is approximately 590. Thus, the residual on the Poisson process based on flying hours is $665 - 590 = 75$.

Sortie Based:

If X is approximately Poisson ($0.0148 * 38,666$), the expected value of X is $(0.0148)(38,666) = 572.257$. Also, the variance of X is 572.257, while the standard deviation, σ , is $\sqrt{572.257} = 23.92$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (524,620).

The probability that $(X \leq 524 \text{ or } X \geq 620) = 0.02179 + (1 - 0.97704) = 0.04475$.

The actual number of 44 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 665. The expected number of 44 maintenance actions is approximately 572. Thus, the residual on the Poisson process based on sorties is $665 - 572 = 93$.

Poisson Process Calculations for the 45 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties.
Thus, Z is Poisson (0.0038) based on flying hours $[186/48,337.5 = 0.0038]$.

Z is Poisson (0.0058) based on sorties $[186/32,057 = 0.0058]$.

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0038 * 60,254.5$), the expected value of X is $(0.0038)(60,254.5) = 228.967$. Also, the variance of X is 228.967, while the standard deviation, σ , is $\sqrt{228.967} = 15.13$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (199,259).

The probability that $(X \leq 199 \text{ or } X \geq 259) = 0.02384 + (1 - 0.97648) = 0.04736$.

The actual number of 45 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 965. The expected number of 45 maintenance actions is approximately 229. Thus, the residual on the Poisson process based on flying hours is $965 - 229 = 736$.

Sortie Based:

If X is approximately Poisson ($0.0058 * 38,666$), the expected value of X is $(0.0058)(38,666) = 224.263$. Also, the variance of X is 224.263, while the standard deviation, σ , is $\sqrt{224.263} = 14.98$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (194,254).

The probability that $(X \leq 194 \text{ or } X \geq 254) = 0.02156 + (1 - 0.97653) = 0.04503$.

The actual number of 45 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 965. The expected number of 45 maintenance actions is approximately 224. Thus, the residual on the Poisson process based on sorties is $965 - 224 = 741$.

Poisson Process Calculations for the 46 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0058) based on flying hours [$284/48,337.5 = 0.0058$].

Z is Poisson (0.0088) based on sorties [$284/32,057 = 0.0088$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0058 * 60,254.5$), the expected value of X is $(0.0058)(60,254.5) = 349.476$. Also, the variance of X is 349.476, while the standard deviation, σ , is $\sqrt{349.476} = 18.69$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (312,387).

The probability that $(X \leq 312 \text{ or } X \geq 387) = 0.02246 + (1 - 0.97765) = 0.04481$.

The actual number of 46 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 808. The expected number of 46 maintenance actions is approximately 349. Thus, the residual on the Poisson process based on flying hours is $808 - 349 = 459$.

Sortie Based:

If X is approximately Poisson ($0.0088 * 38,666$), the expected value of X is $(0.0088)(38,666) = 340.261$. Also, the variance of X is 340.261, while the standard deviation, σ , is $\sqrt{340.261} = 18.45$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (303,377).

The probability that $(X \leq 303 \text{ or } X \geq 377) = 0.02163 + (1 - 0.97684) = 0.04479$.

The actual number of 46 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 808. The expected number of 46 maintenance actions is approximately 340. Thus, the residual on the Poisson process based on sorties is $808 - 340 = 468$.

Poisson Process Calculations for the 47 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0028) based on flying hours [$133/48,337.5 = 0.0028$].

Z is Poisson (0.0041) based on sorties [$133/32,057 = 0.0041$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0028 * 60,254.5$), the expected value of X is $(0.0028)(60,254.5) = 168.713$. Also, the variance of X is 168.713, while the standard deviation, σ , is $\sqrt{168.713} = 12.99$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (143,195).

The probability that $(X \leq 143 \text{ or } X \geq 195) = 0.02388 + (1 - 0.97849) = 0.04539$.

The actual number of 47 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 280. The expected number of 47 maintenance actions is approximately 169. Thus, the residual on the Poisson process based on flying hours is $280 - 169 = 111$.

Sortie Based:

If X is approximately Poisson ($0.0041 * 38,666$), the expected value of X is $(0.0041)(38,666) = 158.531$. Also, the variance of X is 158.531, while the standard deviation, σ , is $\sqrt{158.531} = 12.59$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (133,184).

The probability that $(X \leq 133 \text{ or } X \geq 184) = 0.02116 + (1 - 0.97844) = 0.04272$.

The actual number of 47 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 280. The expected number of 47 maintenance actions is approximately 158. Thus, the residual on the Poisson process based on sorties is $280 - 158 = 122$.

Poisson Process Calculations for the 49 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0010) based on flying hours [$49/48,337.5 = 0.0010$].

Z is Poisson (0.0015) based on sorties [$49/32,057 = 0.0015$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0010 * 60,254.5$), the expected value of X is $(0.0010)(60,254.5) = 60.255$. Also, the variance of X is 60.255, while the standard deviation, σ , is $\sqrt{60.255} = 7.76$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (45,76).

The probability that $(X \leq 45 \text{ or } X \geq 76) = 0.0247 + (1 - 0.97873) = 0.04597$.

The actual number of 49 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 118. The expected number of 49 maintenance actions is approximately 60. Thus, the residual on the Poisson process based on flying hours is $118 - 60 = 58$.

Sortie Based:

If X is approximately Poisson ($0.0015 * 38,666$), the expected value of X is $(0.0015)(38,666) = 57.999$. Also, the variance of X is 57.999, while the standard deviation, σ , is $\sqrt{57.999} = 7.62$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (43,73).

The probability that $(X \leq 43 \text{ or } X \geq 73) = 0.02443 + (1 - 0.97582) = 0.04861$.

The actual number of 49 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 118. The expected number of 49 maintenance actions is approximately 58. Thus, the residual on the Poisson process based on sorties is $118 - 58 = 60$.

Poisson Process Calculations for the 51 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0155) based on flying hours [$747/48,337.5 = 0.0155$].

Z is Poisson (0.0233) based on sorties [$747/32,057 = 0.0233$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0155 * 60,254.5$), the expected value of X is $(0.0155)(60,254.5) = 933.945$. Also, the variance of X is 933.945, while the standard deviation, σ , is $\sqrt{933.945} = 30.56$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (873,995).

The probability that $(X \leq 873 \text{ or } X \geq 995) = 0.02306 + (1 - 0.97714) = 0.04592$.

The actual number of 51 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 1,043. The expected number of 51 maintenance actions is approximately 934. Thus, the residual on the Poisson process based on flying hours is $1,043 - 934 = 109$.

Sortie Based:

If X is approximately Poisson ($0.0233 * 38,666$), the expected value of X is $(0.0233)(38,666) = 900.918$. Also, the variance of X is 900.918, while the standard deviation, σ , is $\sqrt{900.918} = 30.02$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (841,961).

The probability that $(X \leq 841 \text{ or } X \geq 961) = 0.02295 + (1 - 0.97735) = 0.0456$.

The actual number of 51 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 1,043. The expected number of 51 maintenance actions is approximately 901. Thus, the residual on the Poisson process based on sorties is $1,043 - 901 = 142$.

Poisson Process Calculations for the 52 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0037) based on flying hours [$181/48,337.5 = 0.0037$].

Z is Poisson (0.0056) based on sorties [$181/32,057 = 0.0056$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0037 * 60,254.5$), the expected value of X is $(0.0037)(60,254.5) = 222.942$. Also, the variance of X is 222.942, while the standard deviation, σ , is $\sqrt{222.942} = 14.93$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (193,253).

The probability that $(X \leq 193 \text{ or } X \geq 253) = 0.02241 + (1 - 0.97795) = 0.04446$.

The actual number of 52 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 284. The expected number of 52 maintenance actions is approximately 223. Thus, the residual on the Poisson process based on flying hours is $284 - 223 = 61$.

Sortie Based:

If X is approximately Poisson ($0.0056 * 38,666$), the expected value of X is $(0.0056)(38,666) = 216.530$. Also, the variance of X is 216.530, while the standard deviation, σ , is $\sqrt{216.530} = 14.71$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (187,246).

The probability that $(X \leq 187 \text{ or } X \geq 246) = 0.02233 + (1 - 0.97742) = 0.04491$.

The actual number of 52 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 284. The expected number of 52 maintenance actions is approximately 217. Thus, the residual on the Poisson process based on sorties is $284 - 217 = 67$.

Poisson Process Calculations for the 55 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties.
Thus, Z is Poisson (0.0041) based on flying hours $[197/48,337.5 = 0.0041]$.

Z is Poisson (0.0061) based on sorties $[197/32,057 = 0.0061]$.

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0041 * 60,254.5$), the expected value of X is $(0.0041)(60,254.5) = 247.043$. Also, the variance of X is 247.043, while the standard deviation, σ , is $\sqrt{247.043} = 15.72$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (216,278).

The probability that $(X \leq 216 \text{ or } X \geq 278) = 0.02416 + (1 - 0.97566) = 0.0485$.

The actual number of 55 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 502. The expected number of 55 maintenance actions is approximately 247. Thus, the residual on the Poisson process based on flying hours is $502 - 247 = 255$.

Sortie Based:

If X is approximately Poisson ($0.0061 * 38,666$), the expected value of X is $(0.0061)(38,666) = 235.863$. Also, the variance of X is 235.863, while the standard deviation, σ , is $\sqrt{235.863} = 15.36$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (205,267).

The probability that $(X \leq 205 \text{ or } X \geq 267) = 0.02218 + (1 - 0.97866) = 0.04352$.

The actual number of 55 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 502. The expected number of 55 maintenance actions is approximately 236. Thus, the residual on the Poisson process based on sorties is $502 - 236 = 266$.

Poisson Process Calculations for the 57 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0013) based on flying hours [$61/48,337.5 = 0.0013$].

Z is Poisson (0.0019) based on sorties [$61/32,057 = 0.0019$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0013 * 60,254.5$), the expected value of X is $(0.0013)(60,254.5) = 78.331$. Also, the variance of X is 78.331, while the standard deviation, σ , is $\sqrt{78.331} = 8.85$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (61,96).

The probability that $(X \leq 61 \text{ or } X \geq 96) = 0.02517 + (1 - 0.97716) = 0.04801$.

The actual number of 57 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 265. The expected number of 57 maintenance actions is approximately 78. Thus, the residual on the Poisson process based on flying hours is $265 - 78 = 187$.

Sortie Based:

If X is approximately Poisson ($0.0019 * 38,666$), the expected value of X is $(0.0019)(38,666) = 73.465$. Also, the variance of X is 73.465, while the standard deviation, σ , is $\sqrt{73.465} = 8.57$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (56,91).

The probability that $(X \leq 56 \text{ or } X \geq 91) = 0.02051 + (1 - 0.97954) = 0.04097$.

The actual number of 57 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 265. The expected number of 57 maintenance actions is approximately 73. Thus, the residual on the Poisson process based on sorties is $265 - 73 = 192$.

Poisson Process Calculations for the 63 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0155) based on flying hours $[747/48,337.5 = 0.0155]$.

Z is Poisson (0.0233) based on sorties $[747/32,057 = 0.0233]$.

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0155 * 60,254.5$), the expected value of X is $(0.0155)(60,254.5) = 933.945$. Also, the variance of X is 933.945, while the standard deviation, σ , is $\sqrt{933.945} = 30.56$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (873,995).

The probability that $(X \leq 873 \text{ or } X \geq 995) = 0.02306 + (1 - 0.97714) = 0.04592$.

The actual number of 63 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 1,057. The expected number of 63 maintenance actions is approximately 934. Thus, the residual on the Poisson process based on flying hours is $1,057 - 934 = 123$.

Sortie Based:

If X is approximately Poisson ($0.0233 * 38,666$), the expected value of X is $(0.0233)(38,666) = 900.918$. Also, the variance of X is 900.918, while the standard deviation, σ , is $\sqrt{900.918} = 30.02$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (841,961).

The probability that $(X \leq 841 \text{ or } X \geq 961) = 0.02295 + (1 - 0.97735) = 0.0456$.

The actual number of 63 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 1,057. The expected number of 63 maintenance actions is approximately 901. Thus, the residual on the Poisson process based on sorties is $1,057 - 901 = 156$.

Poisson Process Calculations for the 65 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0132) based on flying hours $[639/48,337.5 = 0.0132]$.

Z is Poisson (0.0199) based on sorties $[639/32,057 = 0.0199]$.

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0132 * 60,254.5$), the expected value of X is $(0.0132)(60,254.5) = 795.359$. Also, the variance of X is 795.359, while the standard deviation, σ , is $\sqrt{795.359} = 28.20$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (739,852).

The probability that $(X \leq 739 \text{ or } X \geq 852) = 0.02283 + (1 - 0.97770) = 0.04513$.

The actual number of 65 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 771. The expected number of 65 maintenance actions is approximately 795. Thus, the residual on the Poisson process based on flying hours is $771 - 795 = 24$ (by taking absolute value).

Sortie Based:

If X is approximately Poisson ($0.0199 * 38,666$), the expected value of X is $(0.0199)(38,666) = 769.453$. Also, the variance of X is 769.453, while the standard deviation, σ , is $\sqrt{769.453} = 27.74$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (714,825).

The probability that $(X \leq 714 \text{ or } X \geq 825) = 0.02279 + (1 - 0.97739) = 0.0454$.

The actual number of 65 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 771. The expected number of 65 maintenance actions is approximately 769. Thus, the residual on the Poisson process based on sorties is $771 - 769 = 2$.

Poisson Process Calculations for the 71 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties.
Thus, Z is Poisson (0.0125) based on flying hours $[605/48,337.5 = 0.0125]$.

Z is Poisson (0.0189) based on sorties $[605/32,057 = 0.0189]$.

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0125 * 60,254.5$), the expected value of X is $(0.0125)(60,254.5) = 753.181$. Also, the variance of X is 753.181, while the standard deviation, σ , is $\sqrt{753.181} = 27.44$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (698,808).

The probability that $(X \leq 698 \text{ or } X \geq 808) = 0.02216 + (1 - 0.97713) = 0.04503$.

The actual number of 71 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 1,049. The expected number of 71 maintenance actions is approximately 753. Thus, the residual on the Poisson process based on flying hours is $1,049 - 753 = 296$.

Sortie Based:

If X is approximately Poisson ($0.0189 * 38,666$), the expected value of X is $(0.0189)(38,666) = 730.787$. Also, the variance of X is 730.787, while the standard deviation, σ , is $\sqrt{730.787} = 27.03$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (677,785).

The probability that $(X \leq 677 \text{ or } X \geq 785) = 0.02332 + (1 - 0.97754) = 0.04578$.

The actual number of 71 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 1,049. The expected number of 71 maintenance actions is approximately 731. Thus, the residual on the Poisson process based on sorties is $1,049 - 731 = 318$.

Poisson Process Calculations for the 74 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0547) based on flying hours [$2,643/48,337.5 = 0.0547$].

Z is Poisson (0.0824) based on sorties [$2,643/32,057 = 0.0824$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0547 * 60,254.5$), the expected value of X is $(0.0547)(60,254.5) = 3,295.921$. Also, the variance of X is 3,295.921, while the standard deviation, σ , is $\sqrt{3295.921} = 57.41$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (3,181, 3,411).

The probability that $(X \leq 3,181 \text{ or } X \geq 3,411) = 0.02265 + (1 - 0.97749) = 0.04516$.

The actual number of 74 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 3,710. The expected number of 74 maintenance actions is approximately 3,296. Thus, the residual on the Poisson process based on flying hours is $3,710 - 3,296 = 414$.

Sortie Based:

If X is approximately Poisson ($0.0824 * 38,666$), the expected value of X is $(0.0824)(38,666) = 3,186.078$. Also, the variance of X is 3,186.078, while the standard deviation, σ , is $\sqrt{3,186.078} = 56.45$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (3,073, 3,299).

The probability that $(X \leq 3,073 \text{ or } X \geq 3,299) = 0.02257 + (1 - 0.97728) = 0.04529$.

The actual number of 74 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 3,710. The expected number of 74 maintenance actions is approximately 3,186. Thus, the residual on the Poisson process based on sorties is $3,710 - 3,186 = 524$.

Poisson Process Calculations for the 75 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties.
Thus, Z is Poisson (0.0096) based on flying hours $[464/48,337.5 = 0.0096]$.

Z is Poisson (0.0145) based on sorties $[464/32,057 = 0.0145]$.

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0096 * 60,254.5$), the expected value of X is $(0.0096)(60,254.5) = 578.443$. Also, the variance of X is 578.443, while the standard deviation, σ , is $\sqrt{578.443} = 24.05$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (530,627).

The probability that $(X \leq 530 \text{ or } X \geq 627) = 0.02196 + (1 - 0.97824) = 0.04372$.

The actual number of 75 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 725. The expected number of 75 maintenance actions is approximately 578. Thus, the residual on the Poisson process based on flying hours is $725 - 578 = 147$.

Sortie Based:

If X is approximately Poisson ($0.0145 * 38,666$), the expected value of X is $(0.0145)(38,666) = 560.657$. Also, the variance of X is 560.657, while the standard deviation, σ , is $\sqrt{560.657} = 23.68$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (513,608).

The probability that $(X \leq 513 \text{ or } X \geq 608) = 0.02204 + (1 - 0.97723) = 0.04481$.

The actual number of 75 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 725. The expected number of 75 maintenance actions is approximately 561. Thus, the residual on the Poisson process based on sorties is $725 - 561 = 164$.

Poisson Process Calculations for the 76 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0201) based on flying hours [$974/48,337.5 = 0.0201$].

Z is Poisson (0.0304) based on sorties [$974/32,057 = 0.0304$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0201 * 60,254.5$), the expected value of X is $(0.0201)(60,254.5) = 1,211.115$. Also, the variance of X is 1,211.115, while the standard deviation, σ , is $\sqrt{1,211.115} = 34.80$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (1,142, 1,281).

The probability that $(X \leq 1,142 \text{ or } X \geq 1,281) = 0.02352 + (1 - 0.97768) = 0.04584$.

The actual number of 76 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 1,842. The expected number of 76 maintenance actions is approximately 1,211. Thus, the residual on the Poisson process based on flying hours is $1,842 - 1,211 = 631$.

Sortie Based:

If X is approximately Poisson ($0.0304 * 38,666$), the expected value of X is $(0.0304)(38,666) = 1,175.446$. Also, the variance of X is 1,175.446, while the standard deviation, σ , is $\sqrt{1,175.446} = 34.28$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (1,107, 1,244).

The probability that $(X \leq 1,107 \text{ or } X \geq 1,244) = 0.02294 + (1 - 0.97723) = 0.04571$.

The actual number of 76 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 1,842. The expected number of 76 maintenance actions is approximately 1,175. Thus, the residual on the Poisson process based on sorties is $1,842 - 1,175 = 667$.

Poisson Process Calculations for the 97 Work Unit Code Level:

Let Z be the number of maintenance actions in a specified period of flying hours or sorties. Thus, Z is Poisson (0.0036) based on flying hours [$172/48,337.5 = 0.0036$].

Z is Poisson (0.0054) based on sorties [$172/32,057 = 0.0054$].

Let X be a random variable denoting the number of maintenance actions in a fixed time period of either 60,254.5 flying hours or 38,666 sorties.

Recall, if X has a Poisson distribution with parameter λ , then $E(X) = V(X) = \lambda$.

Flying Hour Based:

If X is approximately Poisson ($0.0036 * 60,254.5$), the expected value of X is $(0.0036)(60,254.5) = 216.916$. Also, the variance of X is 216.916, while the standard deviation, σ , is $\sqrt{216.916} = 14.73$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (187,246).

The probability that $(X \leq 187 \text{ or } X \geq 246) = 0.02100 + (1 - 0.97595) = 0.04505$.

The actual number of 97 maintenance actions experienced from the 1994 F-15C data set for 60,254.5 flying hours is 476. The expected number of 97 maintenance actions is approximately 216. Thus, the residual on the Poisson process based on flying hours is $476 - 216 = 260$.

Sortie Based:

If X is approximately Poisson ($0.0054 * 38,666$), the expected value of X is $(0.0054)(38,666) = 208.796$. Also, the variance of X is 208.796, while the standard deviation, σ , is $\sqrt{208.796} = 14.45$.

By establishing a $\pm 2\sigma$ confidence interval, the interval would range from approximately (180,238).

The probability that $(X \leq 180 \text{ or } X \geq 238) = 0.02312 + (1 - 0.97835) = 0.04477$.

The actual number of 97 maintenance actions experienced from the 1994 F-15C data set for 38,666 sorties is 476. The expected number of 97 maintenance actions is approximately 209. Thus, the residual on the Poisson process based on sorties is $476 - 209 = 267$.

Hypotheses Testing for Equal Demand Rates Between 1993 and 1994 Data Sets

The two sample, two-tailed F test for equal population variances assumes: 1. Both sampled populations are normally distributed and 2. The samples are random and independent (McClave and Benson, 1988:445). The two-tailed hypotheses test is represented as:

$$\begin{aligned}H_o: \sigma_1^2 &= \sigma_2^2 \\H_a: \sigma_1^2 &\neq \sigma_2^2\end{aligned}$$

However, under the assumption of the Poisson distribution, the mean and variance of a Poisson random variable are both equal to λ (McClave and Benson, 1988:237). Therefore, the two-tailed hypotheses test is equivalent to:

$$\begin{aligned}H_o: \lambda_{93} &= \lambda_{94} \\H_a: \lambda_{93} &\neq \lambda_{94}\end{aligned}$$

The 93 represents the 1993 data set, while the 94 represents the 1994 data set.

$$\text{Test Statistic: } F = \frac{\text{Larger Sample Variance}}{\text{Smaller Sample Variance}} = \frac{s_1^2}{s_2^2} \text{ when } s_1^2 > s_2^2 \text{ (or } F = \frac{s_2^2}{s_1^2} \text{ when } s_2^2 > s_1^2)$$

Rejection Region: $F > F_{\alpha/2}$ when $s_1^2 > s_2^2$ where $F_{\alpha/2}$ is based on $v_1 = n_1 - 1$ and $v_2 = n_2 - 1$ degrees of freedom (or $F > F_{\alpha/2}$ when $s_2^2 > s_1^2$ where $F_{\alpha/2}$ is based on $v_1 = n_2 - 1$ and $v_2 = n_1 - 1$ degrees of freedom (McClave and Benson, 1988:445).

Each two digit work unit code level will now be tested with the two-tailed hypotheses test. For each test, $\alpha = 0.05$, $n_1 = 247$, and $n_2 = 340$. Thus, v_1 and v_2 are large enough to approach infinity and $F_{\alpha/2} = F_{0.025} \approx 1.00$ (Appendix B, Table IX, McClave and Benson).

Work Unit Code Level 11:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 237/48,337.5 = 0.0049$$

$$1994 \text{ Variance} = s_2^2 = 794/60,254.5 = 0.01318$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01318}{0.0049} = 2.69 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $2.69 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 237/32,057 = 0.0074$$

$$1994 \text{ Variance} = s_2^2 = 794/38,666 = 0.02053$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.02053}{0.0074} = 2.77 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $2.77 > 1.00$.

Work Unit Code Level 12:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 136/48,337.5 = 0.0028$$

$$1994 \text{ Variance} = s_2^2 = 318/60,254.5 = 0.00528$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.00528}{0.0028} = 1.89 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.89 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 136/32,057 = 0.0042$$

$$1994 \text{ Variance} = s_2^2 = 318/38,666 = 0.00822$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.00822}{0.0042} = 1.96 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.96 > 1.00$.

Work Unit Code Level 13:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 1,560/48,337.5 = 0.0323$$

$$1994 \text{ Variance} = s_2^2 = 2,092/60,254.5 = 0.0347$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.0347}{0.0323} = 1.07 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.07 > 1.00$.

Sortie Based:

$H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 1,560/32,057 = 0.0487$$

$$1994 \text{ Variance} = s_2^2 = 2,092/38,666 = 0.0541$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.0541}{0.0487} = 1.11 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.11 > 1.00$.

Work Unit Code Level 14:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 314/48,337.5 = 0.0065$$

$$1994 \text{ Variance} = s_2^2 = 1,121/60,254.5 = 0.0186$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.0186}{0.0065} = 2.86 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $2.86 > 1.00$.

Sortie Based:

$H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 314/32,057 = 0.0098$$

$$1994 \text{ Variance} = s_2^2 = 1,121/38,666 = 0.02899$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.02899}{0.0098} = 2.96 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $2.96 > 1.00$.

Work Unit Code Level 23:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 1,023/48,337.5 = 0.0212$$

$$1994 \text{ Variance} = s_2^2 = 2,069/60,254.5 = 0.0343$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.0343}{0.0212} = 1.62 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.62 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 1,023/32,057 = 0.0319$$

$$1994 \text{ Variance} = s_2^2 = 2,069/38,666 = 0.0535$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.0535}{0.0319} = 1.68 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.68 > 1.00$.

Work Unit Code Level 24:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 471/48,337.5 = 0.0097$$

$$1994 \text{ Variance} = s_2^2 = 1,114/60,254.5 = 0.01849$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01849}{0.0097} = 1.91 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.91 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 471/32,057 = 0.0147$$

$$1994 \text{ Variance} = s_2^2 = 1,114/38,666 = 0.02881$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.02881}{0.0147} = 1.96 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.96 > 1.00$.

Work Unit Code Level 41:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 301/48,337.5 = 0.0062$$

$$1994 \text{ Variance} = s_2^2 = 667/60,254.5 = 0.01107$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01107}{0.0062} = 1.79 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.79 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 301/32,057 = 0.0094$$

$$1994 \text{ Variance} = s_2^2 = 667/38,666 = 0.01725$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01725}{0.0094} = 1.84 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.84 > 1.00$.

Work Unit Code Level 42:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 301/48,337.5 = 0.0062$$

$$1994 \text{ Variance} = s_2^2 = 766/60,254.5 = 0.01271$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01271}{0.0062} = 2.05 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $2.05 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 301/32,057 = 0.0094$$

$$1994 \text{ Variance} = s_2^2 = 766/38,666 = 0.01981$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01981}{0.0094} = 2.11 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $2.11 > 1.00$.

Work Unit Code Level 44:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 475/48,337.5 = 0.0098$$

$$1994 \text{ Variance} = s_2^2 = 665/60,254.5 = 0.01104$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01104}{0.0098} = 1.13 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.13 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 475/32,057 = 0.0148$$

$$1994 \text{ Variance} = s_2^2 = 665/38,666 = 0.01720$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01720}{0.0148} = 1.16 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.16 > 1.00$.

Work Unit Code Level 45:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 186/48,337.5 = 0.0038$$

$$1994 \text{ Variance} = s_2^2 = 965/60,254.5 = 0.01602$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01602}{0.0038} = 4.22 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $4.22 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 186/32,057 = 0.0058$$

$$1994 \text{ Variance} = s_2^2 = 965/38,666 = 0.02496$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.02496}{0.0058} = 4.30 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $4.30 > 1.00$.

Work Unit Code Level 46:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 284/48,337.5 = 0.0058$$

$$1994 \text{ Variance} = s_2^2 = 808/60,254.5 = 0.01341$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01341}{0.0058} = 2.31 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $2.31 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 284/32,057 = 0.0088$$

$$1994 \text{ Variance} = s_2^2 = 808/38,666 = 0.02090$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.02090}{0.0088} = 2.38 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $2.38 > 1.00$.

Work Unit Code Level 47:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 133/48,337.5 = 0.0028$$

$$1994 \text{ Variance} = s_2^2 = 280/60,254.5 = 0.00465$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.00465}{0.0028} = 1.66 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.66 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 133/32,057 = 0.0041$$

$$1994 \text{ Variance} = s_2^2 = 280/38,666 = 0.00724$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.00724}{0.0041} = 1.77 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.77 > 1.00$.

Work Unit Code Level 49:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 49/48,337.5 = 0.0010$$

$$1994 \text{ Variance} = s_2^2 = 118/60,254.5 = 0.00196$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.00196}{0.0010} = 1.96 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.96 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 49/32,057 = 0.0015$$

$$1994 \text{ Variance} = s_2^2 = 118/38,666 = 0.00305$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.00305}{0.0015} = 2.03 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $2.03 > 1.00$.

Work Unit Code Level 51:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 747/48,337.5 = 0.0155$$

$$1994 \text{ Variance} = s_2^2 = 1,043/60,254.5 = 0.01731$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01731}{0.0155} = 1.12 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.12 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 747/32,057 = 0.0233$$

$$1994 \text{ Variance} = s_2^2 = 1,043/38,666 = 0.02697$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.02697}{0.0233} = 1.16 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.16 > 1.00$.

Work Unit Code Level 52:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 181/48,337.5 = 0.0037$$

$$1994 \text{ Variance} = s_2^2 = 284/60,254.5 = 0.00471$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.00471}{0.0037} = 1.27 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.27 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 181/32,057 = 0.0056$$

$$1994 \text{ Variance} = s_2^2 = 284/38,666 = 0.00734$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.00734}{0.0056} = 1.31 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.31 > 1.00$.

Work Unit Code Level 55:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 197/48,337.5 = 0.0041$$

$$1994 \text{ Variance} = s_2^2 = 502/60,254.5 = 0.00833$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.00833}{0.0041} = 2.03 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $2.03 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 197/32,057 = 0.0061$$

$$1994 \text{ Variance} = s_2^2 = 502/38,666 = 0.01298$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01298}{0.0061} = 2.13 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $2.13 > 1.00$.

Work Unit Code Level 57:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 61/48,337.5 = 0.0013$$

$$1994 \text{ Variance} = s_2^2 = 265/60,254.5 = 0.00439$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.00439}{0.0013} = 3.38 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $3.38 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 61/32,057 = 0.0019$$

$$1994 \text{ Variance} = s_2^2 = 265/38,666 = 0.00685$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.00685}{0.0019} = 3.61 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $3.61 > 1.00$.

Work Unit Code Level 63:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 747/48,337.5 = 0.0155$$

$$1994 \text{ Variance} = s_2^2 = 1,057/60,254.5 = 0.01754$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01754}{0.0155} = 1.13 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.13 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 747/32,057 = 0.0233$$

$$1994 \text{ Variance} = s_2^2 = 1,057/38,666 = 0.02734$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.02734}{0.0233} = 1.17 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.17 > 1.00$.

Work Unit Code Level 65:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 639/48,337.5 = 0.0132$$

$$1994 \text{ Variance} = s_2^2 = 771/60,254.5 = 0.0128$$

$$\text{Test Statistic: } F = \frac{s_1^2}{s_2^2} = \frac{0.0132}{0.0128} = 1.03 \text{ because } s_1^2 > s_2^2.$$

Result: Accept $H_o: \lambda_{93} = \lambda_{94}$ because $F \approx F_{\alpha/2}$ or $1.03 \approx 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 639/32,057 = 0.0199$$

$$1994 \text{ Variance} = s_2^2 = 771/38,666 = 0.0199$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.0199}{0.0199} = 1.00 \text{ because } s_2^2 = s_1^2.$$

Result: Accept $H_o: \lambda_{93} = \lambda_{94}$ because $F = F_{\alpha/2}$ or $1.00 = 1.00$.

Work Unit Code Level 71:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 605/48,337.5 = 0.0125$$

$$1994 \text{ Variance} = s_2^2 = 1,049/60,254.5 = 0.0174$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.0174}{0.0125} = 1.39 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.39 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 605/32,057 = 0.0189$$

$$1994 \text{ Variance} = s_2^2 = 1,049/38,666 = 0.02713$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.02713}{0.0189} = 1.44 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.44 > 1.00$.

Work Unit Code Level 74:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 2,643/48,337.5 = 0.0547$$

$$1994 \text{ Variance} = s_2^2 = 3,710/60,254.5 = 0.06157$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.06157}{0.0547} = 1.13 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.13 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 2,643/32,057 = 0.0824$$

$$1994 \text{ Variance} = s_2^2 = 3,710/38,666 = 0.09595$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.09595}{0.0824} = 1.16 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.16 > 1.00$.

Work Unit Code Level 75:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 464/48,337.5 = 0.0096$$

$$1994 \text{ Variance} = s_2^2 = 725/60,254.5 = 0.01203$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01203}{0.0096} = 1.25 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.25 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 464/32,057 = 0.0145$$

$$1994 \text{ Variance} = s_2^2 = 725/38,666 = 0.01875$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01875}{0.0145} = 1.29 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.29 > 1.00$.

Work Unit Code Level 76:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 974/48,337.5 = 0.0201$$

$$1994 \text{ Variance} = s_2^2 = 1,842/60,254.5 = 0.03057$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.03057}{0.0201} = 1.52 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.52 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 974/32,057 = 0.0304$$

$$1994 \text{ Variance} = s_2^2 = 1,842/38,666 = 0.04764$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.04764}{0.0304} = 1.57 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $1.57 > 1.00$.

Work Unit Code Level 97:

Flying Hour based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 172/48,337.5 = 0.0036$$

$$1994 \text{ Variance} = s_2^2 = 476/60,254.5 = 0.00790$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.00790}{0.0036} = 2.19 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $2.19 > 1.00$.

Sortie Based: $H_o: \lambda_{93} = \lambda_{94}$ versus $H_a: \lambda_{93} \neq \lambda_{94}$

$$1993 \text{ Variance} = s_1^2 = 172/32,057 = 0.0054$$

$$1994 \text{ Variance} = s_2^2 = 476/38,666 = 0.01231$$

$$\text{Test Statistic: } F = \frac{s_2^2}{s_1^2} = \frac{0.01231}{0.0054} = 2.28 \text{ because } s_2^2 > s_1^2.$$

Result: Reject $H_o: \lambda_{93} = \lambda_{94}$. Conclude $H_a: \lambda_{93} \neq \lambda_{94}$ because $F > F_{\alpha/2}$ or $2.28 > 1.00$.

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Kephart Vita

Captain Steven D. Kephart is from Sanborn, Pennsylvania. He graduated from the Pennsylvania State University in 1986 with a Bachelor of Science degree in Petroleum and Natural Gas Engineering. After receiving his commission into the United States Air Force through the Officer Training School, Captain Kephart completed the supply operations officer course and was assigned to the 436 Military Airlift Wing, Dover AFB, Delaware.

During his tour at Dover AFB, Captain Kephart filled a variety of supply operations positions in support of the C-5A and C-5B aircraft. These positions included: Assistant, Materiel Management Branch; Chief, Management and Systems Branch; and Fuels Management Officer.

In 1991, he was assigned to Aviano AB, Italy, where he served for one year as the 40 Support Wing Fuels Management Officer. In 1992, Captain Kephart transferred to Headquarters, Sixteenth Air Force, also located at Aviano, and was assigned duties as Chief, Supply and Fuels Operations Division. During his tour at Aviano, Captain Kephart supported operations such as PROVIDE COMFORT, RESTORE HOPE, PROVIDE PROMISE, and DENY FLIGHT. He also received distinction as the 1992 Headquarters, United States Air Forces in Europe Outstanding Fuels Officer.

Upon reassignment from overseas, Captain Kephart entered the Air Force Institute of Technology at Wright-Patterson AFB, Ohio, and graduated in 1995 with a Masters degree in Logistics Management. He was subsequently assigned to the F-15 Systems Program Office, Warner-Robins ALC, Robins AFB, Georgia.

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Roberts Vita

Captain Richard C. Roberts is from Whitehall, Montana graduating from Whitehall High School in 1982. He received a scholarship to study in Denmark where he graduated from Himmelev Gymnasium, Roskilde, Denmark in 1983. Captain Roberts received his Bachelor of Science degree in Business Administration/ Finance from the Montana School of Mineral Science and Technology in 1988. He was enlisted in the Montana Army National Guard, 2/152nd F Troop, as an Armor Calvary Scout. He subsequently was employed at the State Street Bank and Trust in Boston as a Mutual Funds Account Controller before attending Officer Training School in 1989.

Captain Roberts was initially assigned to the 62nd Supply Squadron, McChord AFB, Washington. He deployed to Al Minhad Air Base, United Arab Emirates in August 1990 to support Operations DESERT SHIELD/STORM. As a Second Lieutenant, he was appointed the Chief of Supply for the 388th Tactical Fighter Wing until March 1991. Captain Roberts' next assignment was to San Vito Air Station, Italy in 1991 where he was the Deputy Chief of Supply until the base closed in 1994.

He was accepted into the Air Force Institute of Technology at Wright-Patterson AFB, Ohio in 1994 and graduated with a Masters degree in Logistics Management in 1995. His follow-on assignment was to the Air Force Security Assistance Center at Wright-Patterson AFB. Captain Roberts is married to the former Susan Ann Graham of Falmouth, Massachusetts. They have two children, Sarah and Matthew.

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| 13. ABSTRACT (Maximum 200 words) Current Air Force demand forecasting systems, D041 and REALM, assume reparable demand is solely flying hour driven. The purpose of this study is to evaluate the relationship between demands, flying hours, and number of sorties at the work unit code level and improve reparable demand forecasting. A three phase methodology is used to analyze the demand, flying hour, and sortie relationship. The first phase uses multiple regression to determine a relationship at various work unit code levels. Multiple regression provides limited correlation between demands, flying hours, and sorties. The second phase uses Poisson regression to evaluate the integer, count nature of the demands variable used in the analysis. Poisson regression also exhibits poor correlation between demands, flying hours, and sorties. The third phase fits a Poisson process to the data and produces better results than multiple or Poisson regression. However, the Poisson process performs poorly in estimating future demands at the work unit code level, based on historical flying hour and sortie demand rate occurrences. Despite research at the work unit code level, the study results support previous demand forecasting research, which has been unable to demonstrate an accurate demand, flying hour, and sortie relationship. | | | | |
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